



WORKSHOP: Material Failure of Metals

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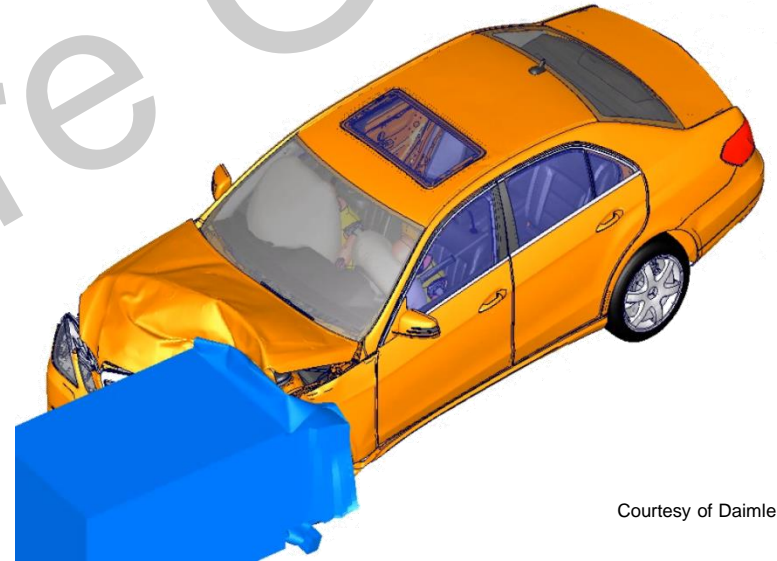
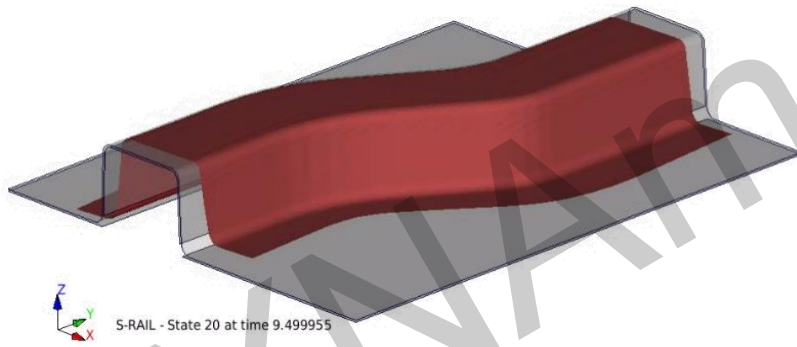
Agenda

- Motivation
- Preliminaries (experimental evidence, triaxiality, ...)
- Aspect that influence failure prediction
- Short description of the GISSMO model
 - Stress state dependence
 - Plasticity model
 - Non-proportional loadings
- Calibration of a GISSMO material card
 - Usual methods
 - Live demo (reverse engineering)
- Mesh dependence/regularization in GISSMO
- Summary

Motivation

Damage and failure play an important role in simulation

Focus on the application in crash simulation (and metal forming)



Courtesy of Daimler AG

What should I have in the input?

Does it fail or not?

Material behavior

Aspects that influence failure and its prediction

- Stress state dependence
- Plastic yielding (yield surface shape) and plastic flow
- Non-proportional loading / strain path dependence
- Instability / localization issues / mesh dependence
- Element formulation (shells, solids, under/fully integrated, ...)
- Pre-strain and pre-damage
- Anisotropy (in plasticity and in failure properties)
- Strain rate dependence (adiabatic process at high strain rates!)
- Heat affected zones due to welding
- Scattering of material properties
-

Ductile fracture

Experimental evidence regarding fracture of metals

Early works by authors like Bridgman, Rice, Tracey, Mackenzie, Hancock, Brown, among others, experimentally observed that fracture was dependent on the **triaxiality ratio of stress**.

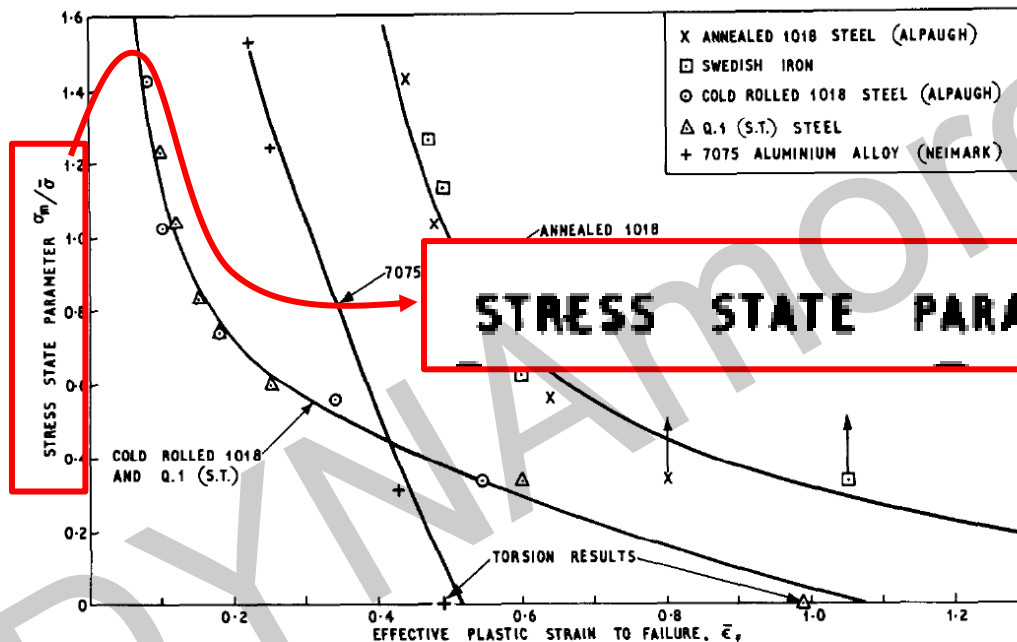


Fig. 19. Comparison of failure initiation curves.

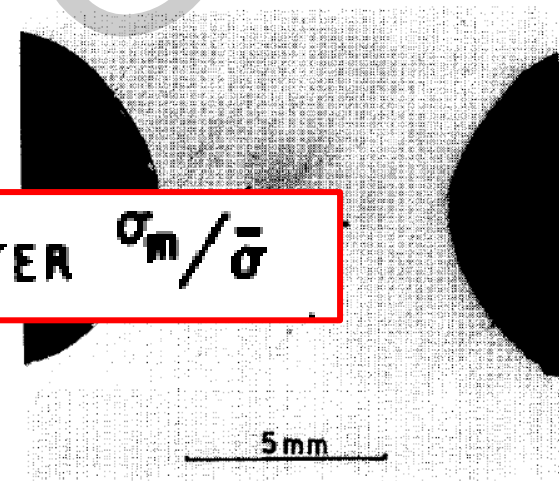


Fig. 1. Cracking in notch tension specimen.

Source: Mackenzie et al. 1977. On the influence of state of stress on ductile failure initiation in high strength steels.

Preliminaries

Some usual definitions regarding the true stress tensor

The true stress tensor is symmetric and can be split in two parts, i.e.,

$$\boldsymbol{\sigma} = \begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ & \sigma_{22} & \sigma_{23} \\ & & \sigma_{33} \end{bmatrix} = \underbrace{\mathbf{s} + \frac{1}{3}\text{tr}(\boldsymbol{\sigma})\mathbf{I}}_{\text{mean stress}} = \underbrace{\begin{bmatrix} s_{11} & s_{12} & s_{13} \\ & s_{22} & s_{23} \\ & & s_{33} \end{bmatrix}}_{\text{stress deviator}} - \underbrace{p}_{\text{hydrostatic pressure}} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

The principal stress tensor and its invariants are given by

$$\boldsymbol{\sigma} = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{bmatrix} \quad \begin{aligned} I_1 &= \sigma_1 + \sigma_2 + \sigma_3 \\ I_2 &= \sigma_1\sigma_2 + \sigma_2\sigma_3 + \sigma_1\sigma_3 \\ I_3 &= \sigma_1\sigma_2\sigma_3 \end{aligned} \quad \begin{aligned} J_1 &= s_1 + s_2 + s_3 = 0 \\ J_2 &= \frac{1}{6} \left[(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 \right] \\ J_3 &= s_1s_2s_3 = \frac{2}{27}I_1^3 - \frac{1}{3}I_1I_2 + I_3 \end{aligned}$$

