

# Recent Developments in LS-DYNA

LS-DYNA Forum  
October 7, 2014  
Bamberg



**LSTC**  
Livermore Software  
Technology Corp.

# Outline

- Introduction
- Developments
- Conclusions

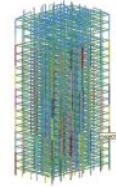
# LS-DYNA Application Areas

Development costs are spread across many industries



## Automotive

Crash and safety  
NVH  
Durability  
FSI



## Structural

Earthquake safety  
Concrete structures  
Homeland security



## Aerospace

Bird strike  
Containment  
Crash



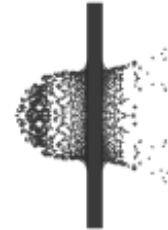
## Electronics

Drop analysis  
Package analysis  
Thermal



## Manufacturing

Stamping  
Forging



## Defense

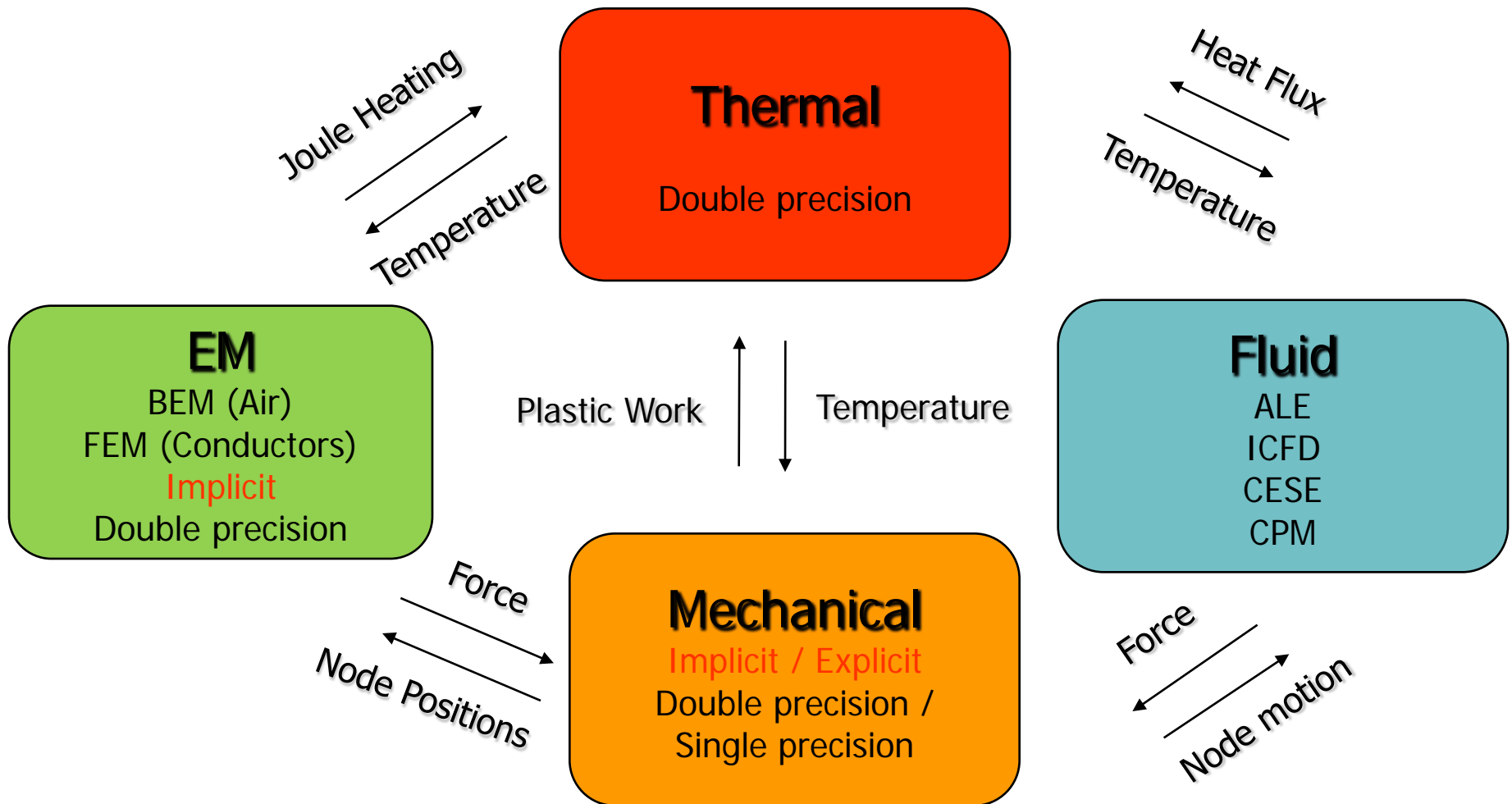
Weapons design  
Blast response  
Penetration  
Underwater Shock Analysis



## Consumer Products

# Accommodates Coupled Simulations

Multiple field equations are strongly coupled



# One Code for Multiple Solutions

One Model



LS-DYNA

Multi-physics

- Structure + Fluid + EM + Heat Transfer + , ..

Multi-stage

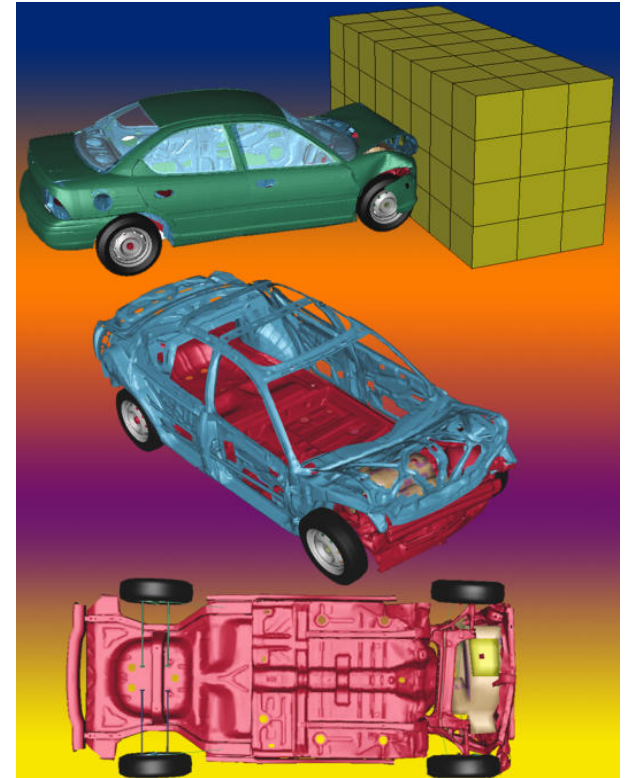
- Implicit + Explicit ....

Multi-scale

- Failure predictions, i.e., spot welds

Multi-formulations

- linear + nonlinear



## Many Results

Manufacturing, Durability, NVH, Crash

# One Model for All Applications

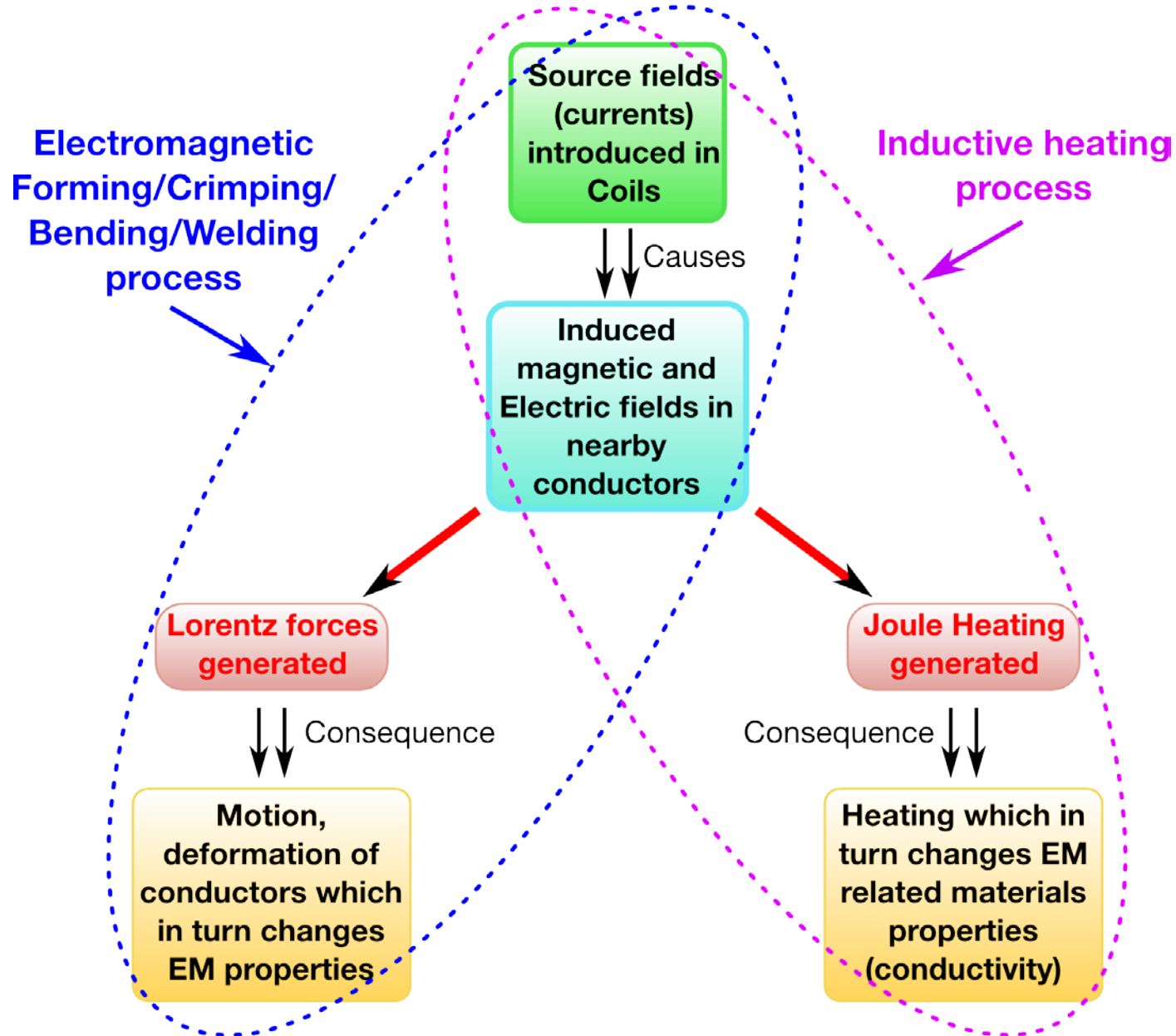
In crash, one structural model is used for frontal, side, offset, and rear impacts. The structural models used for durability, NVH, and FSI calculations are identical to the crash models with possible adjustments related to mesh density. The wide usage of the single model results in a model with fewer modeling errors:

- Only one model to revise for design changes.
- Only one model to check for errors.
- Multi-physics problems can be addressed.
- Easier database management

Initial stress, strain, and thickness distributions from manufacturing analysis are available for all simulations

# Electromagnetics

# EM Solver Introduction

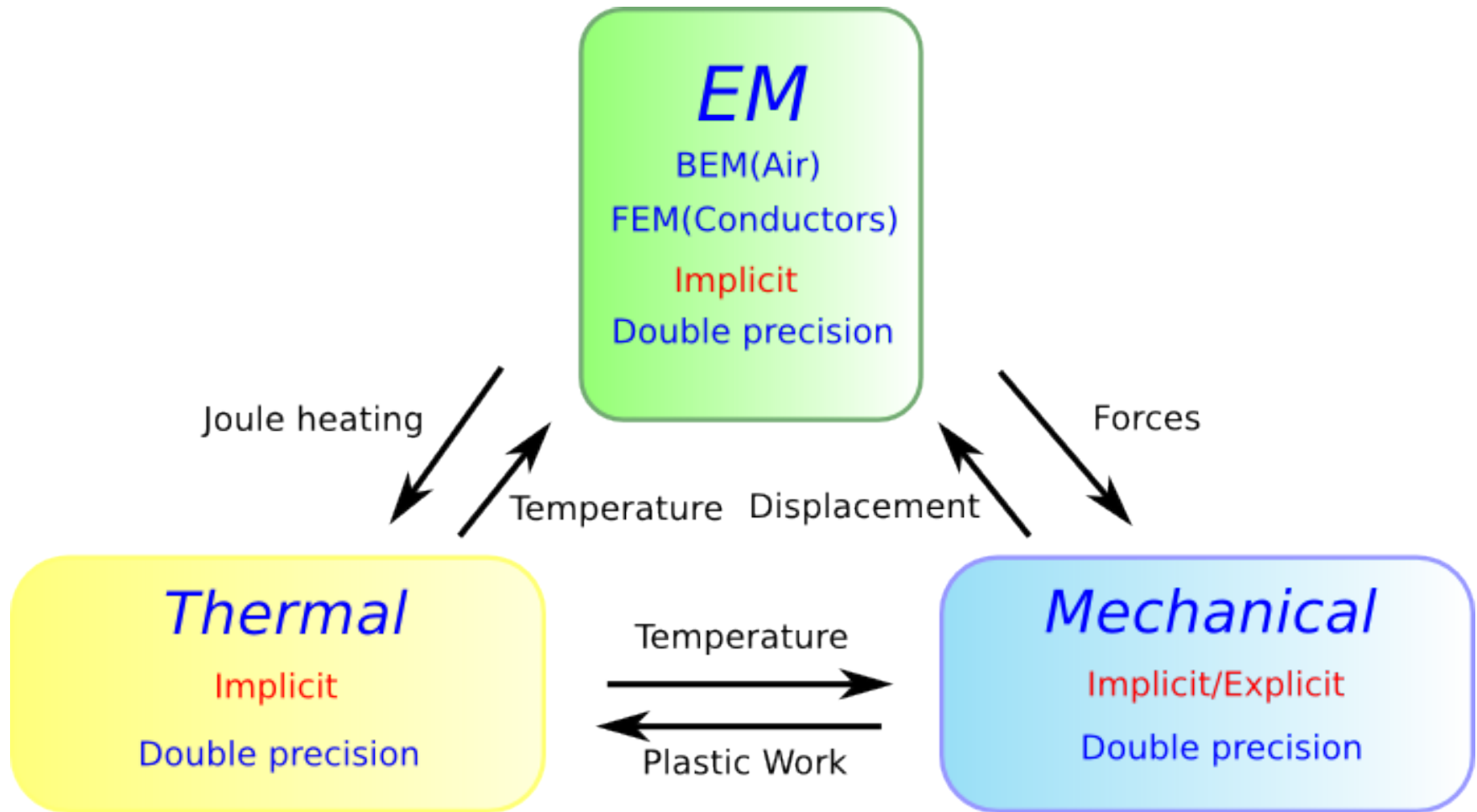




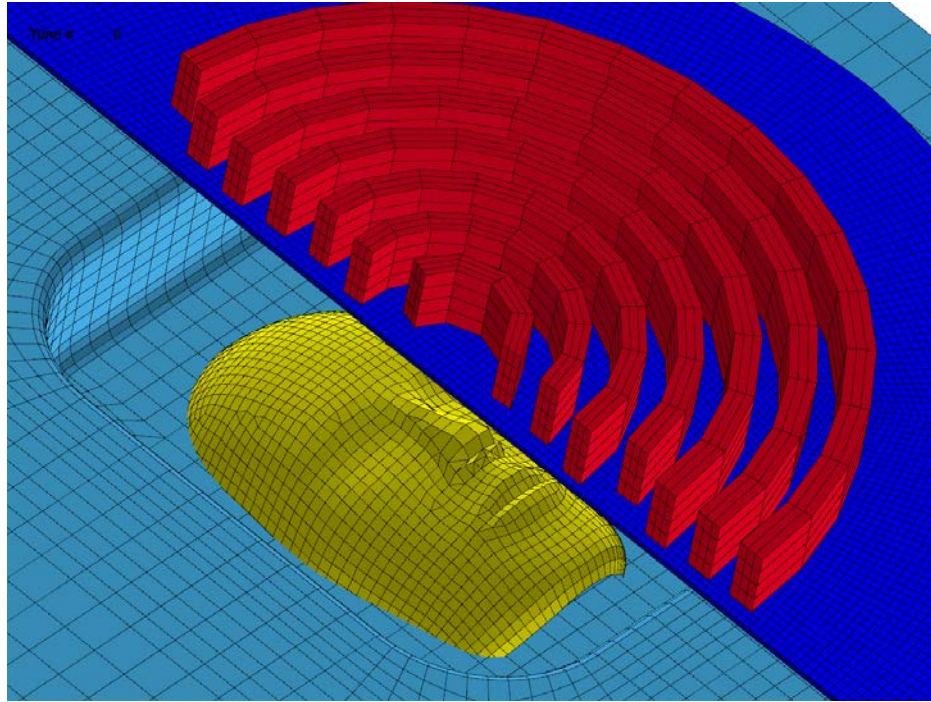
# EM Solver Introduction

- Fully implicit.
- Double precision.
- 2D axisymmetric solver /3D solver.
- SMP and MPP versions available.
- Automatically coupled with LS-DYNA solid and thermal solvers.
- FEM for conducting pieces, no air mesh needed (FEM-BEM method).
- New set of keywords starting with \*EM.

# EM Solver Introduction



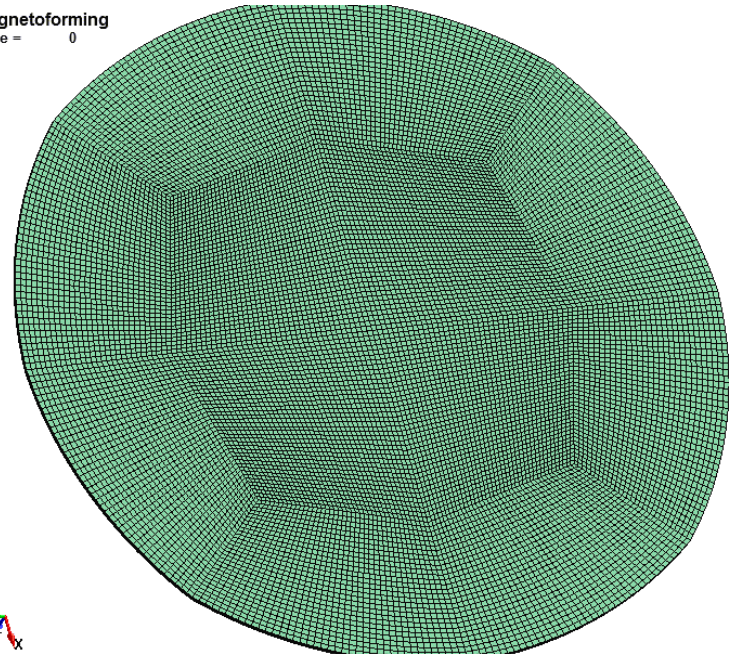
# EM Solver Introduction



## MMF: High velocity forming process

- Forming limits increased
- Springback reduced
- Wrinkling reduced
- High reproducibility

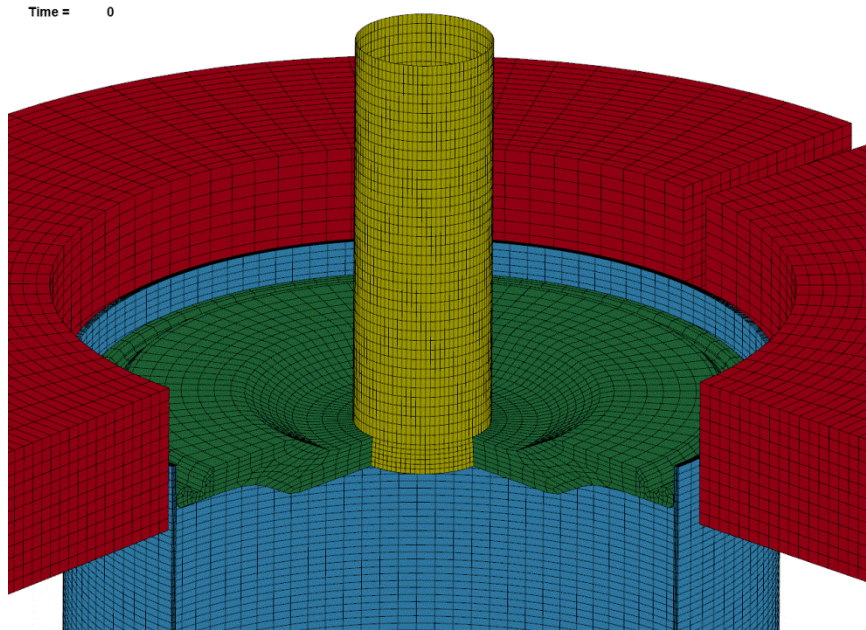
Magnetoforming  
Time = 0



In collaboration with:  
G. Mazars & G. Avrillaud:  
Bmax, Toulouse, France



# EM Solver Introduction



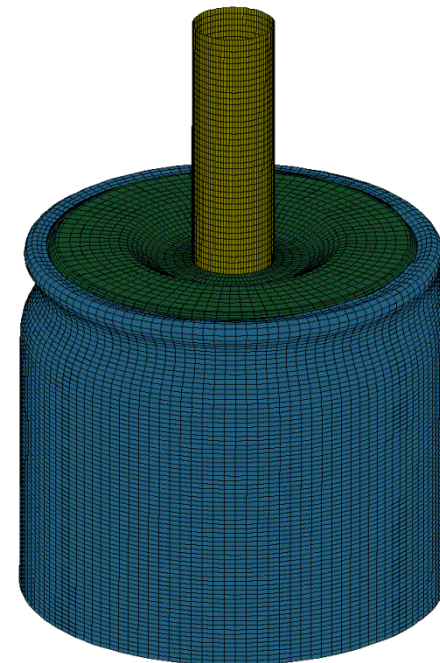
**Magneto crimping**

In collaboration with:  
G. Mazars and G. Avrillaud:  
Bmax, Toulouse, France



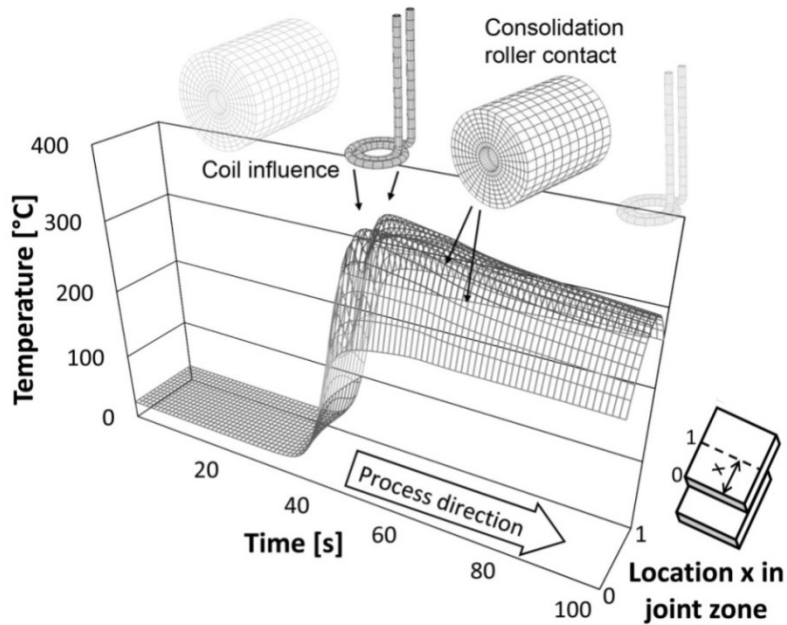
**Tensile test of the crimped parts  
(no EM)**

Time = 0

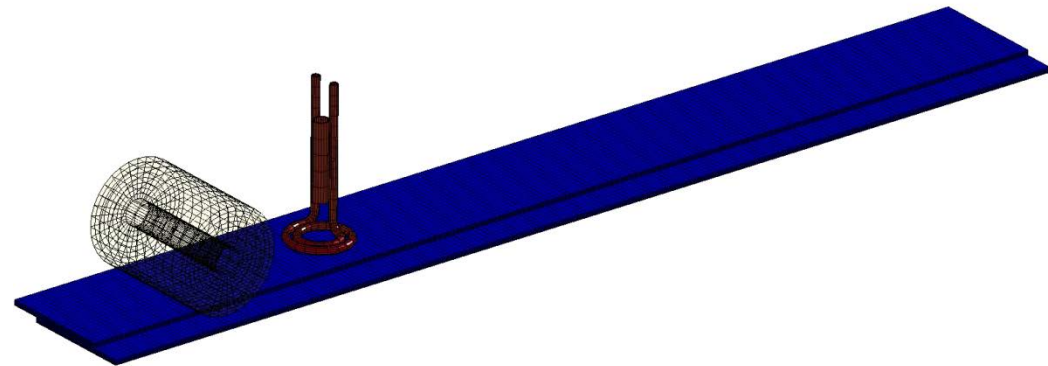


# EM Solver Introduction

Heating by Joule losses  
pressure applied for  
consolidation and maintained  
during cooling.

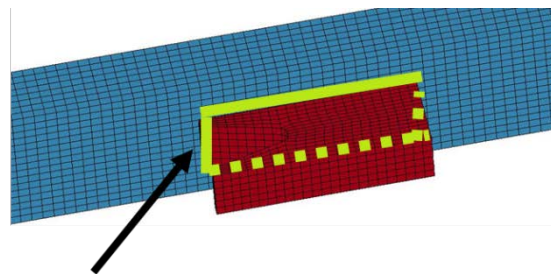
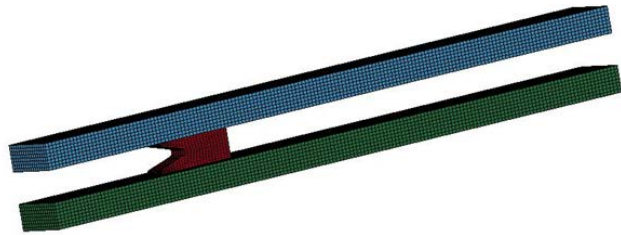
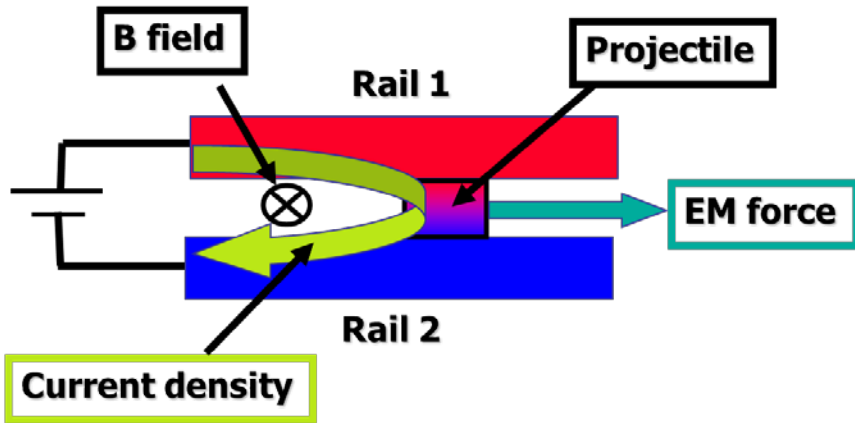


In collaboration with:  
M. Duhovic, Institut für Verbundwerkstoffe,  
Kaiserslautern, Germany

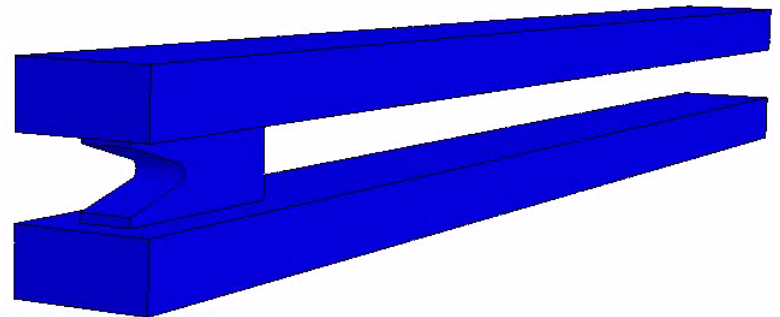
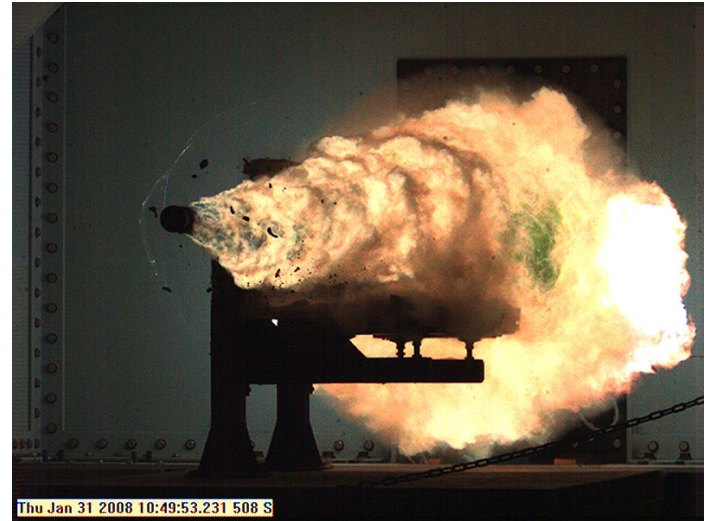


# EM Solver Introduction

## Rail gun simulation

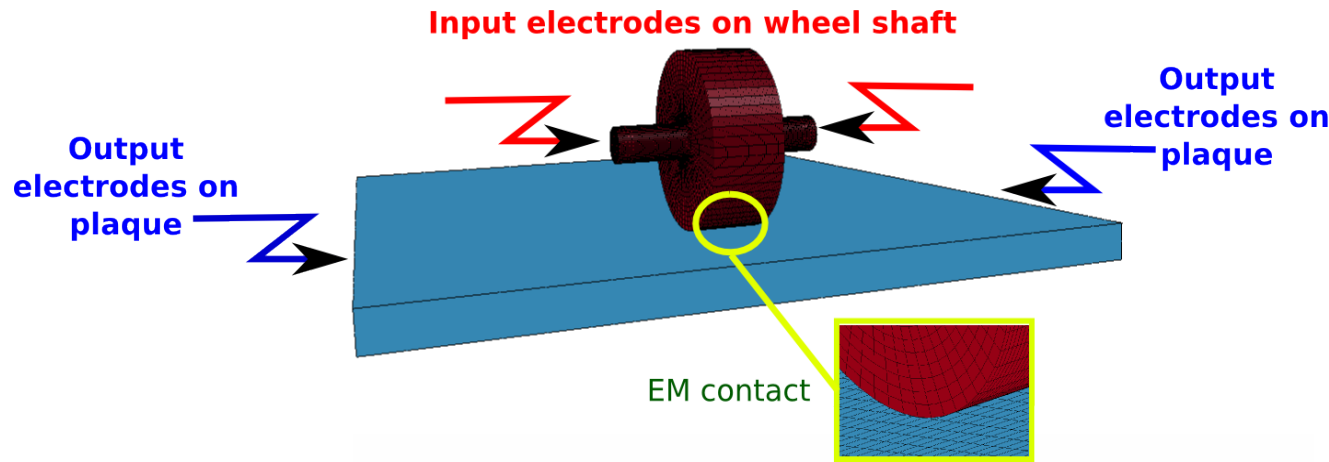


Sliding contact between the rails and the projectile



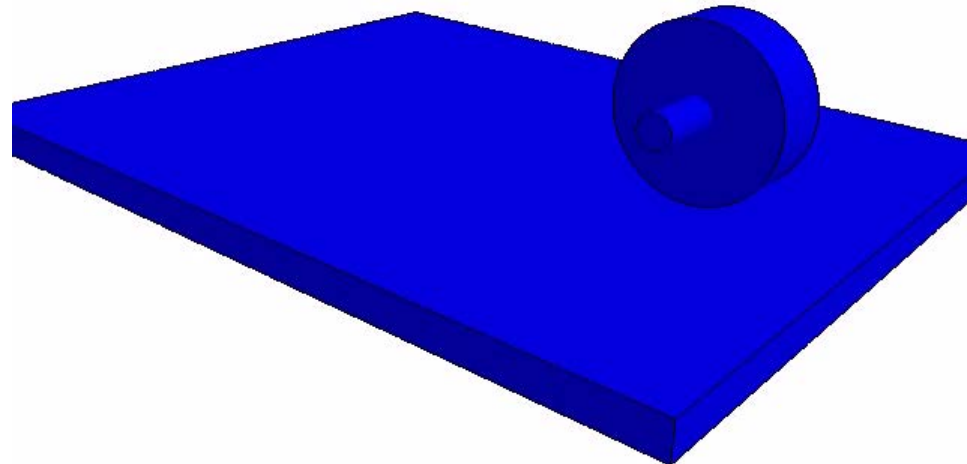
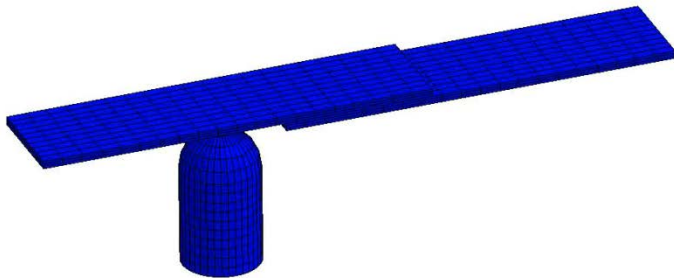
# EM Solver Introduction

- Joule heating study
- Study of shortcuts

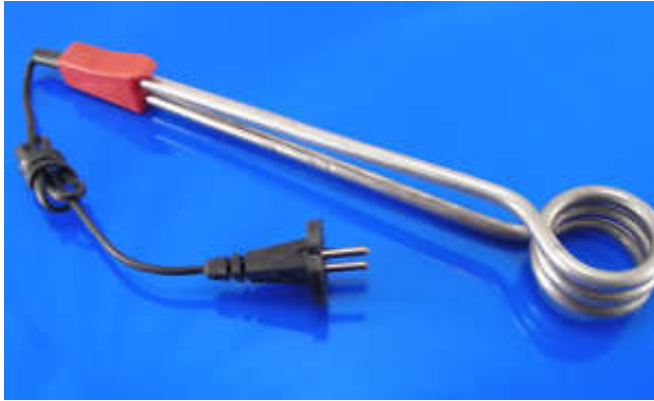


LS-DYNA keyword deck by LS-PrePost  
Time = 0  
Contours of Current density (magnitude)  
min=0, at node# 1  
max=0, at node# 1

Fringe Levels  
0.000e+00  
0.000e+00  
0.000e+00  
0.000e+00  
0.000e+00  
0.000e+00  
0.000e+00  
0.000e+00  
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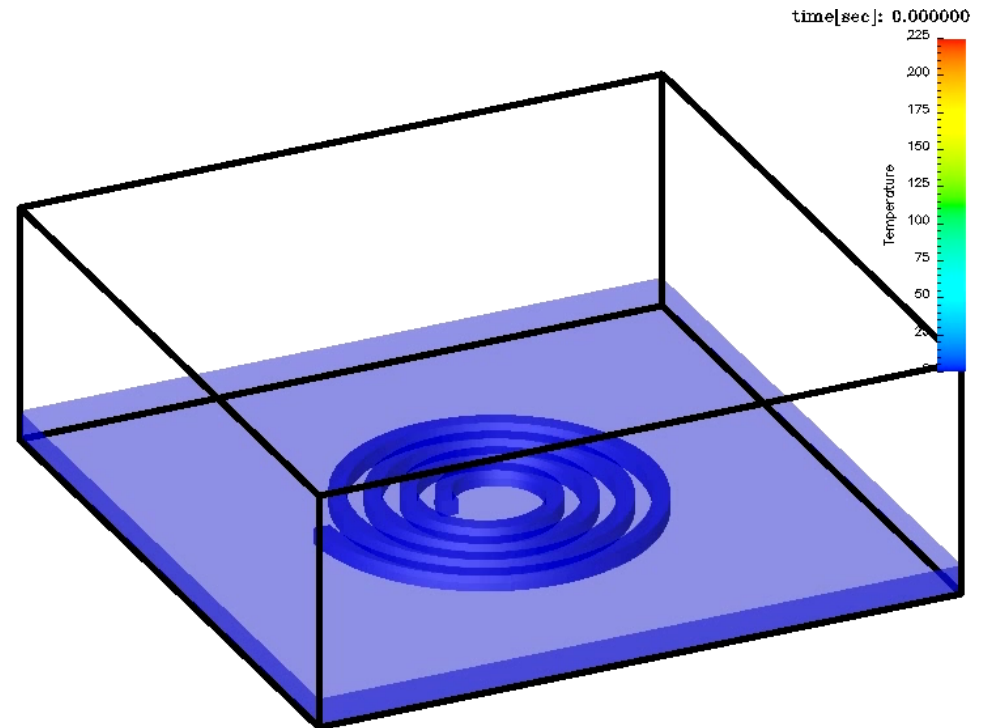