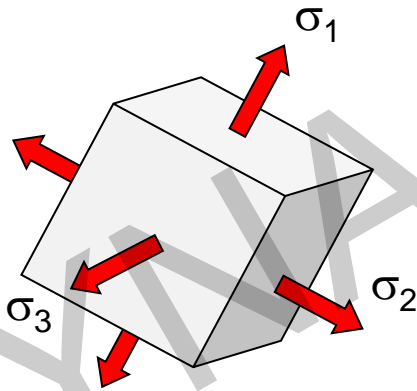
The equivalent or von Mises stress is defined as

$$\sigma_{eq} = \sqrt{3J_2} = \sqrt{\frac{1}{2} \left[(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 \right]}$$

Stress state parameters

Triaxiality

- The term “triaxiality” was apparently coined by Mackenzie, Hancock and Brown in 1977
- It quantifies the ratio among the three principal stresses ($\sigma_1, \sigma_2, \sigma_3$) through a single scalar
- It is a practical parameter for the characterization of material failure



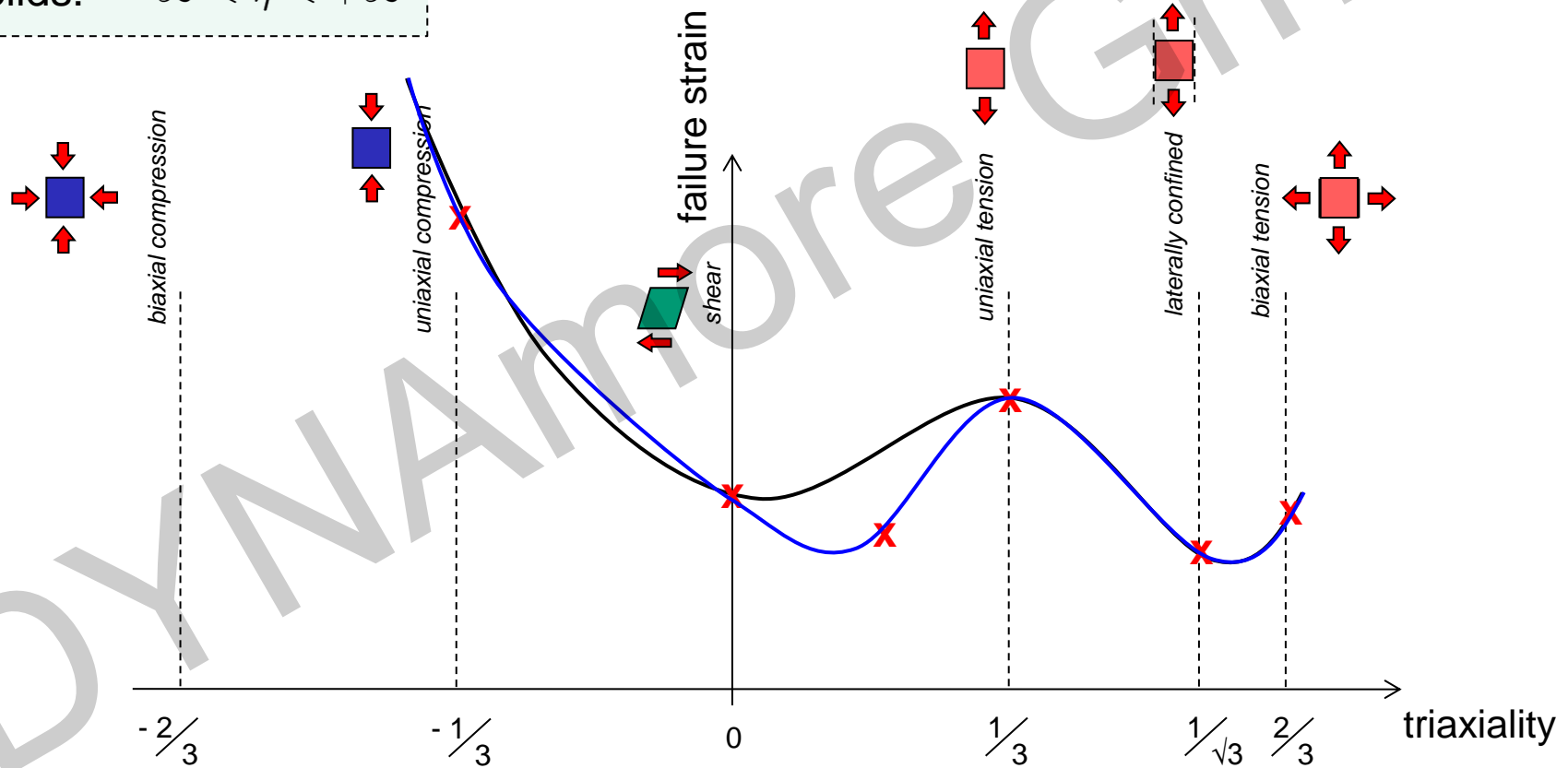
Definition of stress triaxiality: $\eta = -\frac{p}{\sigma_{eq}}$

Stress state parameters

Triaxiality under plane stress

Shells: $-\frac{2}{3} < \eta < \frac{2}{3}$

Solids: $-\infty < \eta < +\infty$



Short description of GISSMO

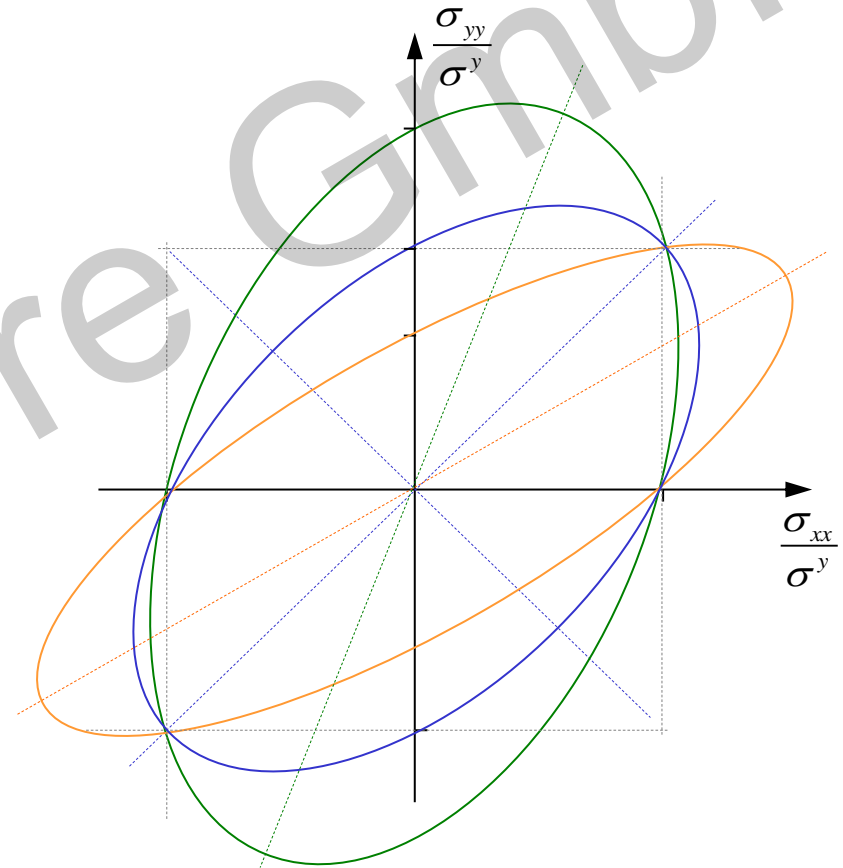
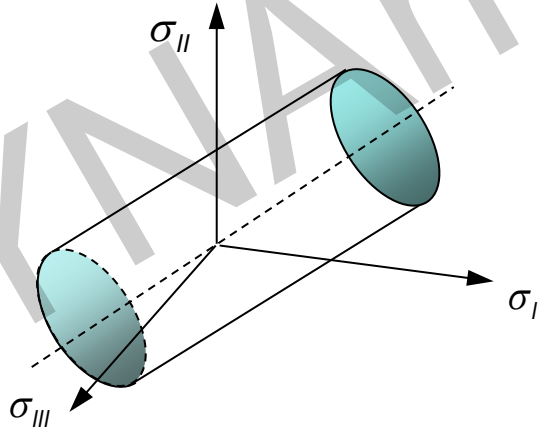
(Generalized Incremental Stress State dependent Model)

Plasticity

Many elasto-plastic models in LS-DYNA

- *MAT_PLASTIC_KINEMATIC
- *MAT_PIECEWISE_LINEAR_PLASTICITY
- *MAT_BARLAT_ANISOTROPIC_PLASTICITY
- *MAT_3-PARAMETER_BARLAT
- *MAT_ANISOTROPIC_VISCOPLASTIC
- *MAT_ORTHO_ELASTIC_PLASTIC
- *MAT_HILL_3R
- *MAT_BARLAT_YLD2000
- *MAT_WTM_STM
- *MAT_CORUS_VEGTER
- *MAT_CAZACU_BARLAT
- *MAT_HILL_90

.....