# EM Solver Introduction



- EM heats up a coil plunged in a kettle
- ICFD with conjugate heat transfer heats up the water

**Water stream lines colored by the temperature level.**





# EM Solver Developments

## Application: Conducting shells

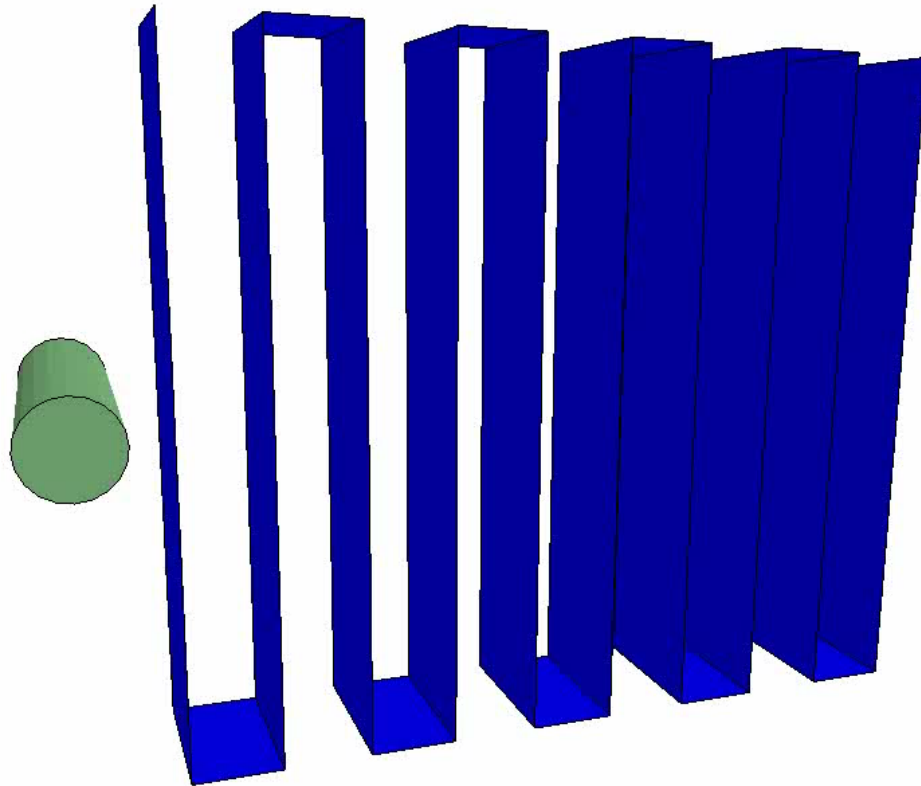
**Development :** Allowing user to define conductive shells in 3D problems.

**Advancement stage :** Conducting shells can be used in the development version with the Eddy Current, Inductive heating and Resistive heating solvers. EM contact also available. SMP and MPP versions fully functional.

- Eddy currents (the combination of inductive diffusive effects) is essentially a 3D phenomenon which means that elements with thickness are usually required for correct solution. Consequently, up to now, only solid conductive elements were allowed and shells could be insulators only.
- However, some users have expressed the wish to be able to use conductive shells in 3D problems in order to maintain their associated mechanical and thermal properties.
- From the EM perspective, those shells are treated like “invisible” solids i.e the EM solver will build an underlying equivalent solid mesh to solve for the EM quantities.

# EM Solver Developments

Time = 0



# EM Solver Developments

## Application: Resistive spot welding and others

**Development : Calculation of a additional resistance term and local Joule heating due to contact occurring between two conductors.**

- Resistive spot welding is a process in which contacting metal surfaces are joined by the heat obtained from resistance to electric current.
- Frequently encountered in the automobile manufacturing industry where it is used almost universally to weld metal sheets together (often automated process).
- Several formulas exist, often variation of Holm's law:

$$R_{contact} = \frac{\rho}{2a}$$

With  $\rho$  the material's resistivity and  $a$  the radius of the equivalent contact circle area.

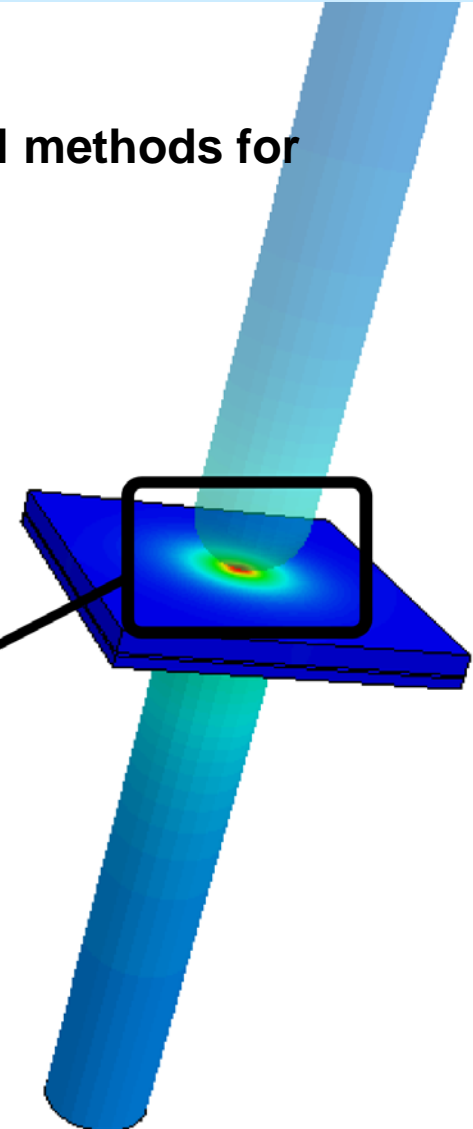
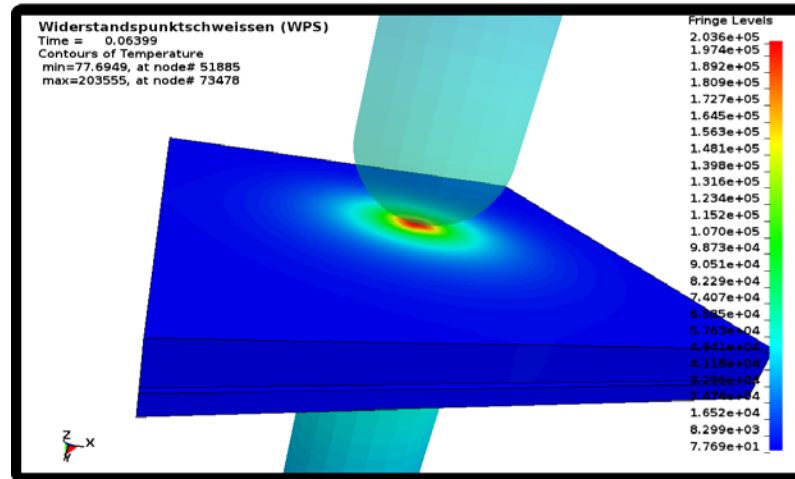
# EM Solver Developments

## Application: Resistive spot welding and others

Status : Available in the development version. Several methods for calculating contact Resistance available. See \*EM\_CONTACT\_RESISTANCE card.

Local heating spot between electrodes and work piece due to Contact Resistance

$$R_{contact} = \frac{\rho}{2a}$$



# EM Solver Developments

## Application: magnetic field lines

**Development : Have the solver compute some magnetic field lines in the air based on user defined parameters for analysis and post processing.**

- The use of the BEM method i.e no air mesh does not allow the visualization of the EM fields in the air hence the interest of this new feature.
- From a starting point given by the user the field line is computed step by step using a explicit numerical integration scheme (RK4, DOP853).
- Approximation methods are available in order to speed up the computation of the second member of the magnetic field line equations :
  - multipole method
  - “multicenter” method : this method has been developed at LSTC
- This feature along with the multicenter/multipole methods are treated as a research project that could be used at a latter stage to compute and store efficiently the BEM matrix and thus speed up the computational time of the EM solver.

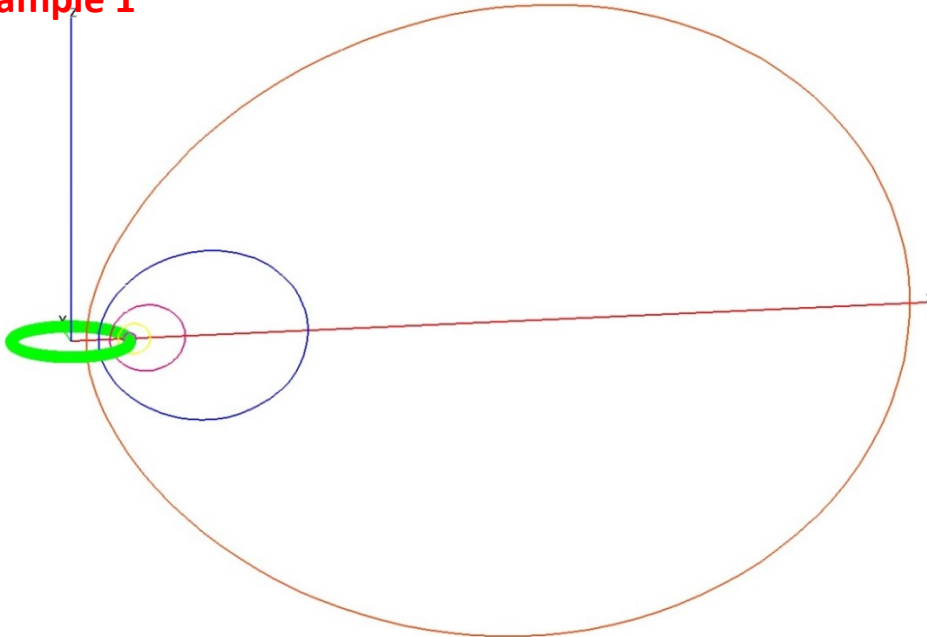
# EM Solver Developments

## Application: Magnetic field lines

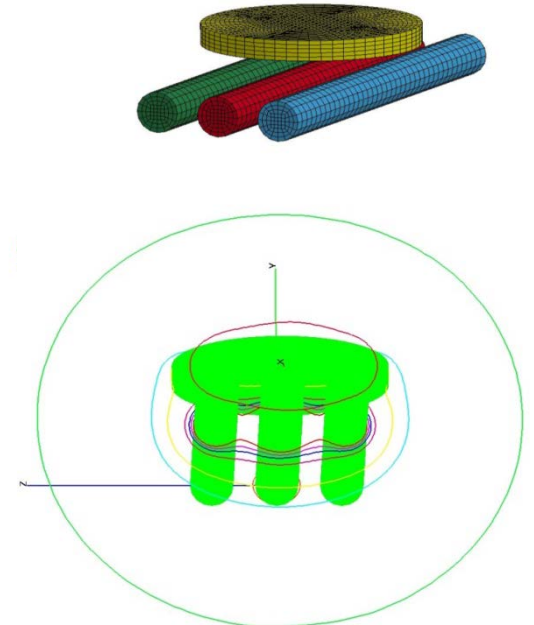
**Current status :** The magnetic field lines are now exported as individual Isprepost readable files at each output state, and will soon be integrated in the d3plot files. In a future development cycle, they will be automatically generated in Isprepost at any time without the user having to specify the output times before the run.

LS-DYNA keyword deck by LS-PrePost

Example 1



Example 2



# EM Solver Developments

## Future Developments

- Symmetry conditions
- Piezoelectric materials
- Magnetic materials

# Incompressible Fluid Dynamics

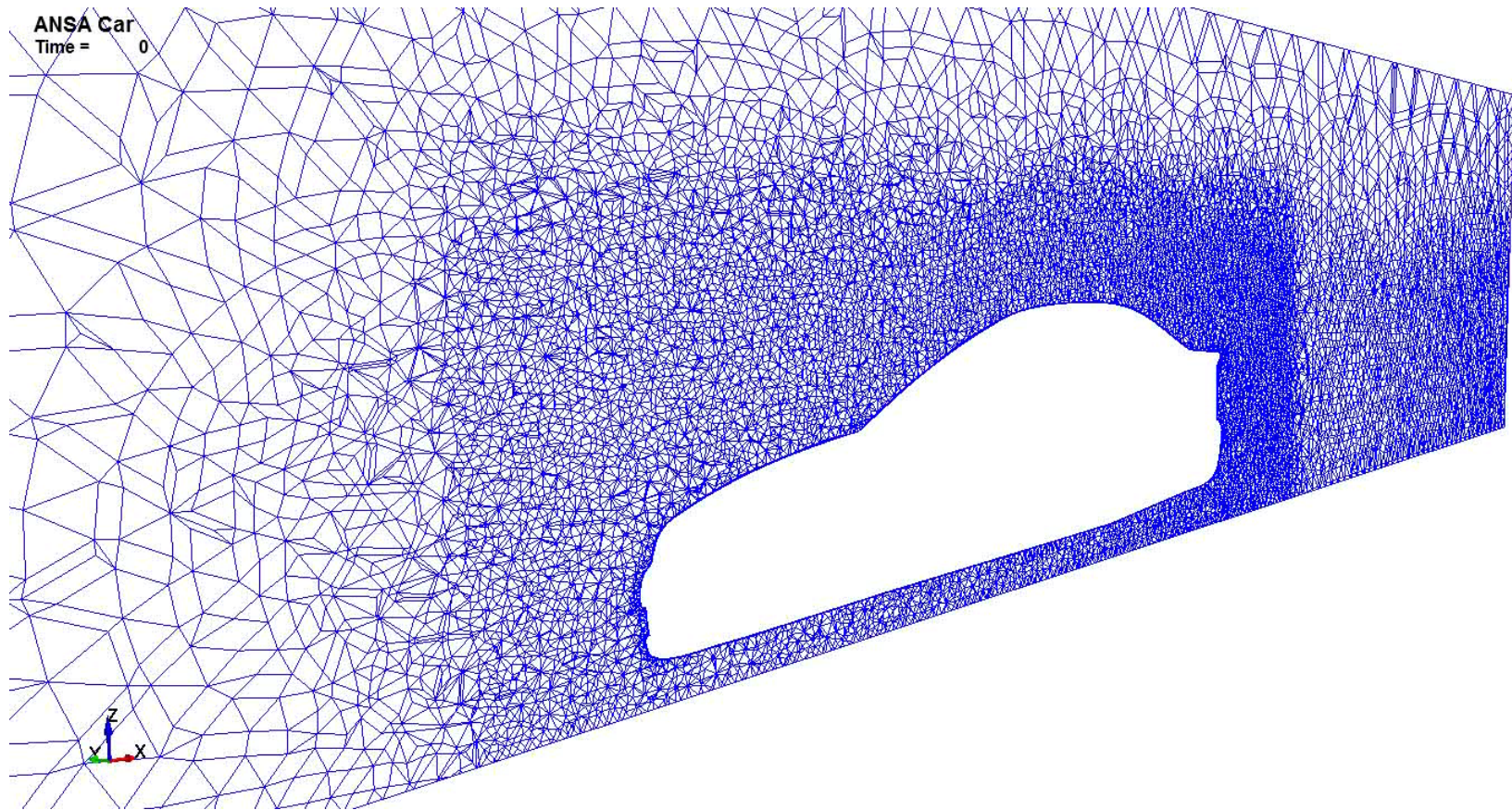


# ICFD Solver Introduction

- A CFD solver for incompressible flows (ICFD solver).
- Fully implicit.
- Double precision.
- SMP and MPP versions available. Highly scalable in MPP.
- New set of keywords starting with \*ICFD/\*MESH.
- Can run as a stand alone CFD solver.