Plasticity

Modular concept with *MAT_ADD_EROSION

*MAT_3-PARAMETER_BARLAT

```

$      MID      RO      E      PR      HR      P1      P2      ITER
      10      7.85E-6      210.0      0.3      3.0      0.0      0.0      0.0
$      M      R00/AB      R45/CB      R90/HB      LCID      E0      SPI      P3
      8      0.75      0.85      1.05      500
$      AOPT      C      P      VLCID      PB
      2
$
$      A1      A2      A3
      1.0      0.0      0.0
$      V1      V2      V3      D1      D2      D3      BETA
      0.0      0.0      0.0
$

```

*MAT_ADD_EROSION

```

$      MID      EXCL      MXPRES      MNEPS      EFFEPS      VOLEPS      NUMFIP      NCS
      10
$      MNPRES      SIGP1      SIGVM      MXEPS      EPSSH      SIGTH      IMPULSE      FAILTM
$      IDAM      DMGTYP      LCSDG      ECRIT      DMGEXP      DCRIT      FADEXP      LCREGD
      1      1      100      -200      2      2.5      400
$      SIZFLG      REFSZ      NAHSV      LCSRS      REGSHR      REGBIAX
      14      1.0      0.0

```

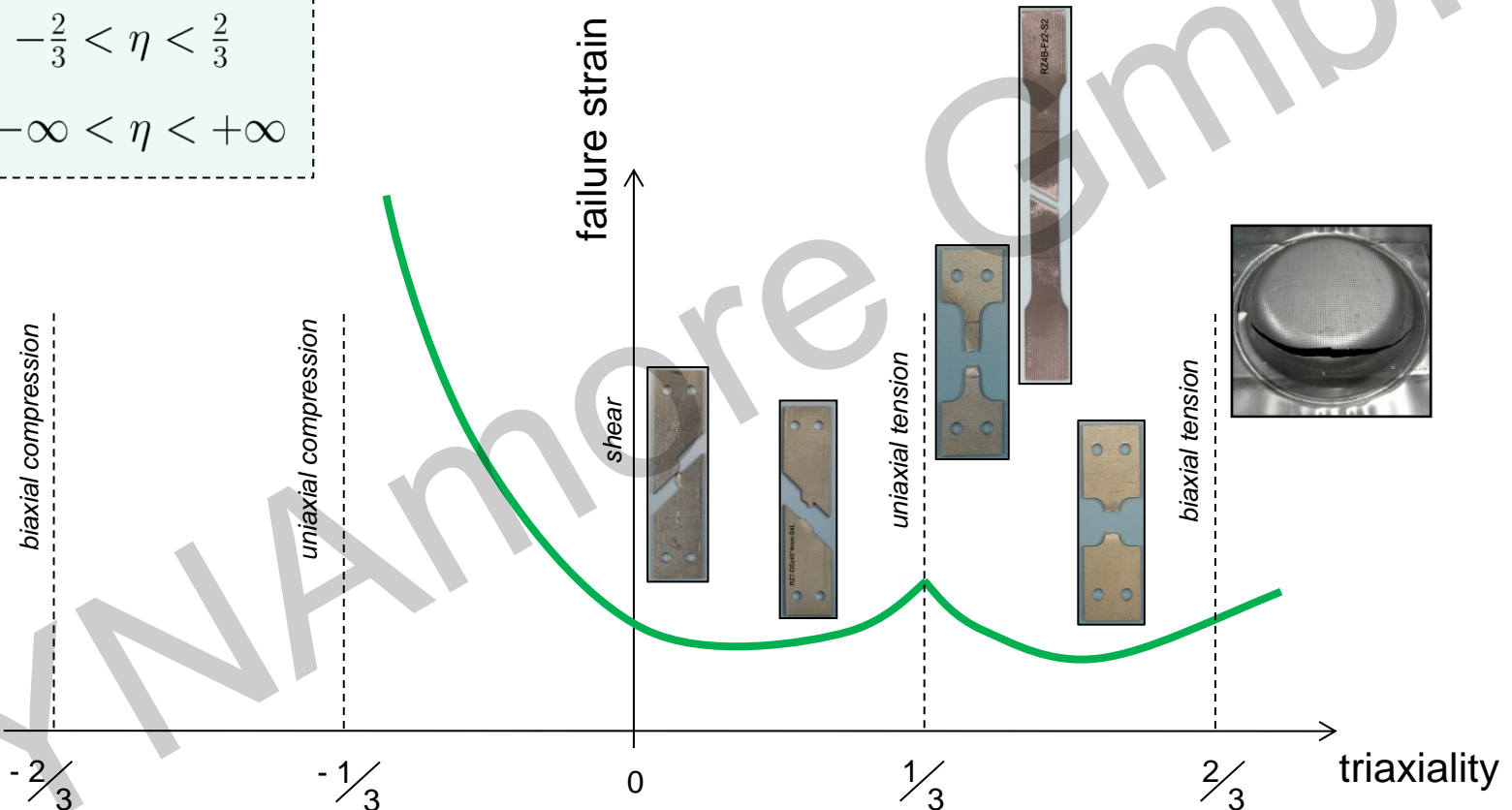
The modular concept of *MAT_ADD_EROSION allows the user to combine GISSMO with any other elasto-plastic model in LS-DYNA

Stress state dependence

Material failure strongly depends on the stress state

Shells: $-\frac{2}{3} < \eta < \frac{2}{3}$

Solids: $-\infty < \eta < +\infty$



Suitable specimens are generally preferred for the accurate calibration of material failure

Stress state dependence

Defining a failure curve (only triaxiality)

```
*MAT_PIECEWISE_LINEAR_PLASTICITY
```

```
$ MID RO E PR SIGY ETAN FAIL TDEL
```

```
$ C P LCSS LCSR VP
```

```
...
```

```
*MAT_ADD_EROSION
```

```
$ MID EXCL MXPRES MNEPS EFFEPS VOLEPS NUMFIP NCS
```

```
$ MNPRES SIGP1 SIGVM MXEPS EPSSH SIGTH IMPULSE FAILTM
```

```
$ IDAM DMGTYP LCSDG ECRIT DMGEXP DCRIT FADEXP LCREGD
```

```
$ 1 1 100 -200 2 2.5 400
```

```
$ SIZEFLG REFSZ NANSV LCSR REGSHR REGBIAX
```

```
$ 14 1.0 0.0
```

```
*DEFINE_CURVE
```

```
$ LCID SIDR SCLA SCLO
```

```
$ 100 0 1.0 1.0
```

```
$ eta epsf
```

```
-0.6666 4.0
```

```
-0.3333 4.0
```

```
0.0000 0.2
```

```
0.3333 0.6
```

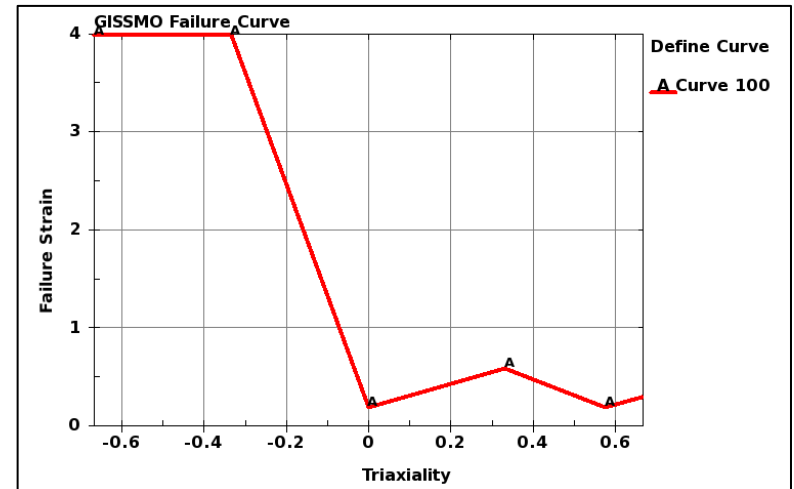
```
0.5775 0.2
```

```
0.6666 0.3
```

Main material model
(e.g. *MAT_024)