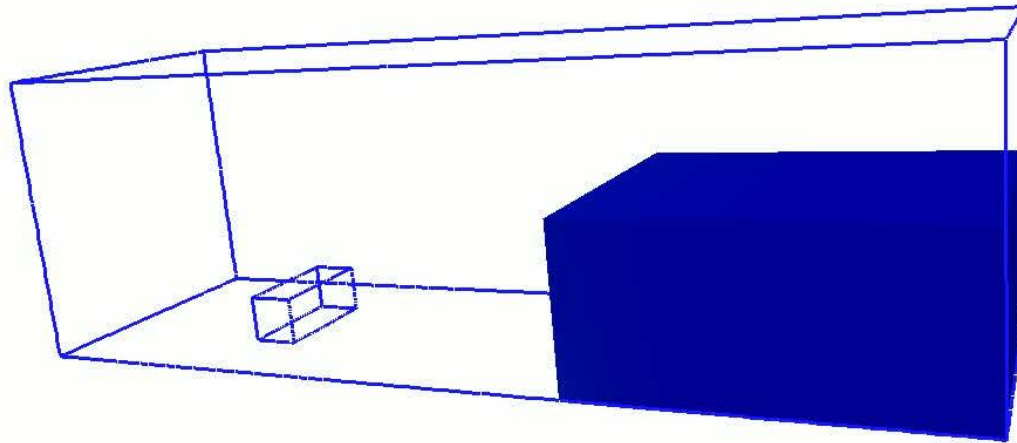
# ICFD Solver Introduction

## Aerodynamics: study of turbulent flow around a car



# ICFD Solver Introduction

**Wave impact: study of pressure forces on a body**

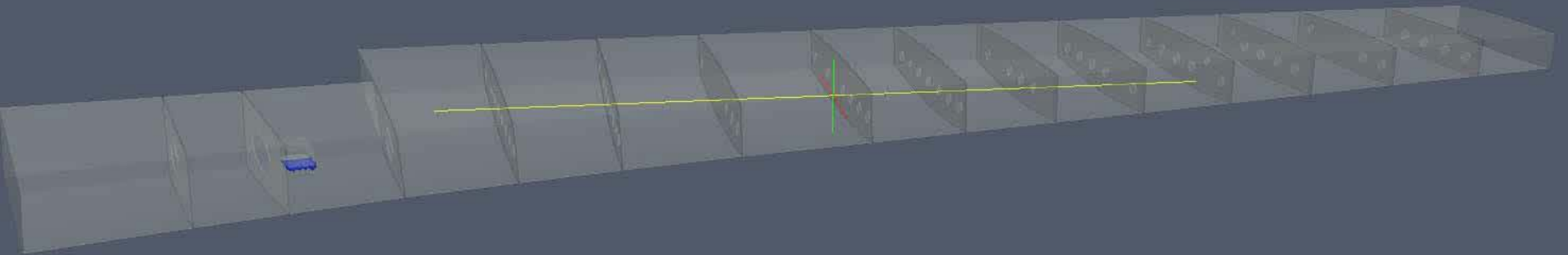


Time: 0.00 s

# ICFD Solver Introduction

**Aircraft wing filling: study of fuel repartition and filling time**

Time: 0.000 s



# ICFD Solver Introduction

## Thoughts about FSI

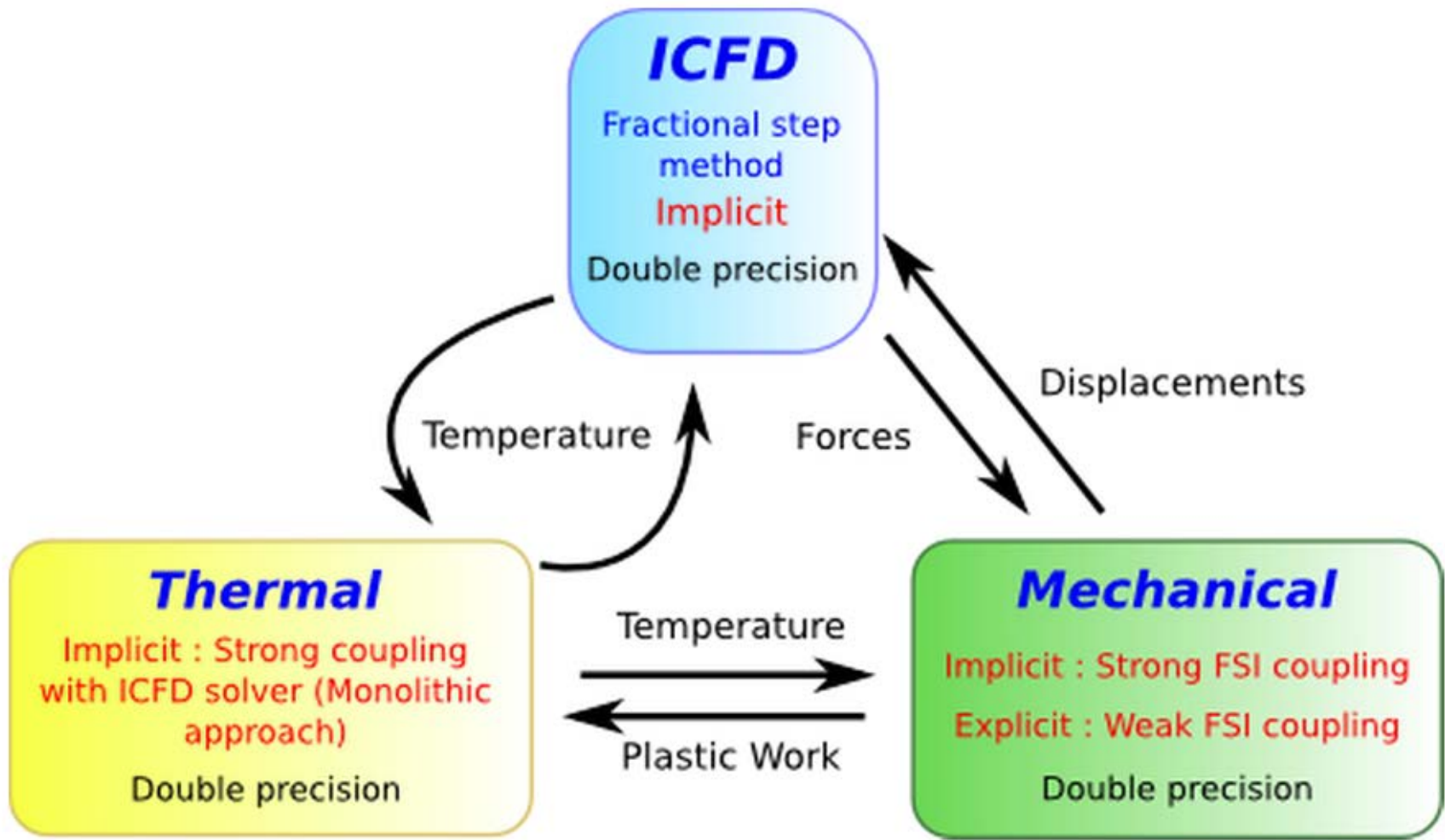
- A widely used approach in engineering is to assume that the FSI problem is linear and to use two different software products and licenses.
- The solid work group finds the need for FSI simulation.
- The geometry is sent to the fluid group which builds a mesh and runs the fluid problem with a CFD solver until it reaches steady state.
- The fluid stresses together with the mesh is brought back to the solid group which handles data with scripts to convert it into the input data for the solid solver.
- The solid solver performs a modal analysis.

# ICFD Solver Introduction

## The LS-DYNA approach

- LS-DYNA has immense solid mechanics capabilities as well as a huge material library. It can run both in explicit and implicit and already has a thermal solver for solids.
- LS-DYNA offers the perfect environment in order to develop a CFD solver allowing complex fluid structure interactions as well as the solving of conjugate heat transfer problems.
- The set up of the coupled problem is greatly simplified with only a few additional keywords necessary.
- On top of the classic “loose” or “explicit” FSI coupling, the ICFD solver offers a state of the art strong coupling method which opens up new applications.
- “All in one code” approach.

# ICFD Solver Introduction



# ICFD Solver Introduction

**Aerodynamics: highly non linear FSI problem  
with flap oscillating in the wind (*Turek benchmark problem*)**

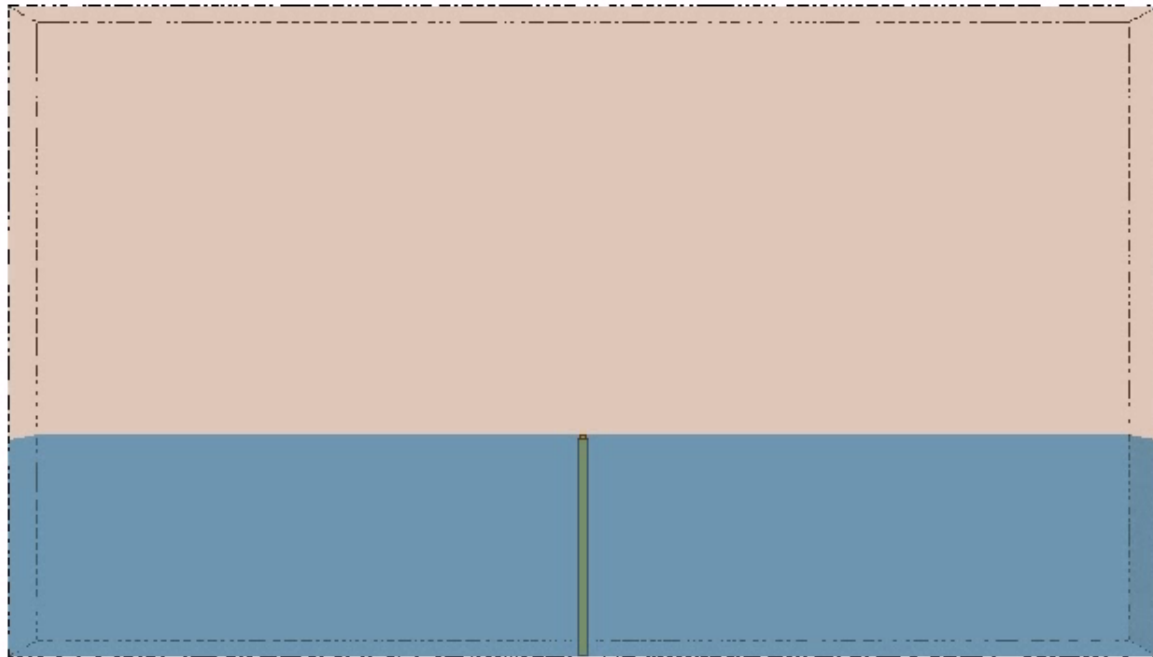




# ICFD Solver Introduction

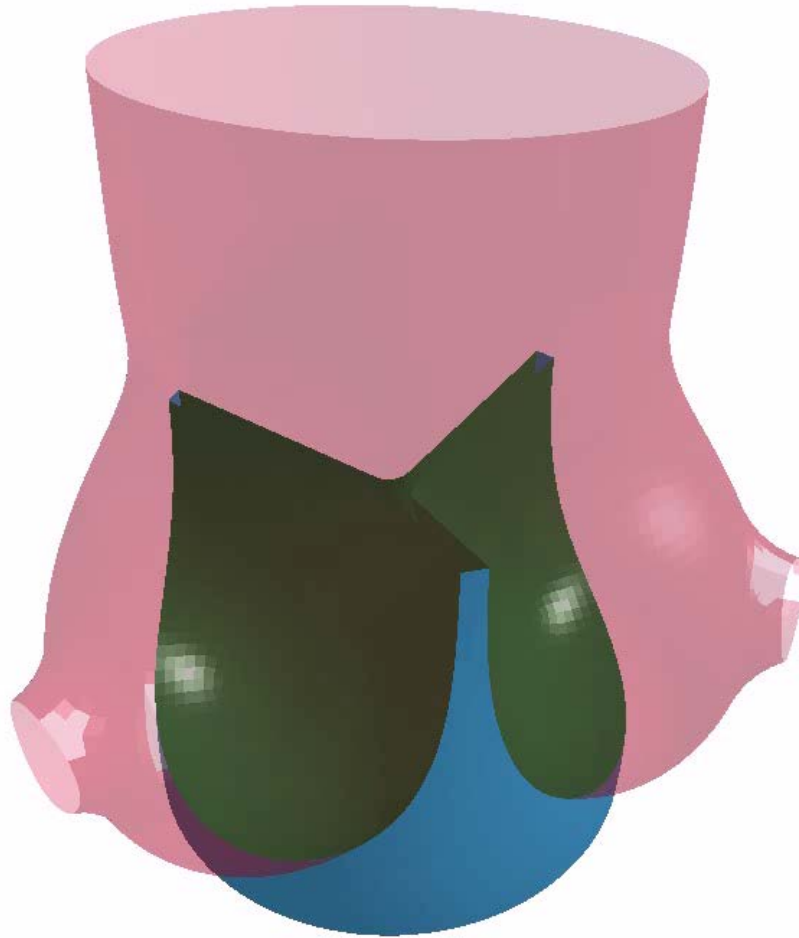
## Sloshing: Flap oscillating in partially filled tank

Time = 0, #nodes=2146228, #elem2d=274382, #elem3d=6403785



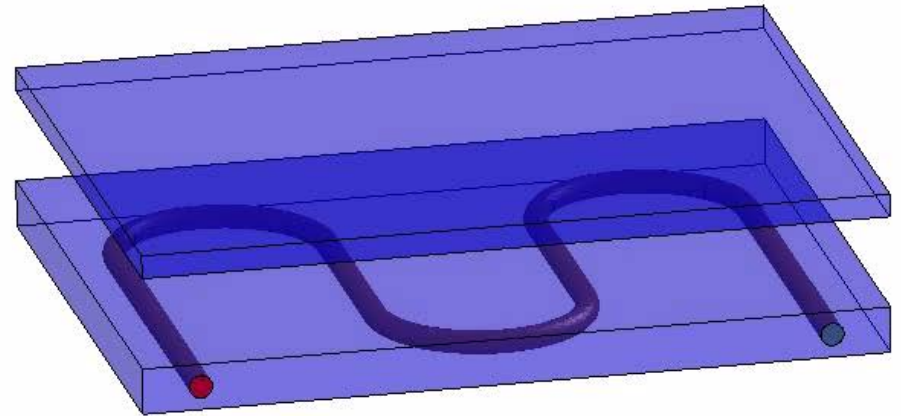
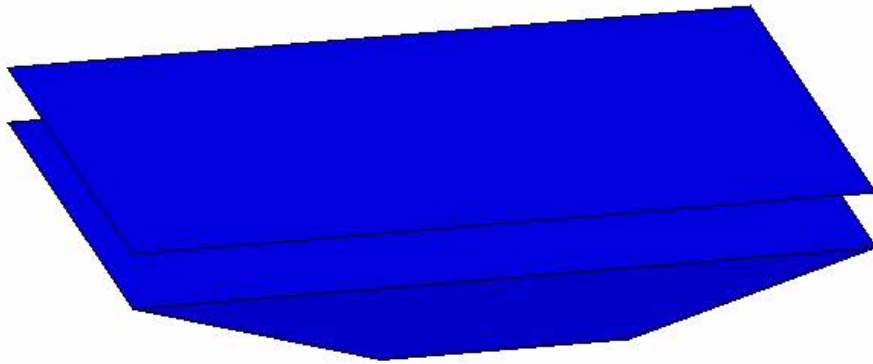
# ICFD Solver Introduction

**Artificial Heart valve: Strong pressure gradients force leaflets open**



# ICFD Solver Introduction

**Stamping and conjugate heat transfer:  
flow in serpentine pipe causes die to cool**



# ICFD Solver Developments

## Application: Multiphase problems

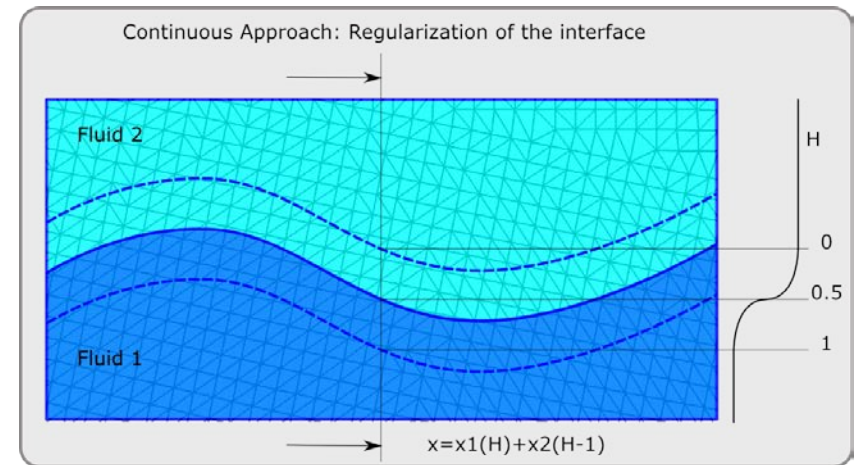
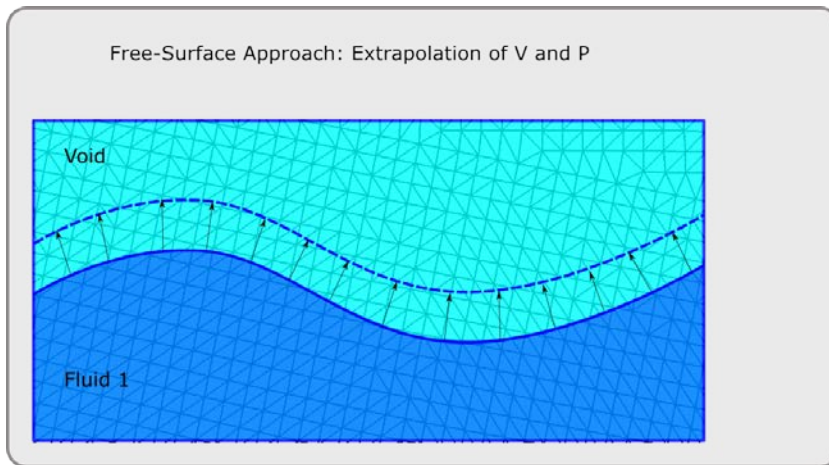
**Development : Being able to solve problems with two fluids of different densities (water+air).**

**Current Status : Implementation stage.**

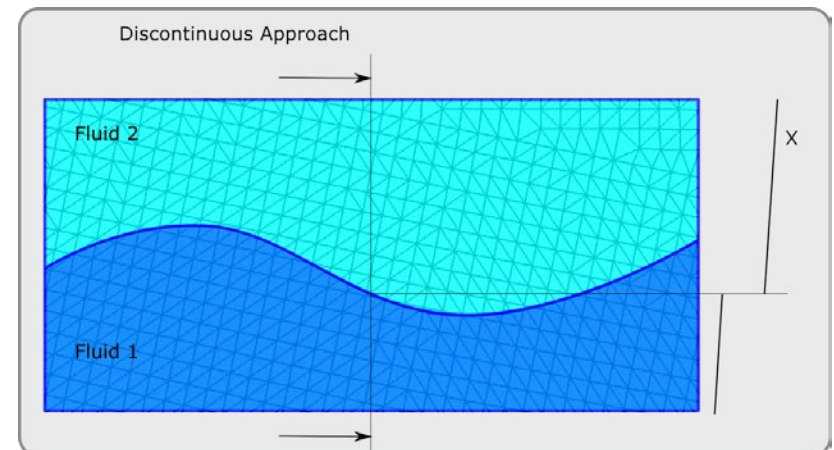
- Numerous applications such as lubrication, droplets, sloshing in closed tanks, etc.
- Two immiscible phases.
- Level Set approach for interface tracking.
- Continuous and discontinuous approach to model pressure jumps (surface tension).
- When the inertial effects of the second fluid can be simplified, the Free surface approach can be used.

# ICFD Solver Developments

## Application: Multiphase problems



- Free surface approach: suitable for problems where the inertial effects of the lighter fluid may be neglected.
- Continuous approach: works in most multiphase problems.
- Discontinuous approach: used in problems where surface tension effects are important.



# ICFD Solver Developments

## Application: Porous Materials

**Development :** Implementation of a generalization of the Navier Stokes equations that will allow the definition of sub-domains with different permeability/porosity.