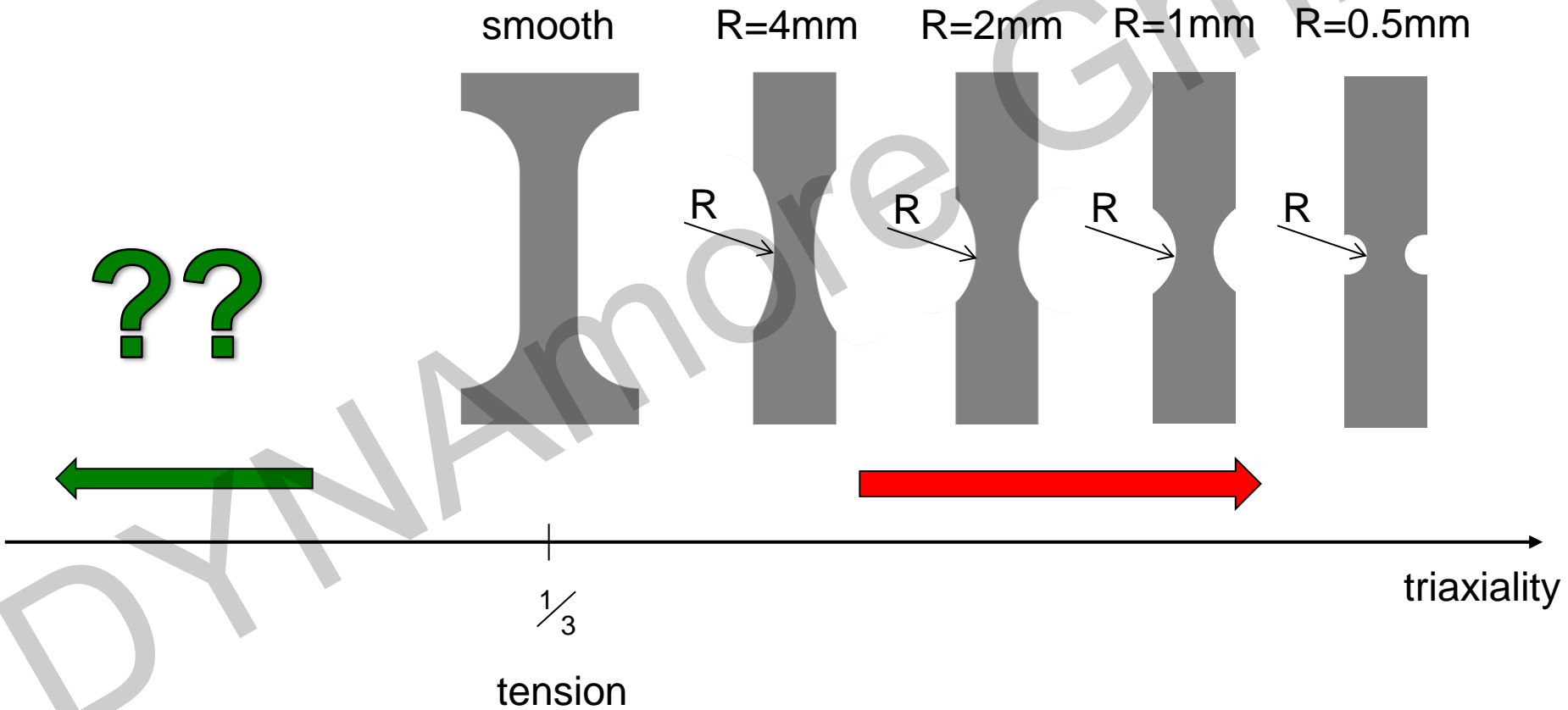
General parameters
in *MAT_ADD_EROSION

GISSMO failure
parameters



Stress state dependence

How to vary the triaxiality in experiments?

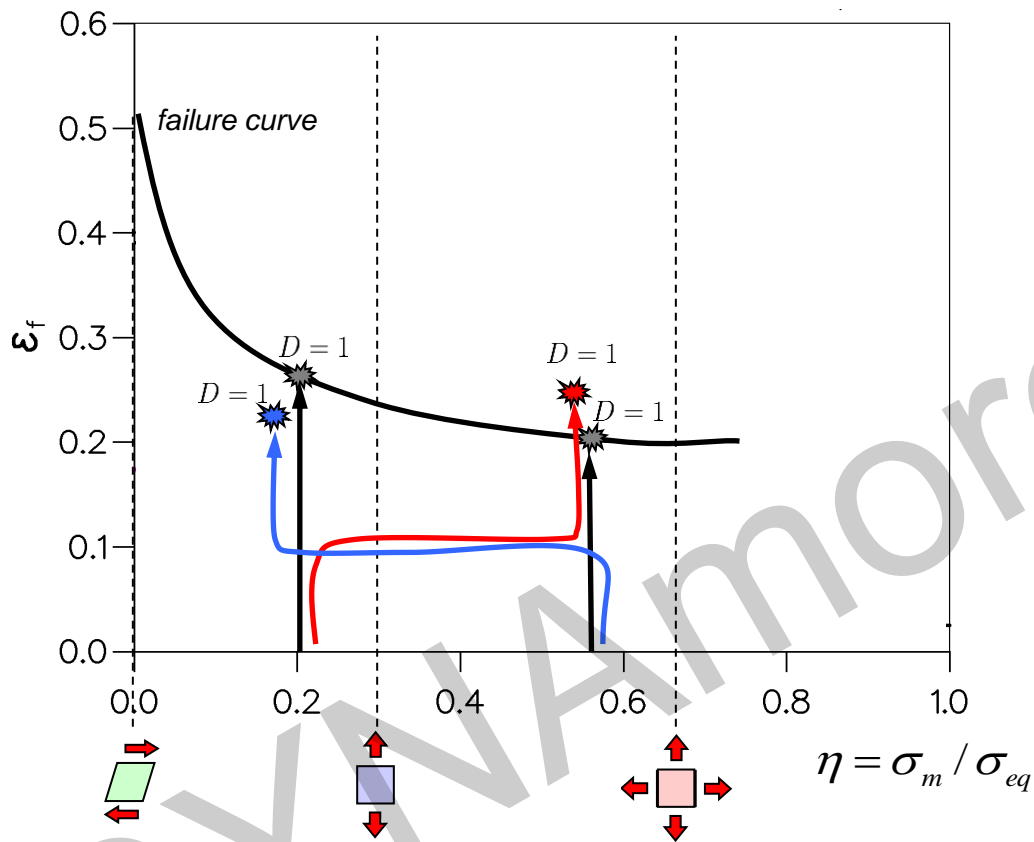


Non-proportional loading

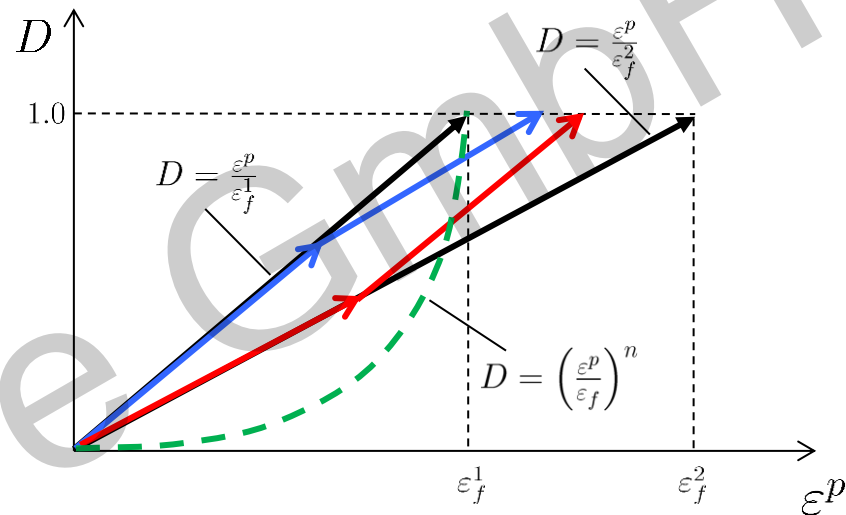
(very important in crash simulations)