**Current Status :** Validation stage. Available in the development version. See 4<sup>th</sup> card of ICFD\_MAT.

Being  $\varepsilon$ ,  $\kappa$  the porosity and the permeability of the medium respectively :

$$\varepsilon = \frac{\text{void volume}}{\text{total volume}}$$

$$\frac{\rho}{\varepsilon} \left[ \frac{\partial u_i}{\partial t} + \frac{\partial \left( \frac{u_i u_j}{\varepsilon} \right)}{\partial x_j} \right] = - \frac{1}{\varepsilon} \frac{\partial (P \varepsilon)}{\partial x_i} + \frac{\mu}{\varepsilon} \frac{\partial^2 u_i}{\partial x_i^2} + \rho g_i - D_i$$

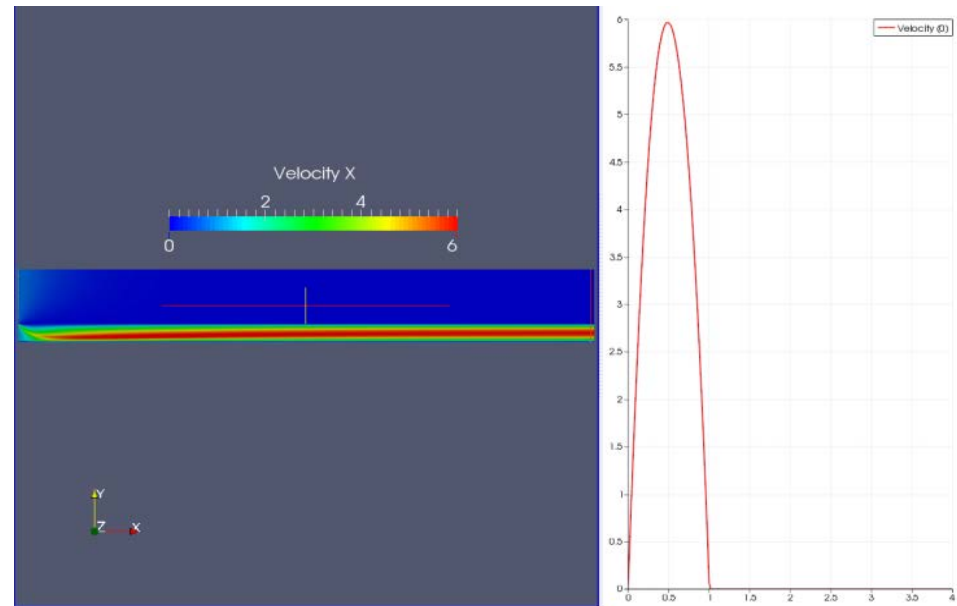
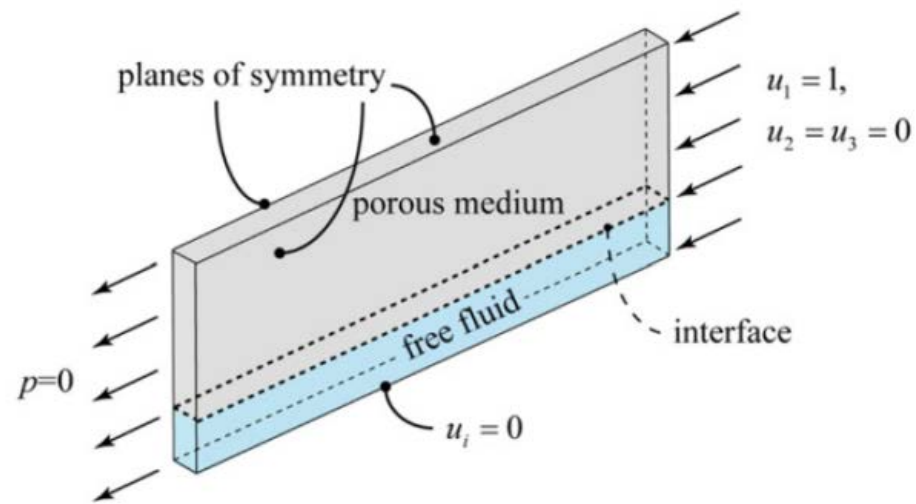
**Ergun correlation :**

$$D_i = - \frac{\mu U_i}{\kappa} + \frac{1,75 \rho |u|}{\sqrt{150} \sqrt{\kappa} \varepsilon^{1.5}} U_i$$

# ICFD Solver Developments

## Application: Porous Materials

**Validation stage: analysis of references cases involving porous and fluid domains. Study of FEM solution and analytical/reference solutions**



# ICFD Solver Developments

## Application: Thermal problems

**Development :** Calculation of the convection coefficient “h” based on a rigorous approach for the estimation of the bulk temperature  $T_m$ .

**Current Status :** Available in the Development version.

$$h = \frac{q}{T_s - T_m}$$

With  $q$  the heat flux,  $T_s$  the temperature at the surface and  $T_m$  the “bulk temperature”

- Frequently used by engineers in cooling applications in order to approximate the effect of the fluid cooling on the structure (See \*BOUNDARY\_CONVECTION).
- The  $h$  can be found in empirical tables based on the fluid properties and the geometry of the pipe.
- However, for complex cases and geometries, it may be useful to run the CFD problem in order to check the value of the  $h$  along the pipe and to look for potential zones or pipe bents where the cooling becomes less or too efficient.



# ICFD Solver Developments

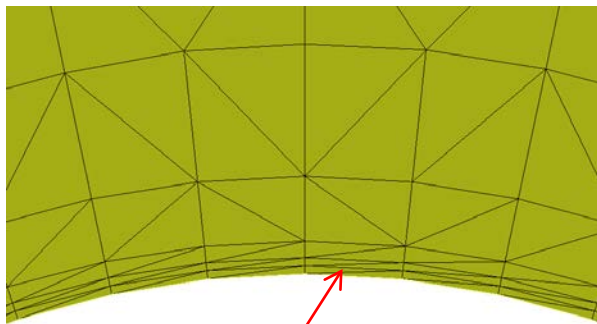
## Application: External and internal aerodynamics with turbulence

Development: Adding more HRN and LRN laws of the wall for the turbulence models. Providing more tools for the boundary layer mesh generation.

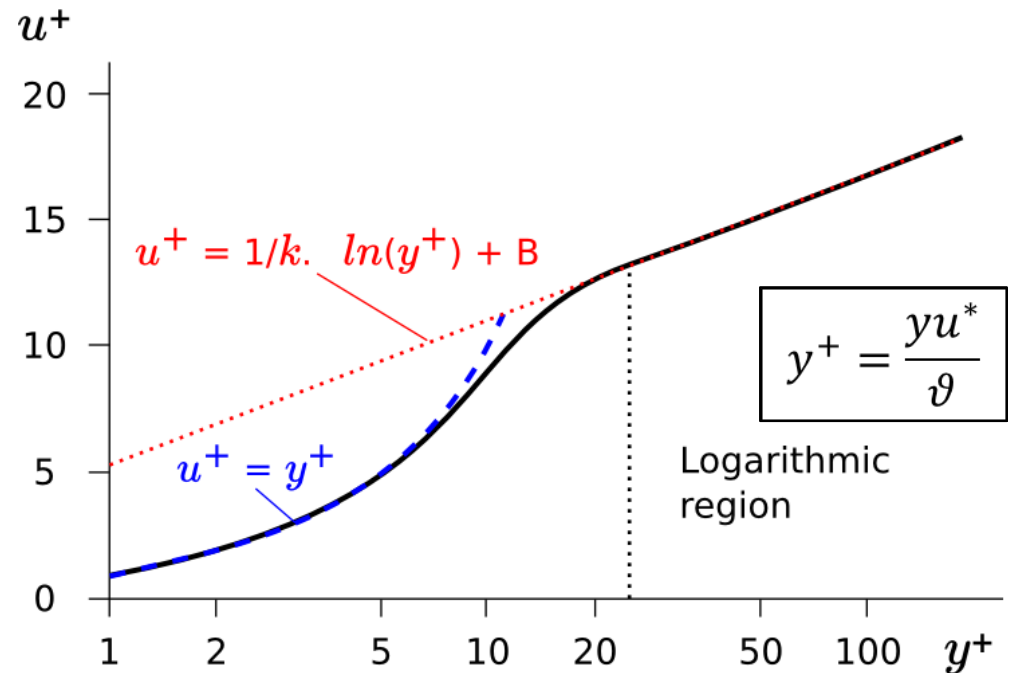
Current status : Implementation stage

$$u^* = \sqrt{\frac{\tau_w}{\rho}}$$

$$u^+ = \frac{U}{u^*}$$



BL mesh



# ICFD Solver Developments

## Future Developments

- A wave generator for free surface problems.
- Porous media with FSI problems.
- Specific porous media models for parachutes.
- Adaptive surface remeshing.
- Embedded approach for FSI problems.

# Rotational Dynamics

# Rotational Dynamics for Implicit

**Rotational dynamics:** the study of vibration of rotating parts (turbine blades, propellers in aircraft and rotating disks in hard disk drives etc.).

The **deformation of rotating components** can cause damage in machines with rapidly rotating parts.

The **resonant vibration** can lead to premature fatigue failure in those rotating components, as well as bearings and support structures.

The **goal of the Rotational Dynamics in LS-DYNA is to study the above vibration-related phenomenon by considering spin softening and gyroscopic effects .**

# Implementation in LS-DYNA

## \*CONTROL\_IMPLICIT\_ROTATIONAL\_DYNAMICS

Variable	sid	stype	omega	vid	nomega	iref		
Type	I	I	F	I	I	I		
Variable	omeg1	omeg2	omeg3	omeg4	omeg5	omeg6	omeg7	omeg8
Type	F	F	F	F	F	F	F	F

**sid:** Set ID of the rotational structure.

**stype:** Set type, 0: part;  
1: part set.

**omega:** Rotating speed.

**vid:** Vector ID to define the rotating axis.  
It is defined in \*DEFINE\_VECTOR.

**nomega:** No. of rotating speeds for Campbell diagram plotting (an example will be given on the next slide).

**iref:** Reference frame. 0: rotating coordinate system;  
1: fixed coordinate system.

**omegi:** The ith rotating speed.

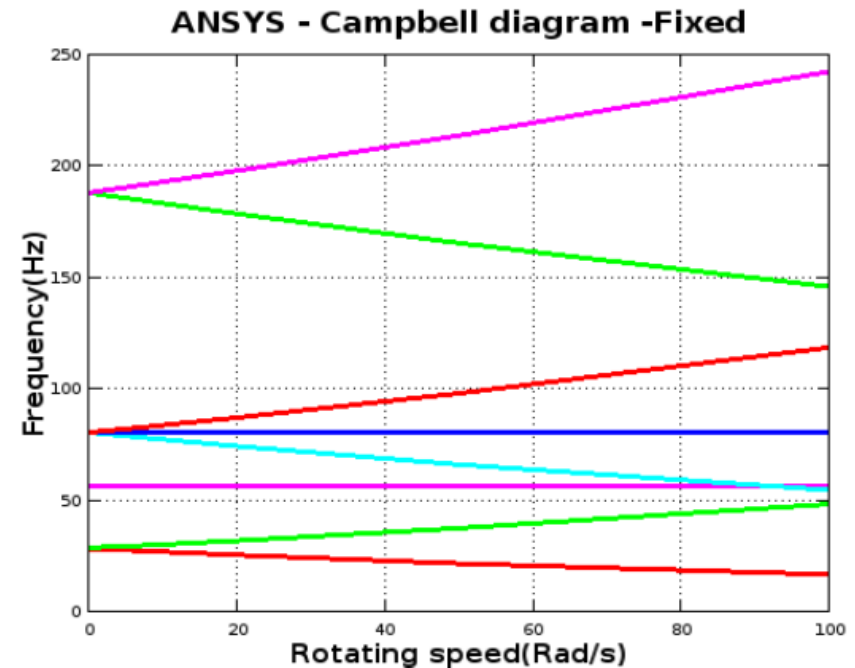
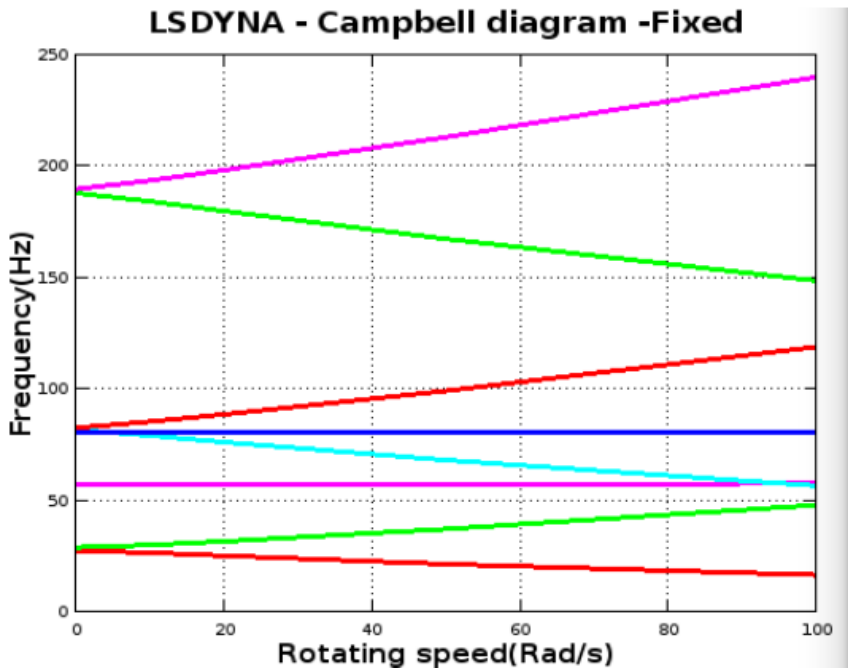
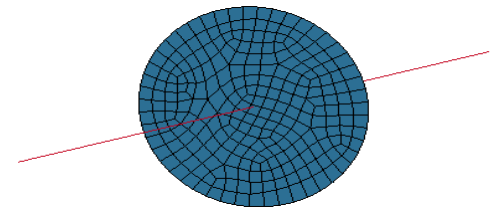
You can define as many cards as the number of rotating sets.

# Modal Analysis Example

**Campbell Diagram:** represents a system's eigen-frequencies as a function of rotating speeds.

**Study case** (on the right): a shaft and a disk, rotating along the shaft axis.

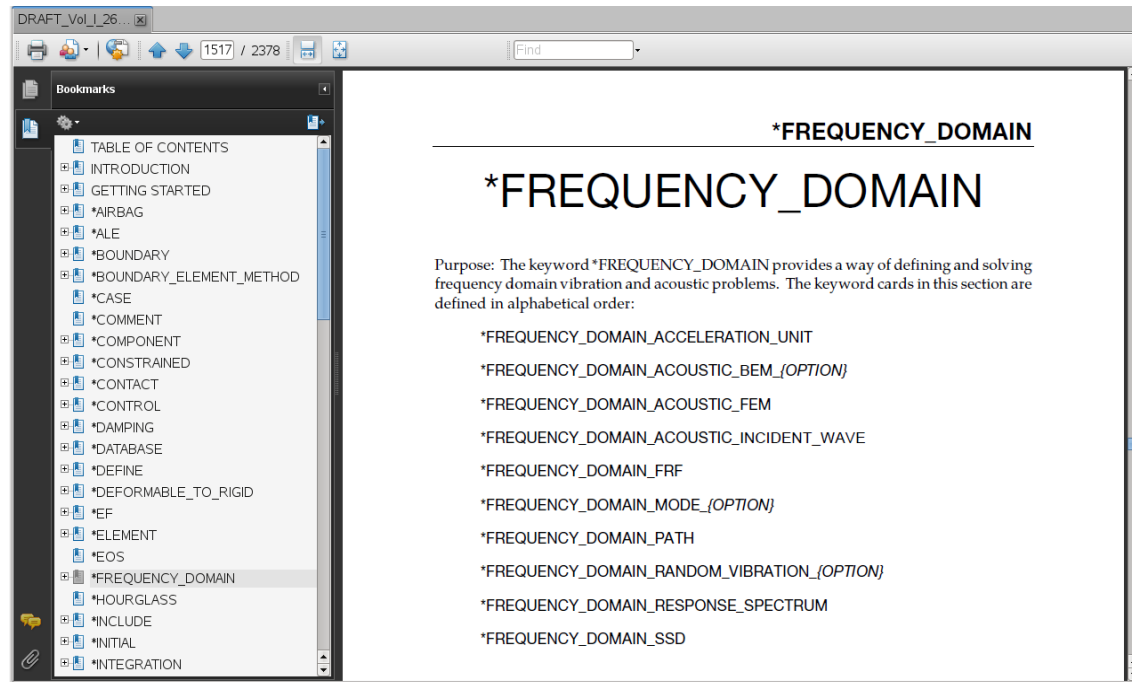
**The following two plots:** give the Campbell diagrams obtained from LS-DYNA and ANSYS.