Non-proportional loading

Accumulation of damage in GISSMO



Damage is incrementally accumulated as a function of the plastic strain increment and of the failure curve. **Failure occurs when $D = 1$!**



$$D = \left(\frac{\epsilon^p}{\epsilon_f(\eta)} \right)^n$$

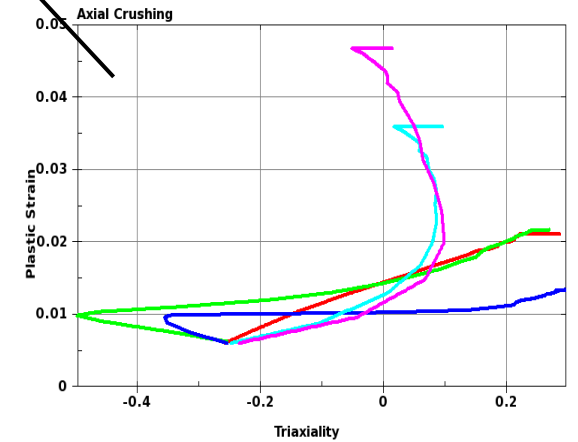
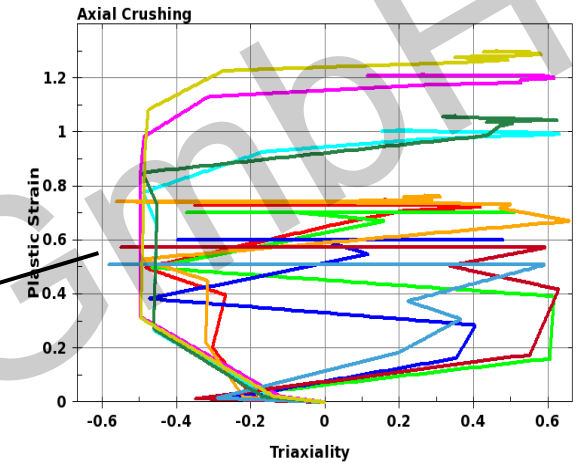
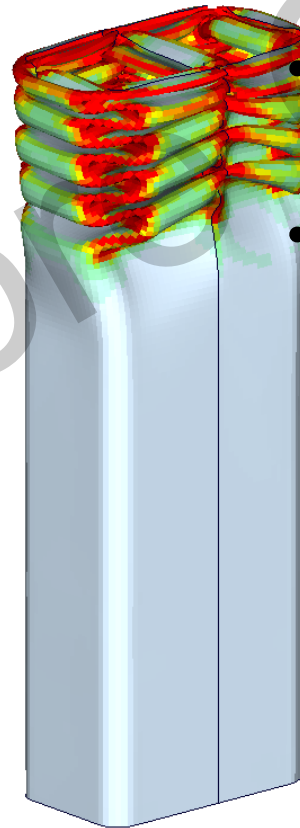
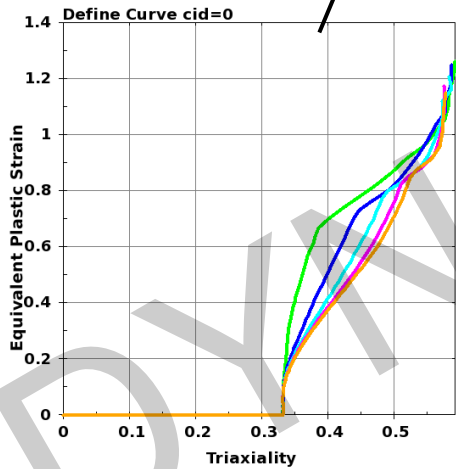
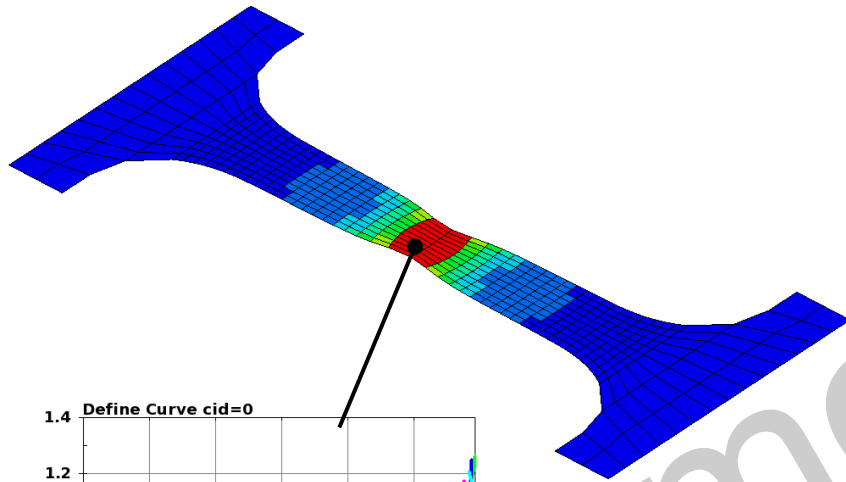


Differentiated under constant triaxiality

$$\Delta D = \frac{n}{\epsilon_f(\eta)} D^{(1-\frac{1}{n})} \Delta \epsilon_{eq}^p$$

Non-proportional loading

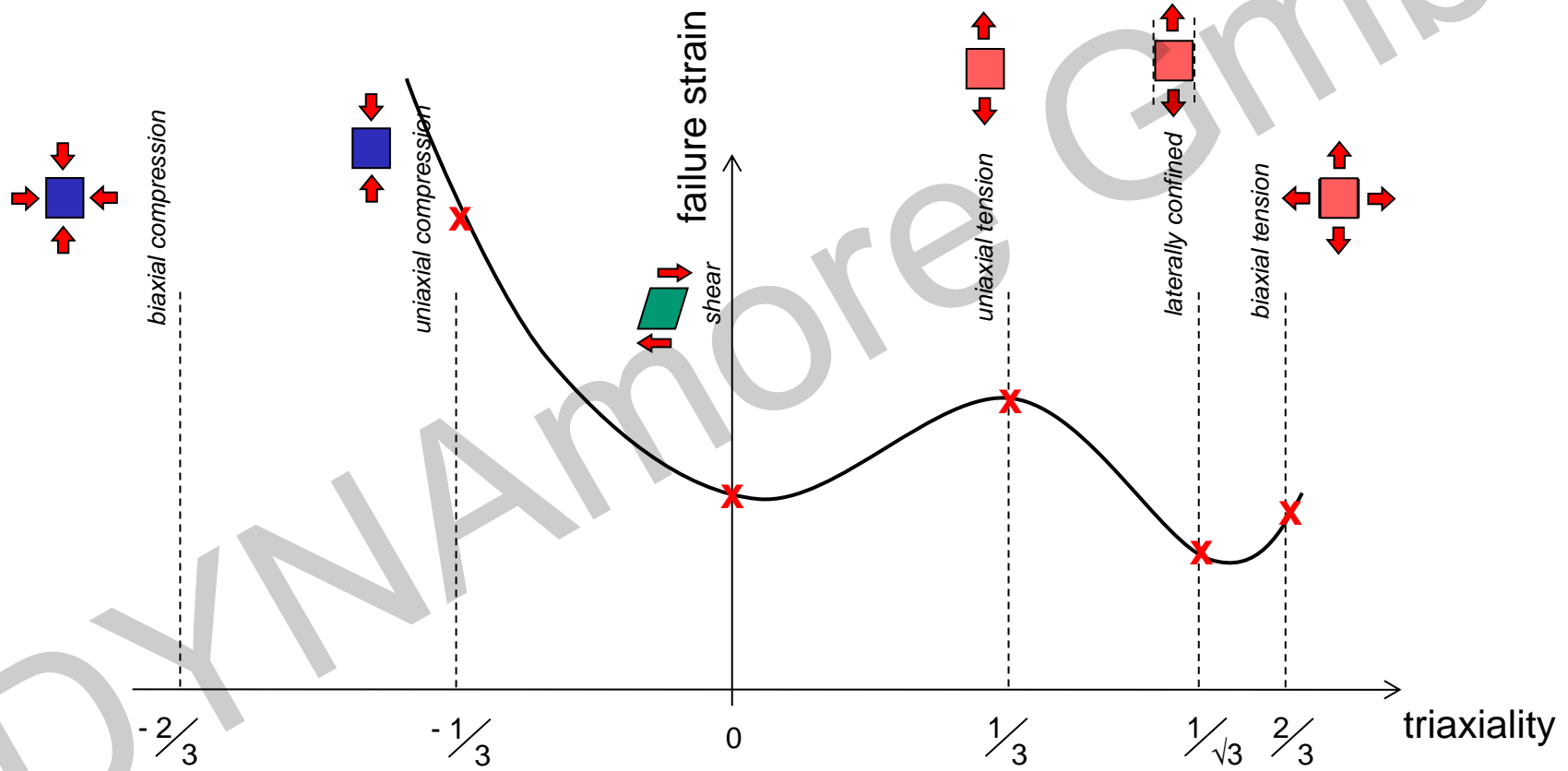
Examples of some strain paths in FE simulation