# Frequency Domain

# Frequency domain features

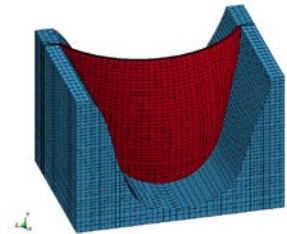
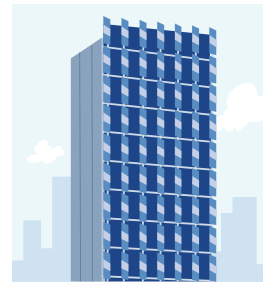
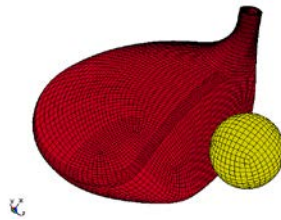
- FRF
- SSD
- Random vibration
- Random fatigue
- Response spectrum analysis
- BEM Acoustics
- FEM Acoustics





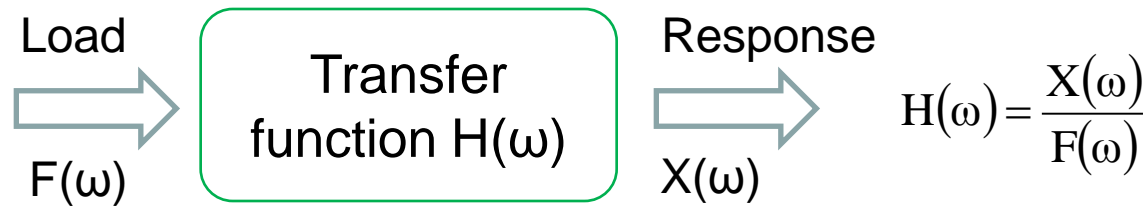
# Application

- NVH of automobiles and aircraft
- Acoustic design and analysis
- Defense industry
- Fatigue of machines and engines
- Civil and hydraulic engineering
- Earthquake engineering

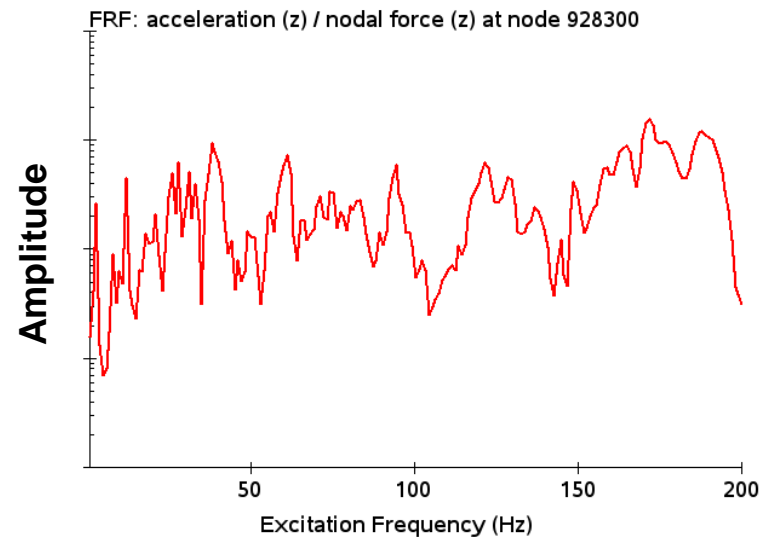
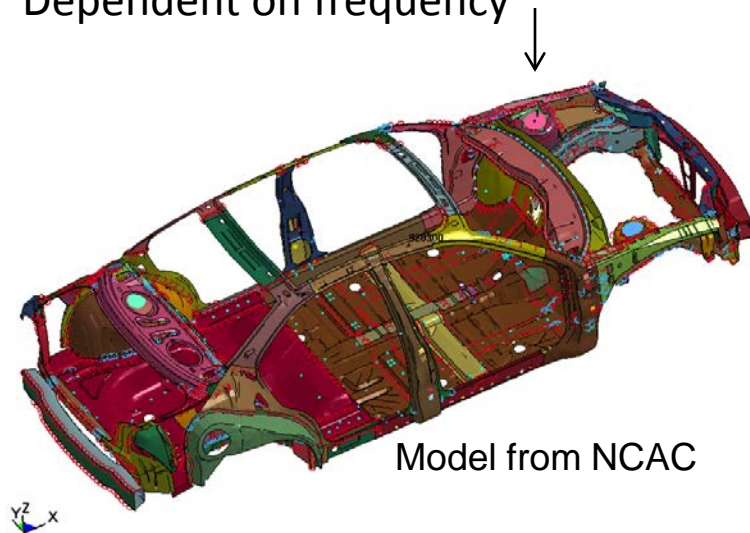


# FRF

## What is FRF?



- FRF, as a transfer function, expresses structural response to applied load as a function of frequency
- Property of structure system
- Dependent on frequency



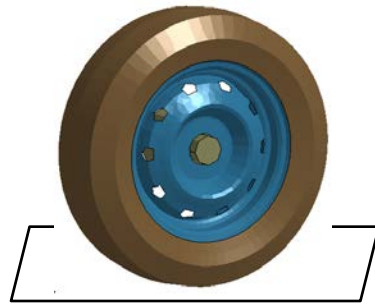
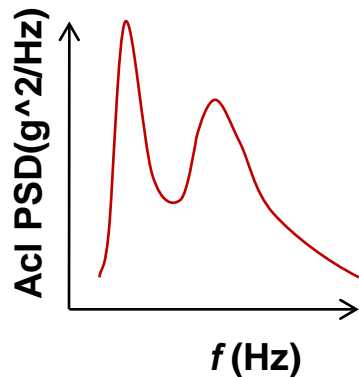
# Random vibration

## Why do we need random vibration analysis?

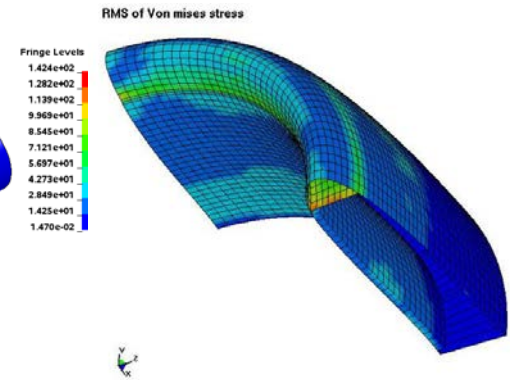
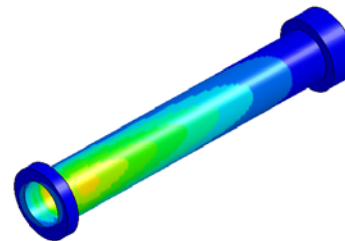
- The loading on a structure is not known in a definite sense
- Many vibration environments are not related to a specific driving frequency
- Examples:
  - Wind-turbine
  - Air flow over a wing or past a car body
  - Acoustic input from jet engine exhaust
  - Earthquake ground motion
  - Wheels running over a rough road
  - Ocean wave loads on offshore platforms

**Input:** PSD (Power Spectral Density), or SPL (Sound Pressure Level)

**Output:** D3PSD, D3RMS, NODOUT\_PSD, ELOUT\_PSD

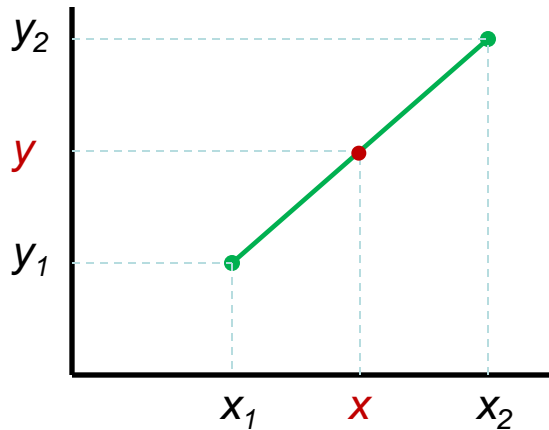


ion analysis Aluminum

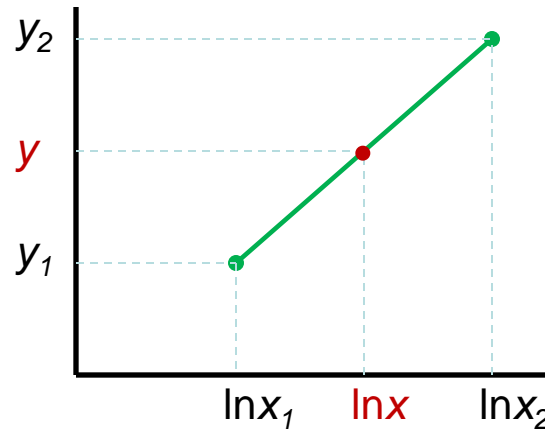


# New interpolation options on PSD curve

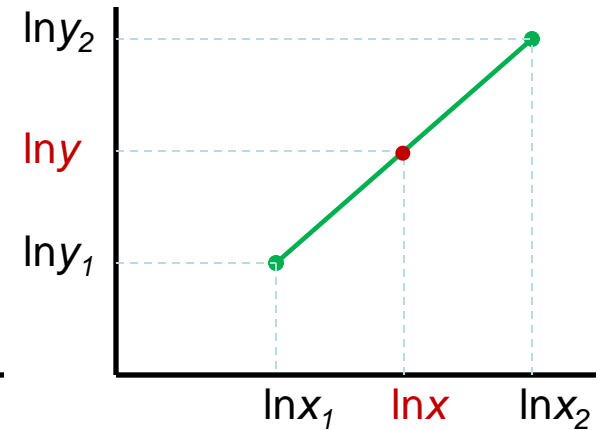
## Three types of interpolation for PSD curves



linear-linear

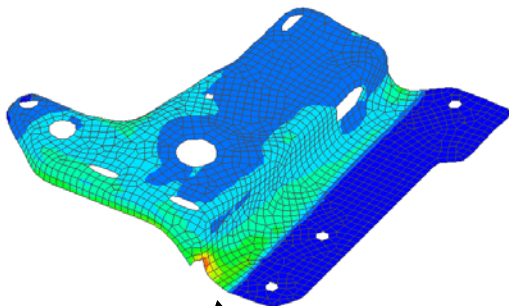
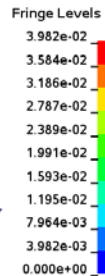


linear-log (semi-log)

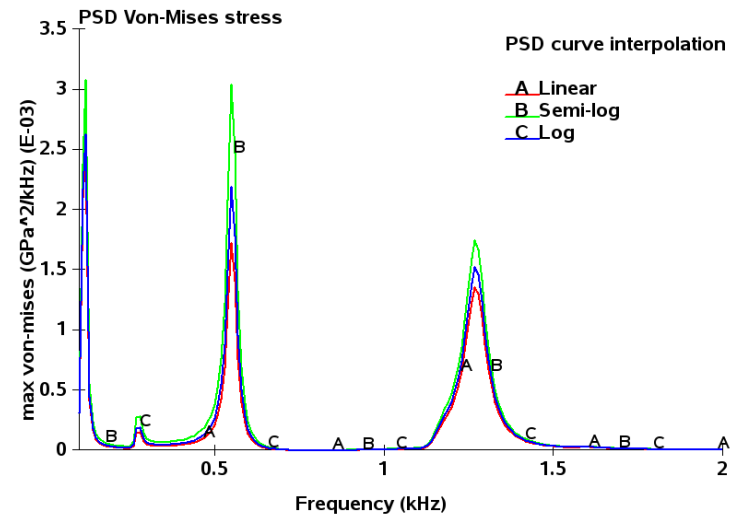


log-log

Random fatigue analysis: 2014-T6 Al  
Contours of Effective Stress (v-m)  
max IP value  
min=0, at elem# 479601  
max=0.0398196, at elem# 479769



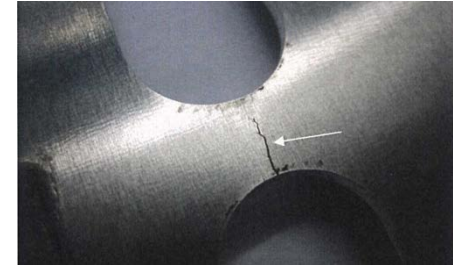
Element with peak stress



# Random vibration fatigue

## \*FREQUENCY\_DOMAIN\_RANDOM\_VIBRATION\_FATIGUE

- Calculate fatigue life of structures under random vibration
- Based on S-N fatigue curve
- Based on probability distribution & Miner's Rule of Cumulative Damage Ratio



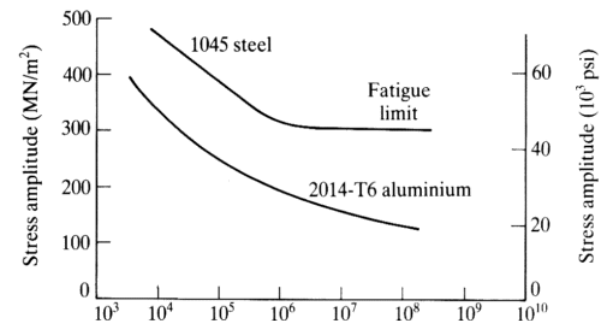
$$R = \sum_i \frac{n_i}{N_i}$$

## \*MAT\_ADD\_FATIGUE

Card 1	1	2	3	4	5	6	7	8
Variable	MID	LCID	LTYPE	A	B	STHRES	SNLIMT	
Type	I	I	I	F	F	F	I	
Default	none	-1	0	0.0	0.0	none	0	

$$N \cdot S^m = a$$

$$\log(S) = a - b \cdot \log(N)$$



Number of cycles,  $N_f$  Source of picture:  
<http://www.efunda.com>

# SSD

## Why do we need SSD (Steady State Dynamics)?

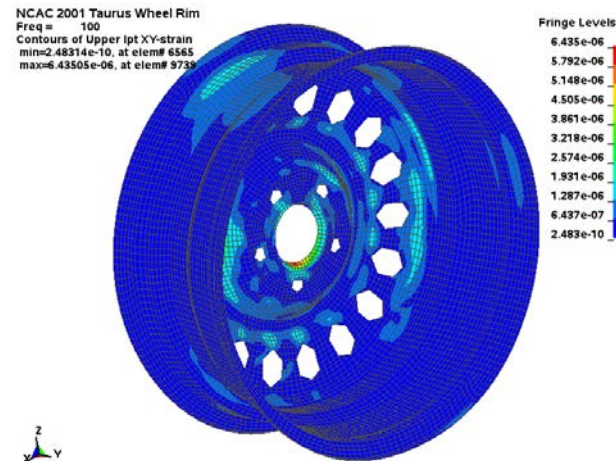
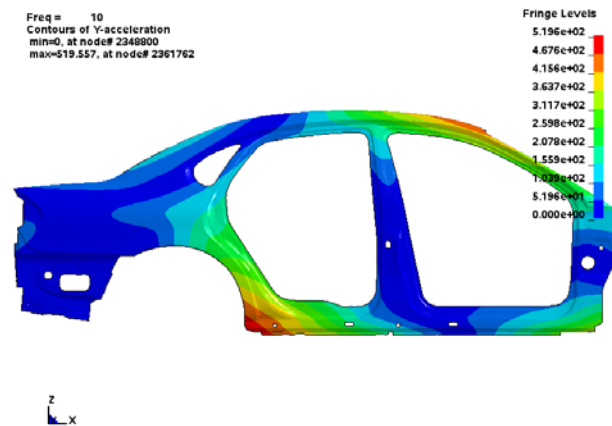
- Harmonic excitation is often encountered in engineering systems. E.g. it is commonly produced by the unbalance in rotating machinery.
- The load may also come from periodic load, e.g. in fatigue test.
- The excitation may also come from uneven base, e.g. the force on tires running on a zig-zag road



**Input:** frequency load spectrum (complex variable)

$$F(t) = F_0 \sin(\omega t + \phi)$$

**Output:** D3SSD, NODOUT\_SSD, ELOUT\_SSD

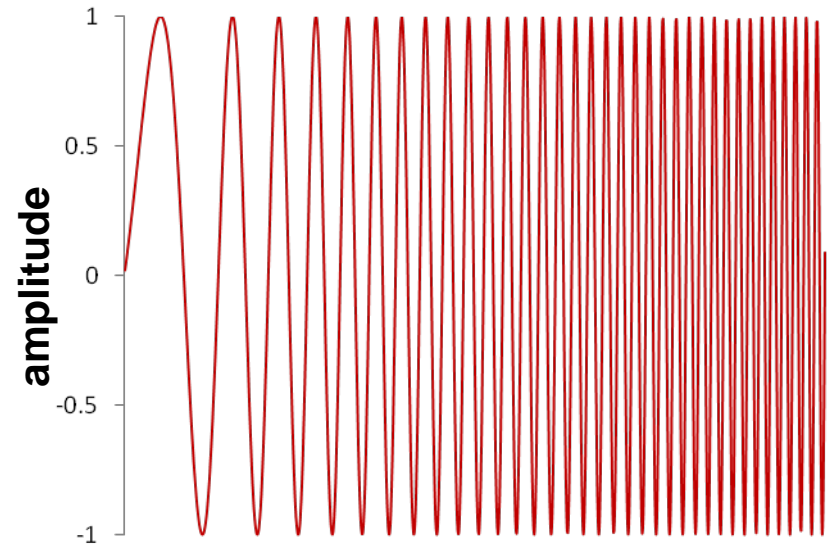


# SSD Fatigue (ongoing development)

## \*FREQUENCY\_DOMAIN\_SSD\_FATIGUE

- Calculate fatigue life of structures under steady state vibration (e.g. sine sweep)
- Based on S-N fatigue curve
- Based on Miner's Rule of Cumulative Damage Ratio
- Rainflow counting algorithm for each frequency for one period

$$R = \sum_i \frac{n_i}{N_i}$$



Sine sweep

# Example of SSD Fatigue

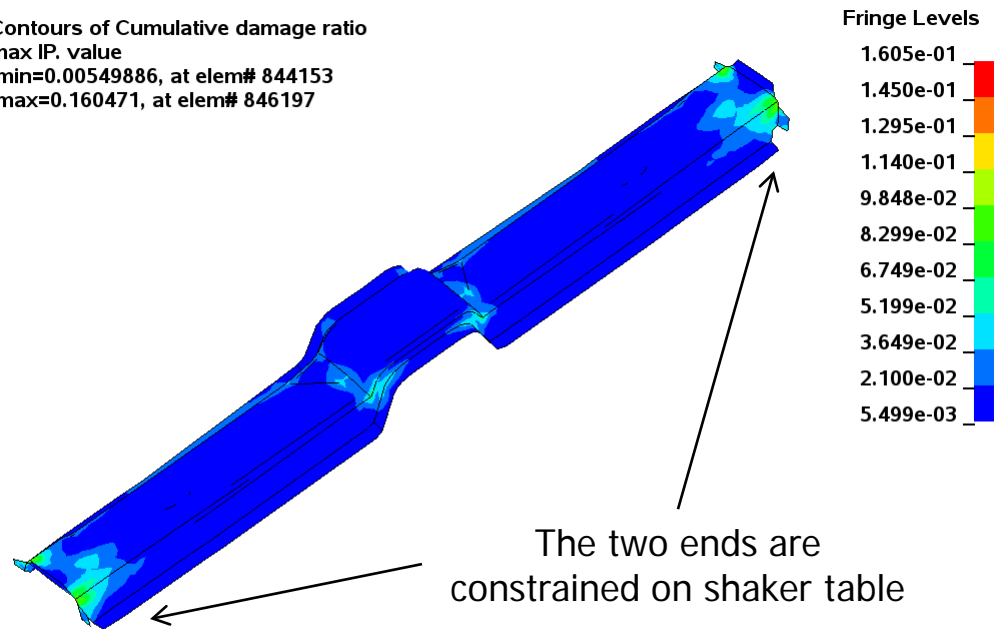
## Loading condition

Freq (Hz)	Acl (g)	Duration (min)
16	0.5	10
20	0.5	10
25	0.5	10
31.5	0.5	10
...	...	...
2000	0.5	10

## S-N fatigue curve

$\sigma$ (MPa)	N
100	$8 \times 10^4$
10	$8 \times 10^5$
1.	$8 \times 10^6$
0.1	$8 \times 10^7$
0.01	$8 \times 10^8$

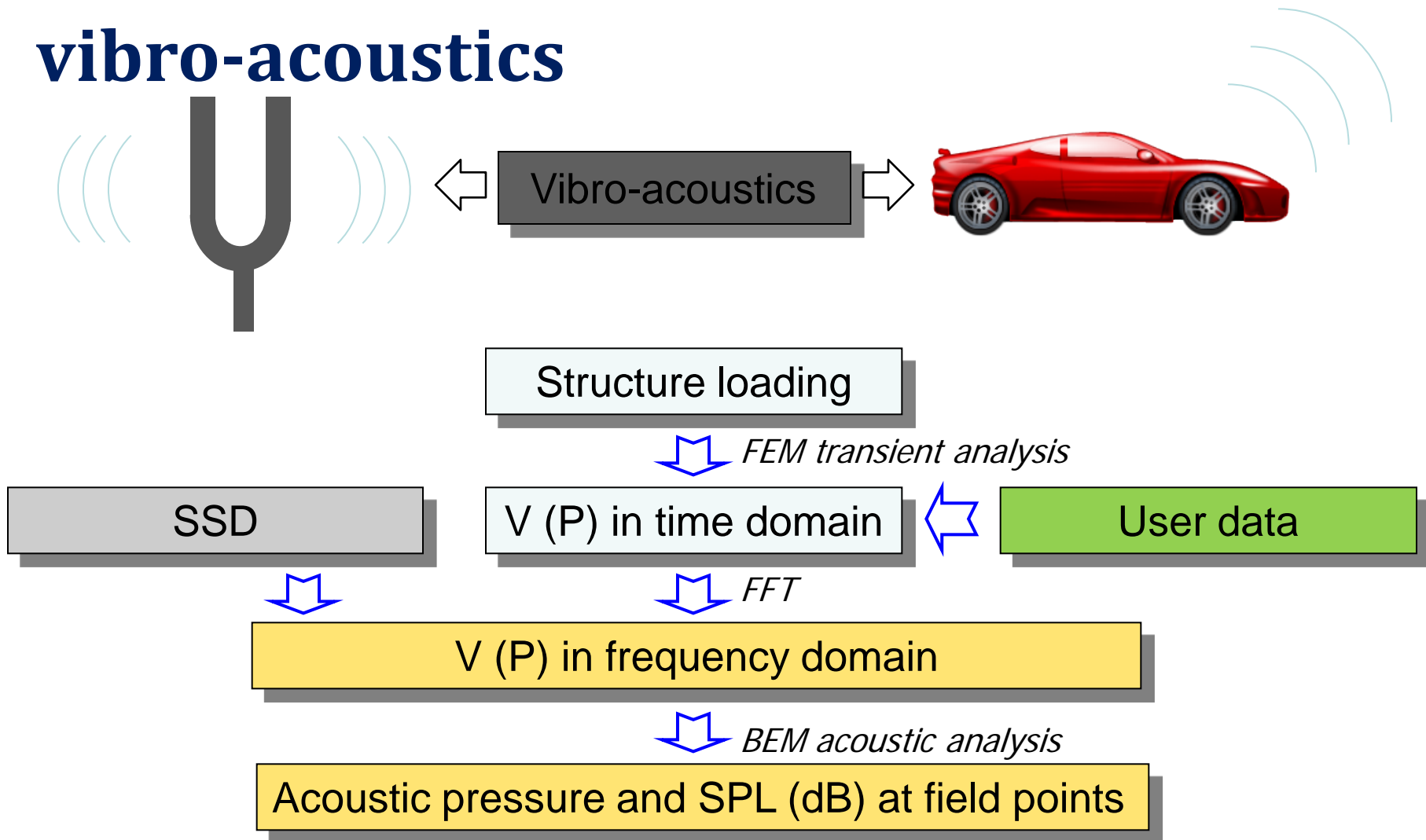
Contours of Cumulative damage ratio  
max IP. value  
min=0.00549886, at elem# 844153  
max=0.160471, at elem# 846197





# BEM Acoustics

## vibro-acoustics



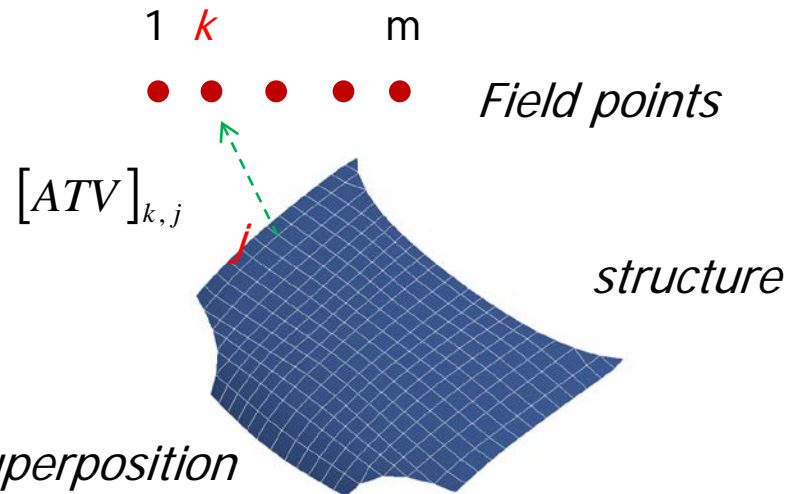
# ATV and MATV

## \*FREQUENCY\_DOMAIN\_ACOUSTIC\_BEM\_ATV \*FREQUENCY\_DOMAIN\_ACOUSTIC\_BEM\_MATV

- ATV calculates acoustic pressure at field points due to unit normal velocity of each surface node.
- MATV calculates acoustic pressure at field points due to vibration in eigenmodes.
- ATV / MATV is dependent on structure model, properties of acoustic fluid as well as location of field points.
- ATV / MATV is useful if the same structure needs to be studied under multiple load cases.

$$\begin{aligned}\{P\} &= [ATV]\{V\} \\ &= [ATV](-i\omega[\Phi]\{q\}) \\ &= -i\omega[ATV][\Phi]\{q\} \\ &= [MATV]\{q\}\end{aligned}$$

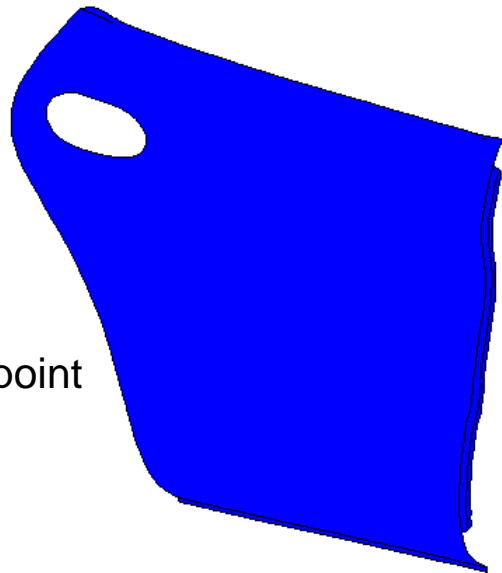
*Velocity vector is obtained by modal superposition*



# Example: ATV of car door model

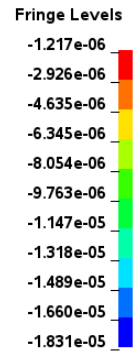
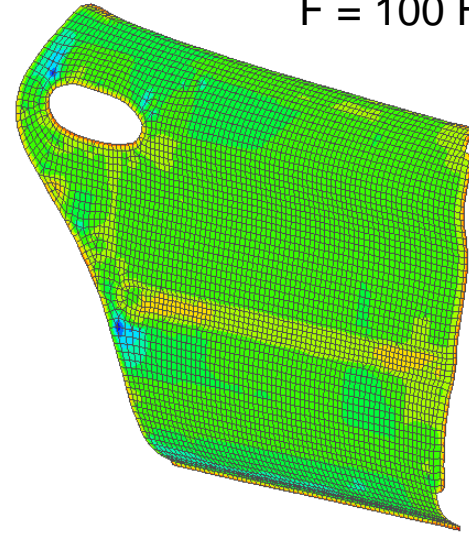


Measure point

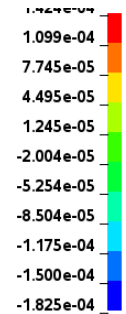
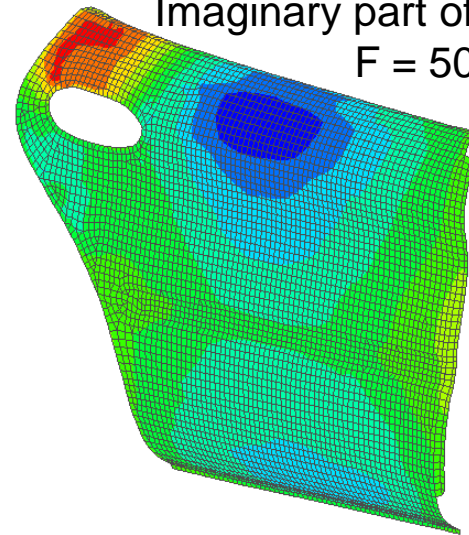


No. of elements: 5198  
No. of frequencies: 101

Real part of ATV  
 $F = 100 \text{ Hz}$

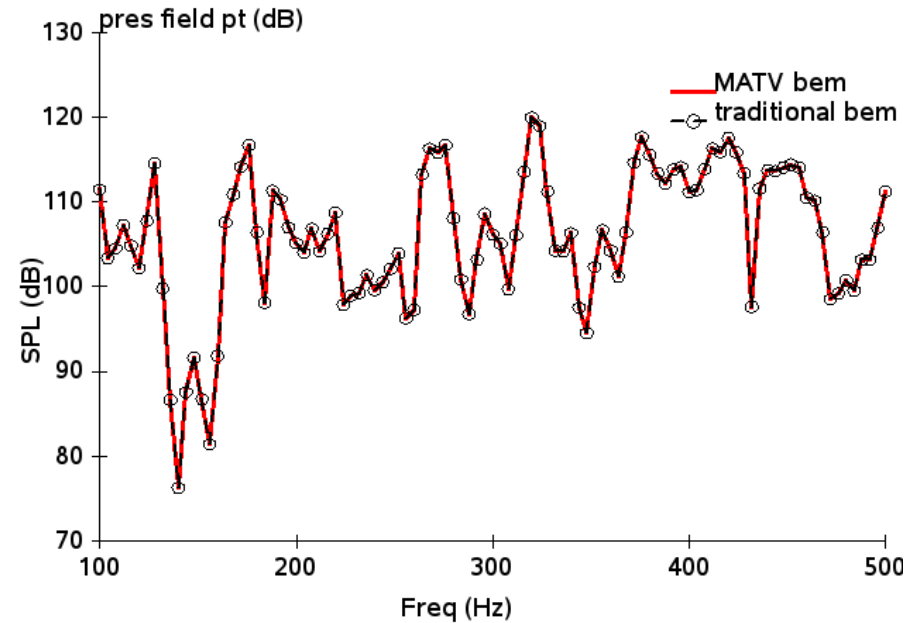
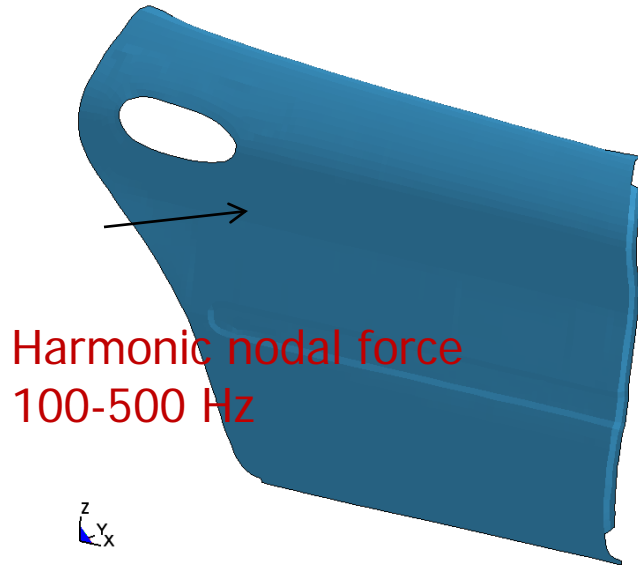


Imaginary part of ATV  
 $F = 500 \text{ Hz}$



(given by d3atv)

# MATV BEM is efficient



## CPU time (1 core)

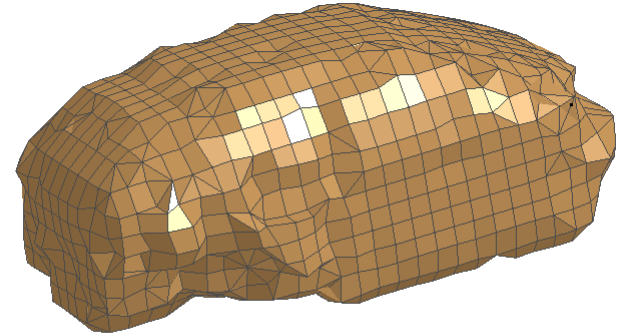
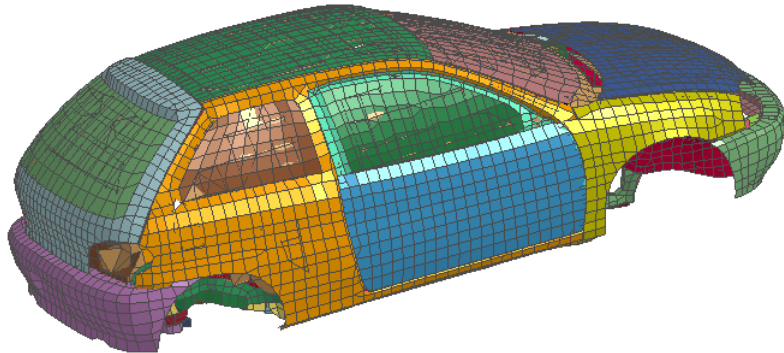
Intel Xeron CPU E5504 @2.00 GHz

Cases	traditional BEM	MATV BEM
1 load case	2 h 39 m 50 s	4 h 40 m 56 s
10 load cases	<b>26 h 38 m 18 s</b>	<b>4 h 41 m 10 s</b>



# D3ACS for BEM acoustics

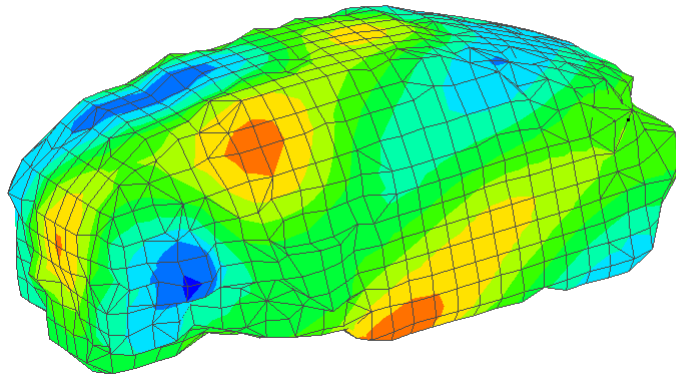
**\*DATABASE\_FREQUENCY\_BINARY\_D3ACS**



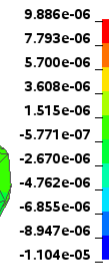
**Acoustic volume of compartment**



Freq = 400



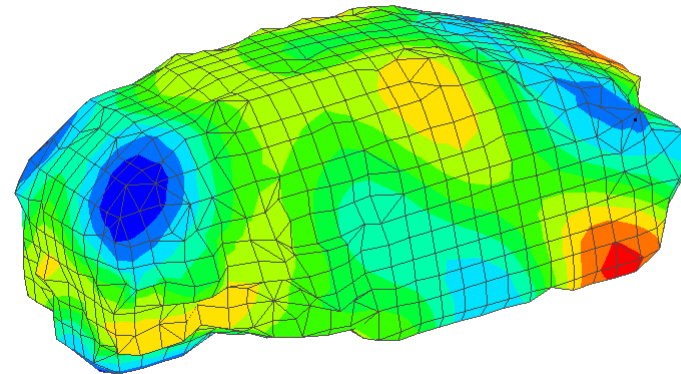
Fringe Levels



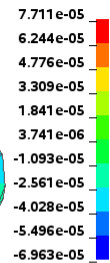
**Real part of surface pressure**



Freq = 400



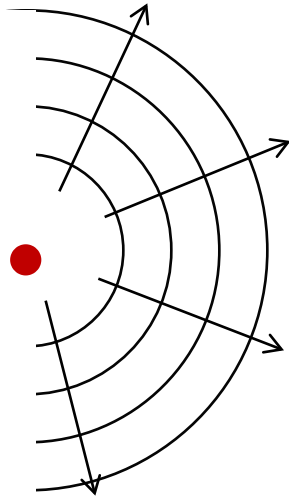
Fringe Levels



**Imaginary part of surface pressure**

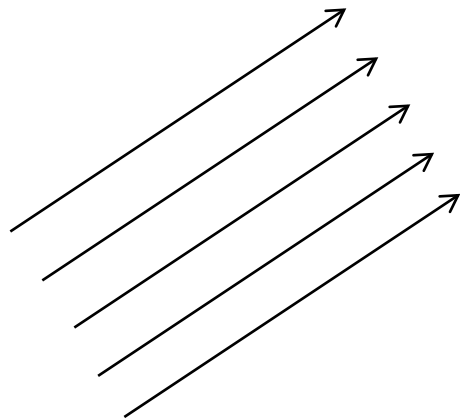
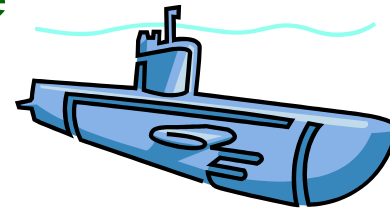
# Incident waves for acoustic analysis