Calibration of a GISSMO material card

Plane stress case (e.g. standard shells)

Identification of failure strain for various triaxialities



Calibration of GISSMO

Usual methods for the identification of the failure strains

Method

Advantages

Disadvantages

Geometrical measurement of the failure strain

- No or few simulations needed for the model calibration
- Results are often good enough

- Somewhat limited precision
- Plasticity model assumed accurate
- Non-proportionality disregarded
- Influence of numerical parameters when simulating real parts

Optical measurement of the failure strain

- No or few simulations needed
- Higher measurement precision than above method

- Plasticity model assumed accurate
- Non-proportionality disregarded
- Influence of numerical parameters

Reverse engineering for failure strain identification

- Influence of numerical parameters inherently considered
- Non-proportionality considered
- Limitations of plasticity model can be somewhat compensated

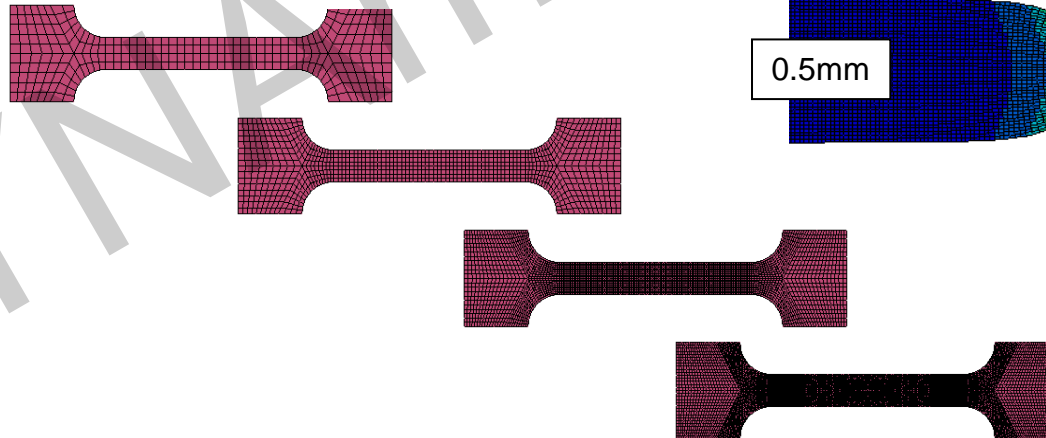
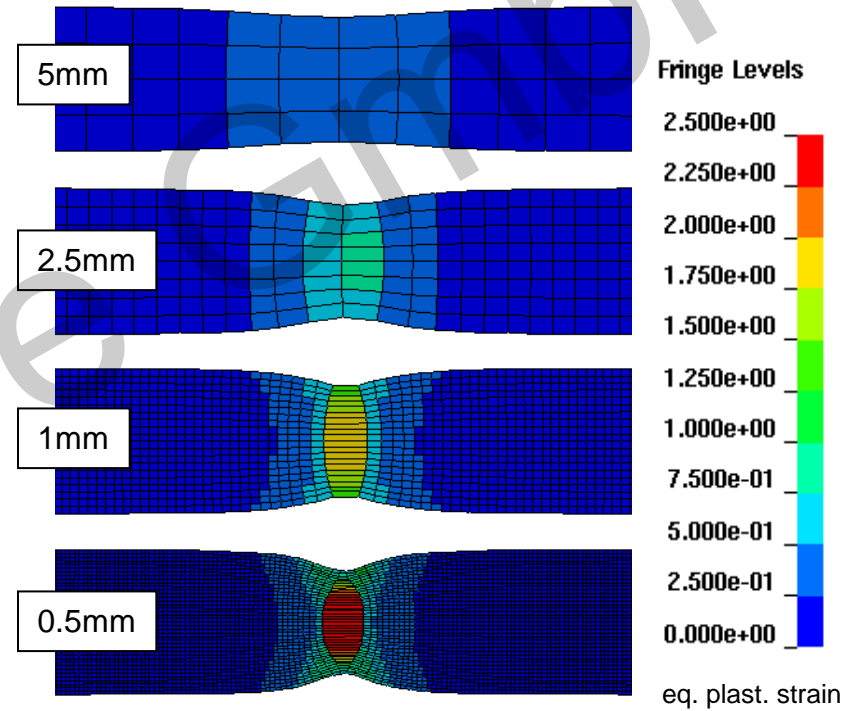
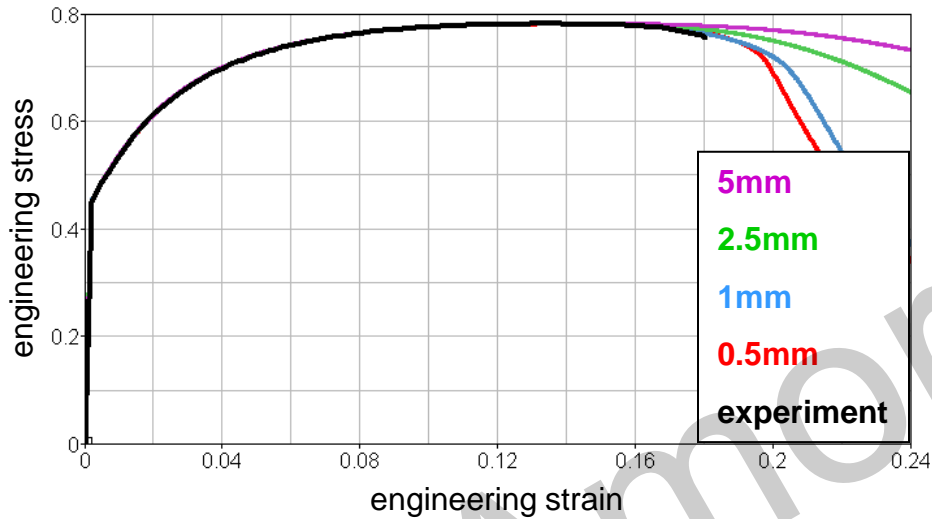
- More simulations needed
- Only valid for the numerical parameters used in the calibration
- Numerical calibration may still not be accurate enough

LIVE DEMO: Reverse engineering method

Spurious mesh dependence

Mesh dependence

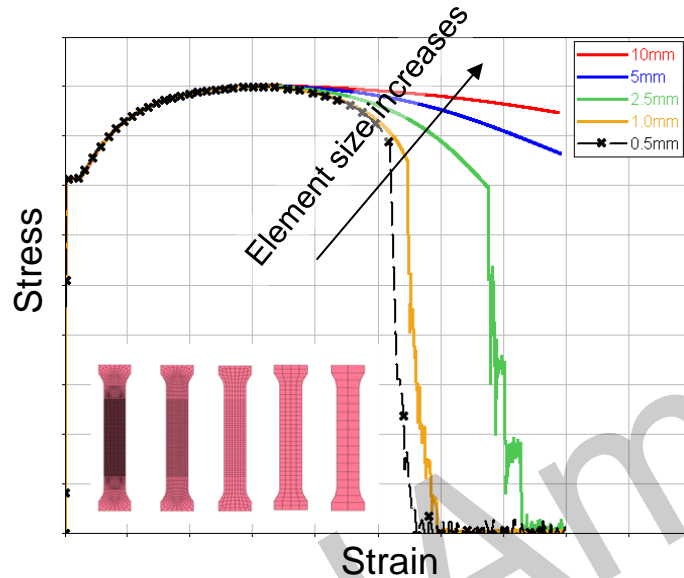
FE solution is spuriously mesh dependent after the necking point



Mesh dependence

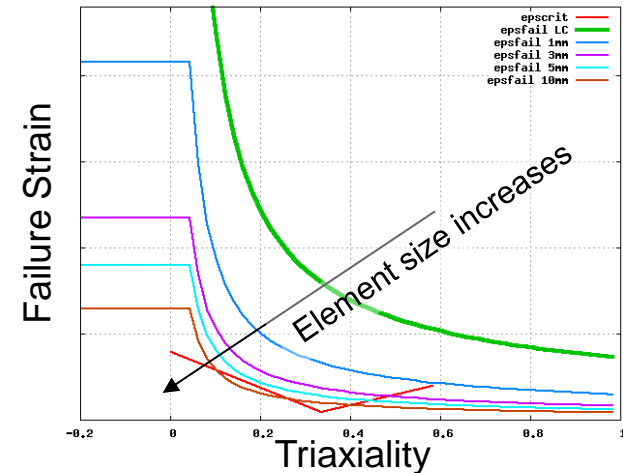
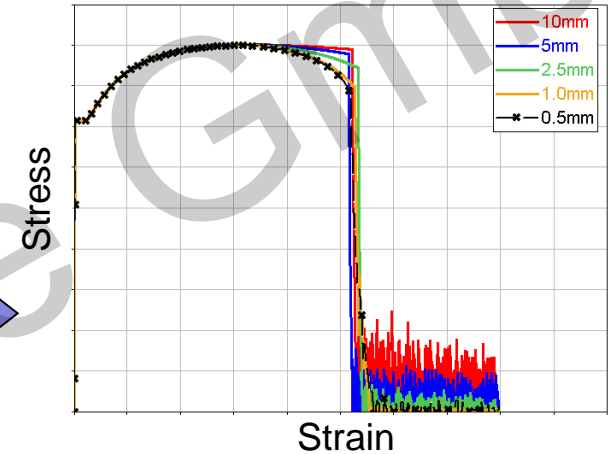
Regularization for different mesh sizes

Mesh size dependence



- Inherent mesh-size dependence of results in the post-critical region
- Simulation (and calibration) of tensile test specimen with different mesh sizes

Simple regularization



Mesh dependence

Regularization for different mesh sizes

*MAT_ADD_EROSION

```

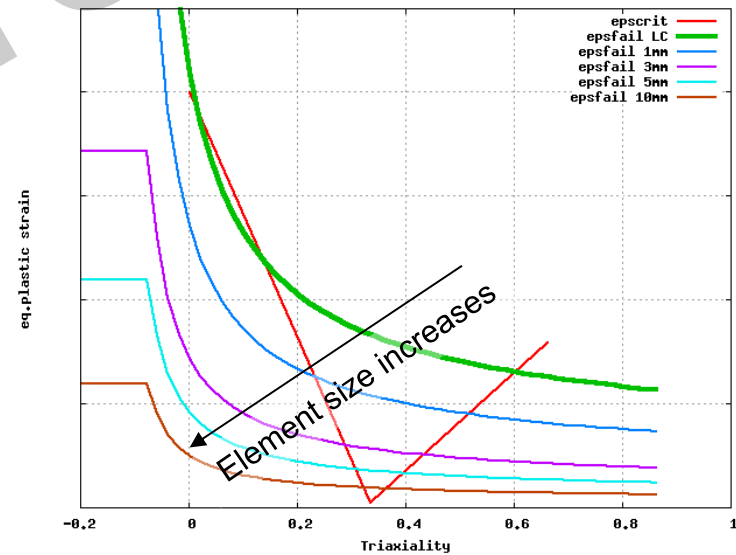
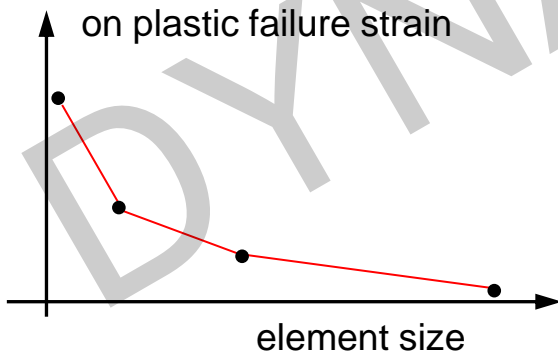
$      MID      EXCL      MXPRES      MNEPS      EFFEPS      VOLEPS      NUMFIP      NCS
      10
$      MNPRES      SIGP1      SIGVM      MXEPS      EPSSH      SIGTH      IMPULSE      FAILTM
$      IDAM      DMGTYP      LCSDBG      ECRIT      DMGEXP      DCRIT      FADEXP      LCREGD
      1          1          100        -200         2          2.5        400
$      SIZEFLG      REFSZ      NAHSV      LCSRS      REGSHR      REGBIAX
                          14          1.0        0.0
  
```

*DEFINE_CURVE

```

$      LCID      SIDR      SCLA      SCLO
      400        0          1.0        1.0
$
$      LE      regfac
      0.25    1.00
      0.50    0.80
      1.00    0.65
      2.50    0.52
      5.00    0.40
      10.00   0.31
  
```

scaling factor
on plastic failure strain



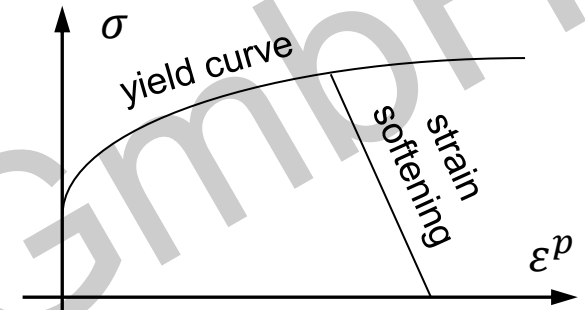
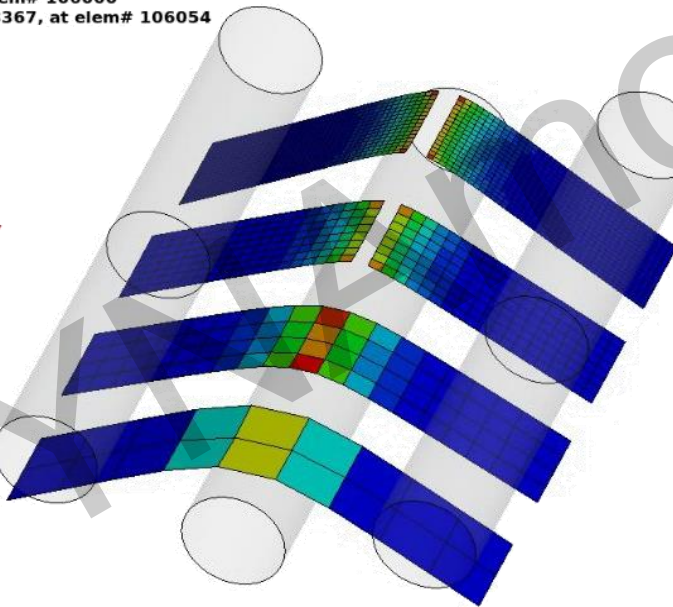
Mesh dependence

Bending example – without regularization

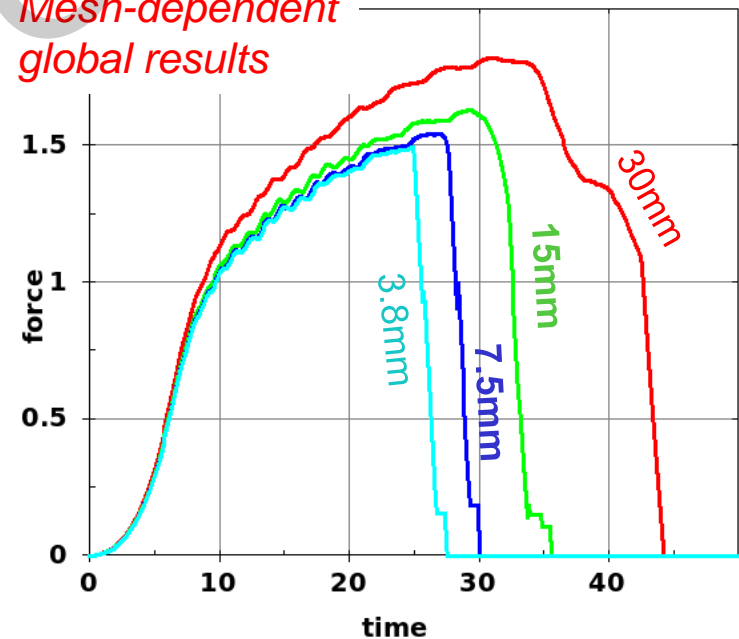
- *MAT_ADD_EROSION with IDAM=1, LCSDG with const. value of 0.055 and ECRIT=0.045
- 3-point bending with different mesh sizes
- Mesh dependent results **without regularization**:

Time = 30.5
Contours of Effective Plastic Strain
outer shell surface
min=0, at elem# 106000
max=0.0428367, at elem# 106054

localization
in one
element row



Mesh-dependent
global results



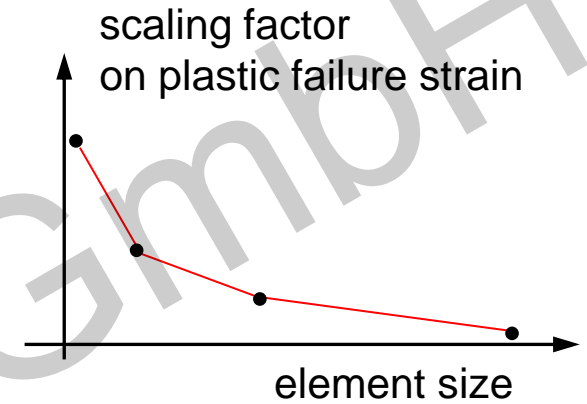
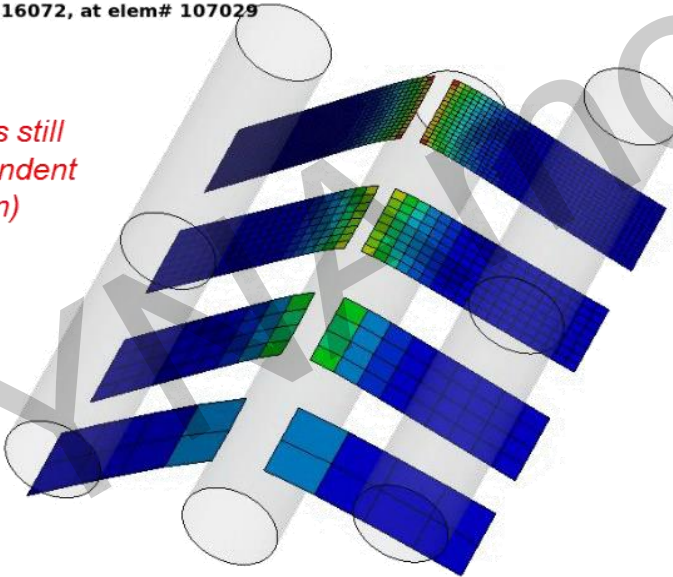
Mesh dependence

Bending example – with regularization

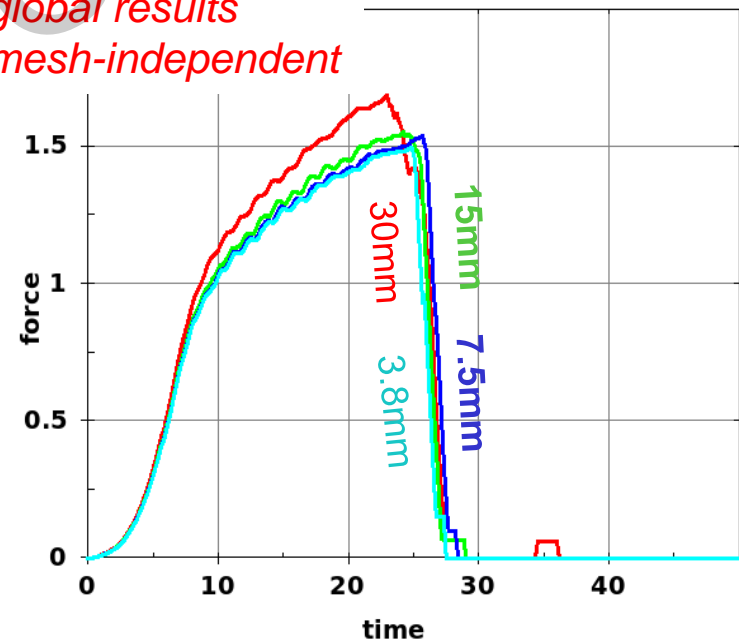
- *MAT_ADD_EROSION with IDAM=1, LCSDG with const. value of 0.055 and ECRIT=0.045
- 3-point bending with different mesh sizes
- Results **with regularization (LCREGD)**:

Time = 30.499
Contours of Effective Plastic Strain
outer shell surface
min=0, at elem# 106000
max=0.0416072, at elem# 107029

*local results still
mesh dependent
(localization)*



*global results
mesh-independent*

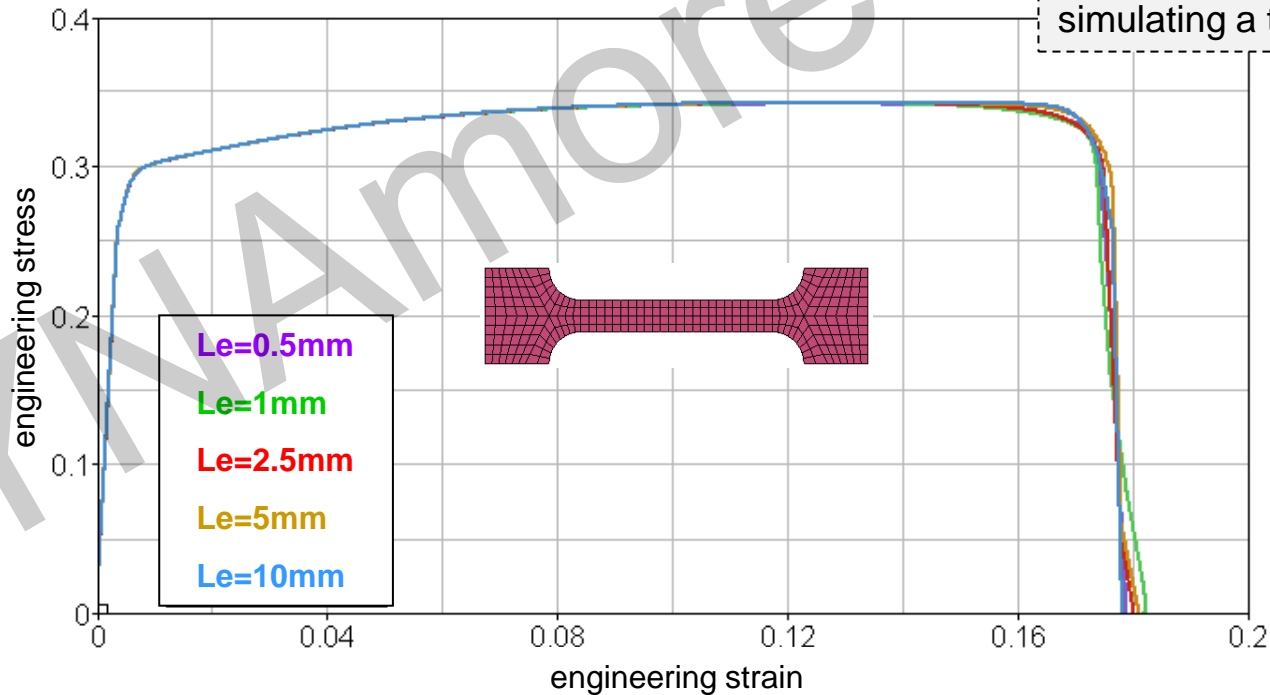


Calibration of GISSMO

Regularization

```
*DEFINE_CURVE
$ Regularizing factors (LCREGD)
$   LCID      SIDR      SCLA      SCLO
$         400         0         1.0         1.0
$
$           eta           epsf
$         0.50          1.00
$         1.00          0.72
$         2.50          0.49
$         5.00          0.35
$        10.00          0.28
```

A GISSMO material card can be iteratively regularized by simulating a tensile test!



Summary

Summary

Final comments

Although the subject of metal failure prediction seems to be far from settled, LS-DYNA currently provides state-of-the-art failure and damage models for the prediction of material ductile fracture. GISSMO belongs to the most advanced of these models and is generally recommended for metal failure prediction in LS-DYNA.

General features of GISSMO:

- Modular structure
- Dependence of the stress state
- Failure as a function of the triaxiality as well as of the Lode angle (shells and volume elements)
- Non-proportional loading is considered through damage accumulation
- Coupling of damage/stress → more realistic strains for large elements
- Numerical tools for treatment of mesh dependence
- Strain rate may be considered
- Possibility of mapping damage from a previous simulation

Summary

Where to get more information

- Presentations at this conference, e.g.:
 - F. Andrade, M. Feucht, A. Haufe:
*On the Prediction of Material Failure in LS-DYNA:
A Comparison Between GISSMO and DIEM*
(Tuesday, 2:50 pm, Crash I – Materials)
- DYNAmore training courses (www.dynamore.de/en/training/seminars)
 - Modeling of metallic materials (10th–11th Nov 2014, Stuttgart)
 - Material failure (12th–13th Nov 2014, Stuttgart)
- LS-DYNA upcoming conferences
 - 2015 European LS-DYNA Conference in Würzburg, Germany
- Papers from previous conferences (www.dynalook.com)

END