\*FREQUENCY\_DOMAIN\_ACOUSTIC\_INCIDENT\_WAVE



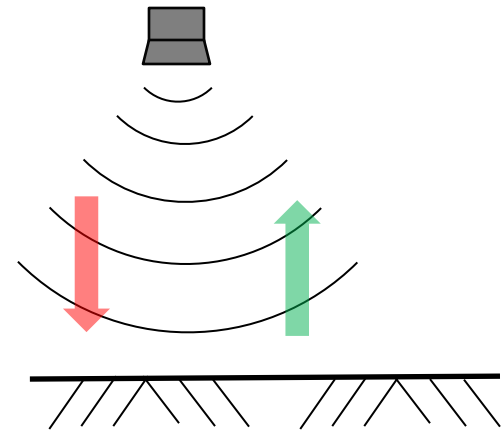
**Spherical wave**

$$p^i = A \frac{e^{-ikR}}{R}$$



**Plane wave**

$$p^i = Ae^{-ik(\alpha x + \beta y + \gamma z)}$$

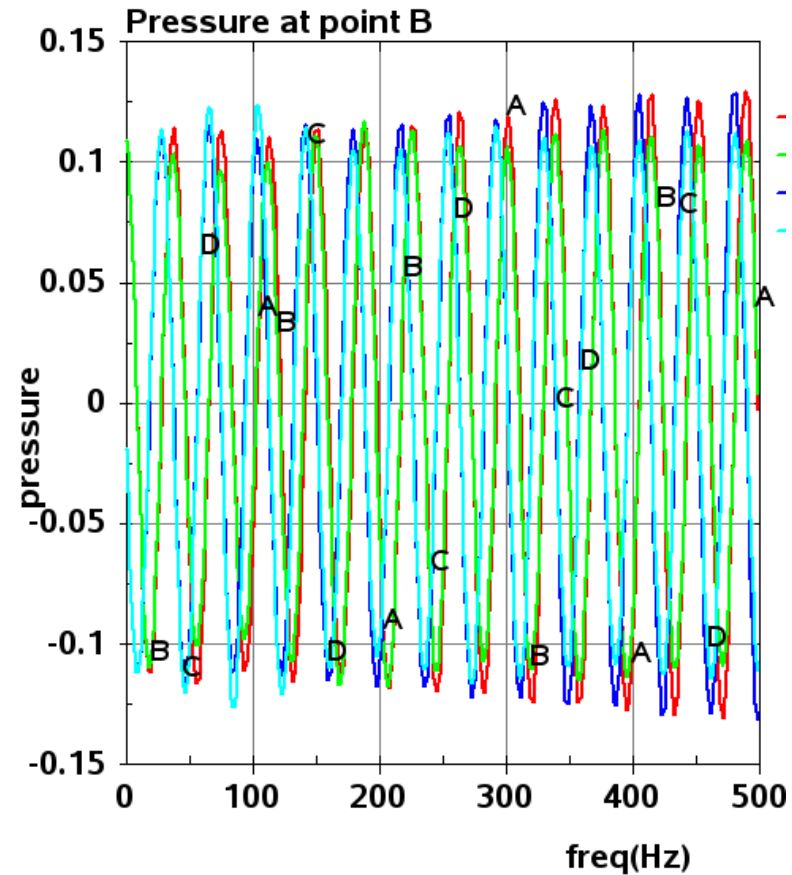
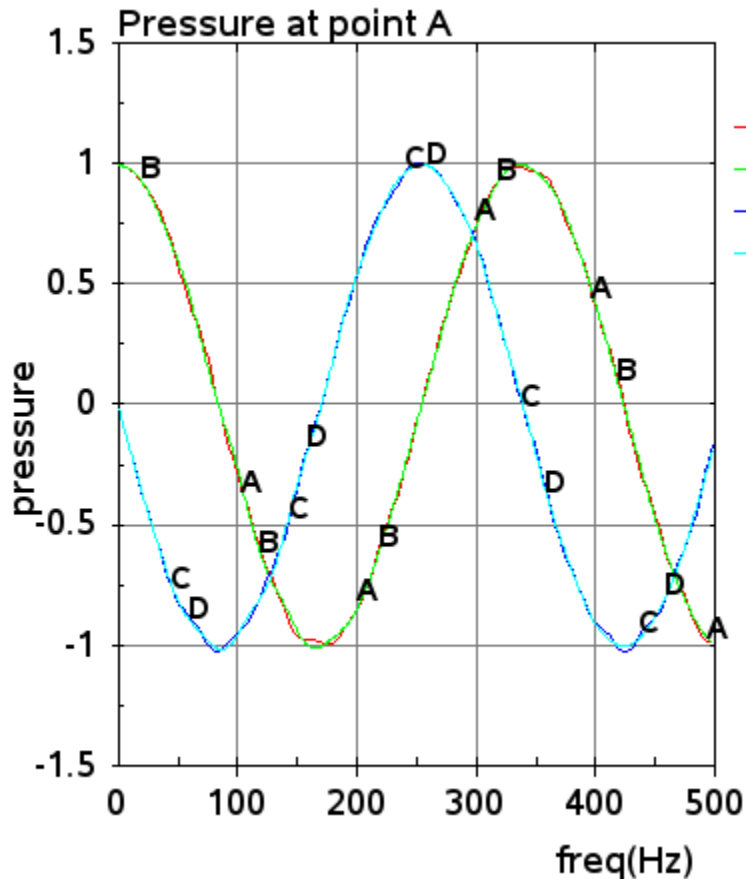


# Example: sound scattering on rigid sphere

Spherical source



A Real part (LS-DYNA)  
B Real part (Analytical)  
C Imaginary part (LS-DYNA)  
D Imaginary part (Analytical)

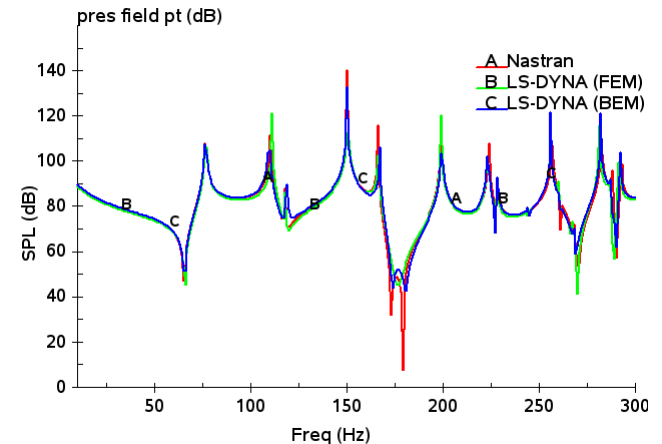
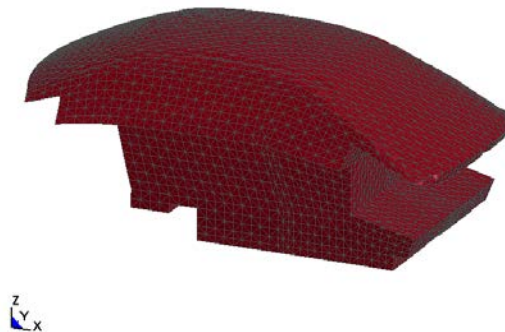
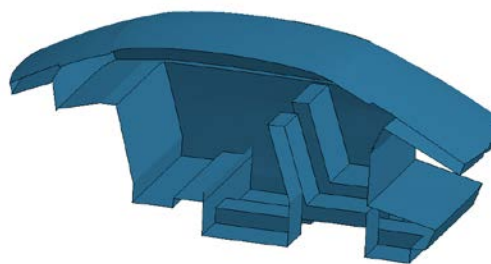


# FEM Acoustics

## \*FREQUENCY\_DOMAIN\_ACOUSTIC\_FEM

- An alternative method for acoustics. It helps predict and improve sound and noise performance of various systems. The FEM simulates the entire propagation volume -- being air or water
- Compute acoustic pressure and SPL (sound pressure level)
- Output frequency range dependent on mesh size
- Available elements: hexahedron, tetrahedron, and pentahedron

**Output:** D3ACS; PRESS\_PA; PRESS\_DB



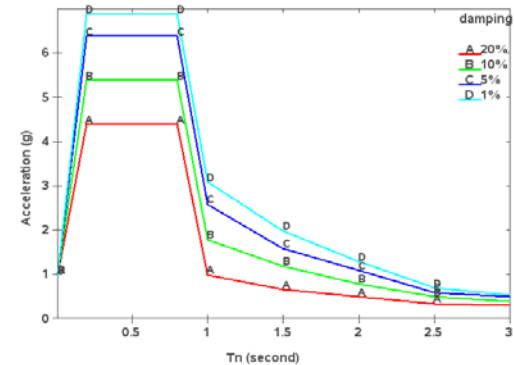


# Response spectrum analysis

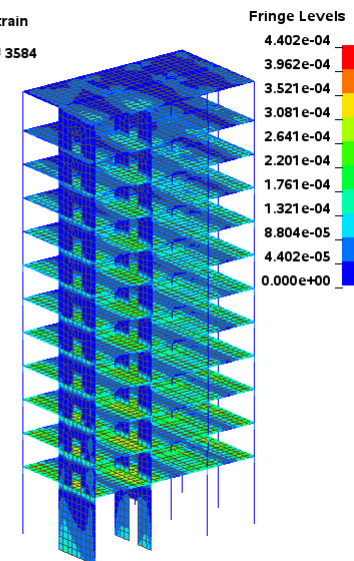
## \*FREQUENCY\_DOMAIN\_RESPONSE\_SPECTRUM

- Various mode combination methods
  - SRSS method
  - NRC Grouping method
  - CQC method
  - Double Sum methods
  - NRL SUM method
- Evaluate peak response of structure
- The input spectrum is dependent on damping (using **\*DEFINE\_TABLE** to define the series of excitation spectrum corresponding to each damping ratio).
- Application: *earthquake engineering*, *nuclear power plants design* etc.
- Strain results are obtained by turning on **STRFLG** in **\*DATABASE\_EXTENT\_BINARY**.

**Output:** D3SPCM



A simple tower model  
Contours of Lower Ipt X-strain  
min=0, at elem# 1  
max=0.00044018, at elem# 3584



# Recent Developments in LS-DYNA (part 2)

LS-DYNA Forum  
October 8, 2014  
Bamberg



**LSTC**  
Livermore Software  
Technology Corp.

# Particle Methods

# Meshless Particle Solvers

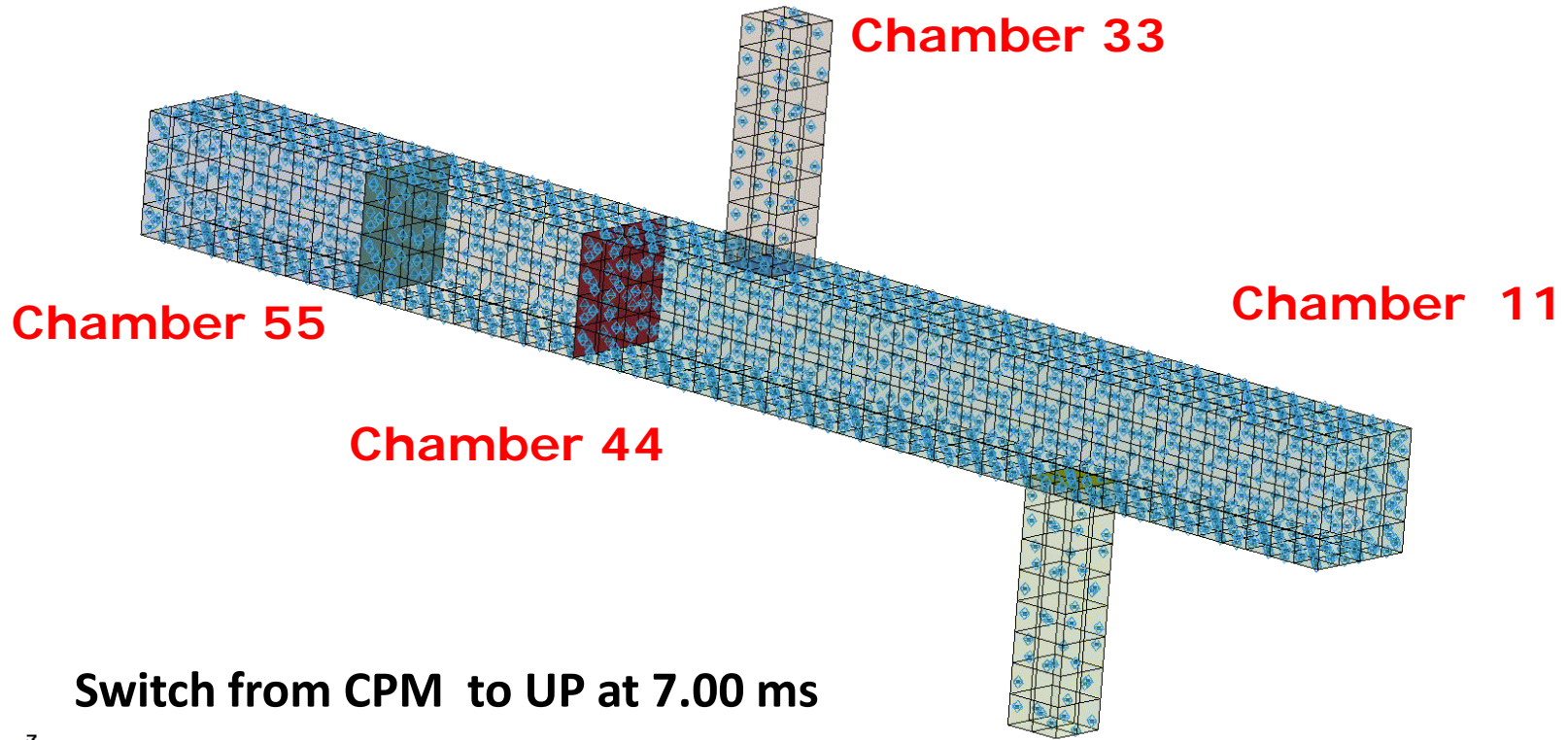
- 1) Particle Gas
  - CPM - Ideal Gas Law
  - Particle Blast - Real Gas Law
- 2) SPH
- 3) Discrete Element Method (DEM)
  - Discrete Element Sphere
  - Discrete Element Method with Bond
- 4) Coupled Multi-Physics Solvers

# Particle Gas, CPM-Ideal Gas Law

- Modeled by ideal gas law:  $pV=nRT$
- The volume of molecules is neglected
- maintain the same Maxwell-Boltzmann velocity distribution at thermal equilibrium
- Work for low pressure and moderate temperature

# CPM/UP switch with Chambers

Time = 0



Switch from CPM to UP at 7.00 ms



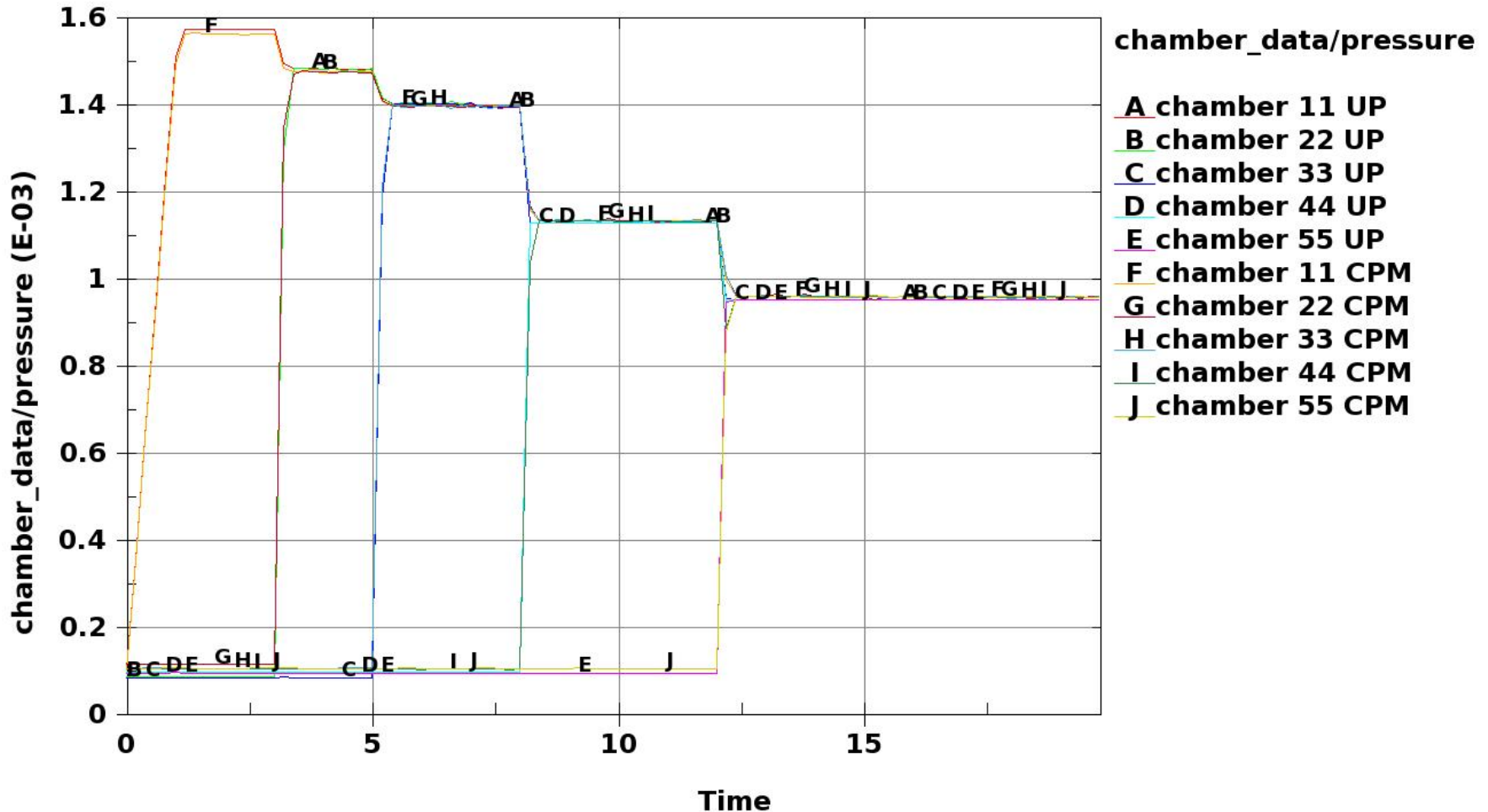
Old: chambers collapsed into one bag

Chamber 22

*New: chambers becomes separated UP domain*

# CPM/UP switch with Chambers

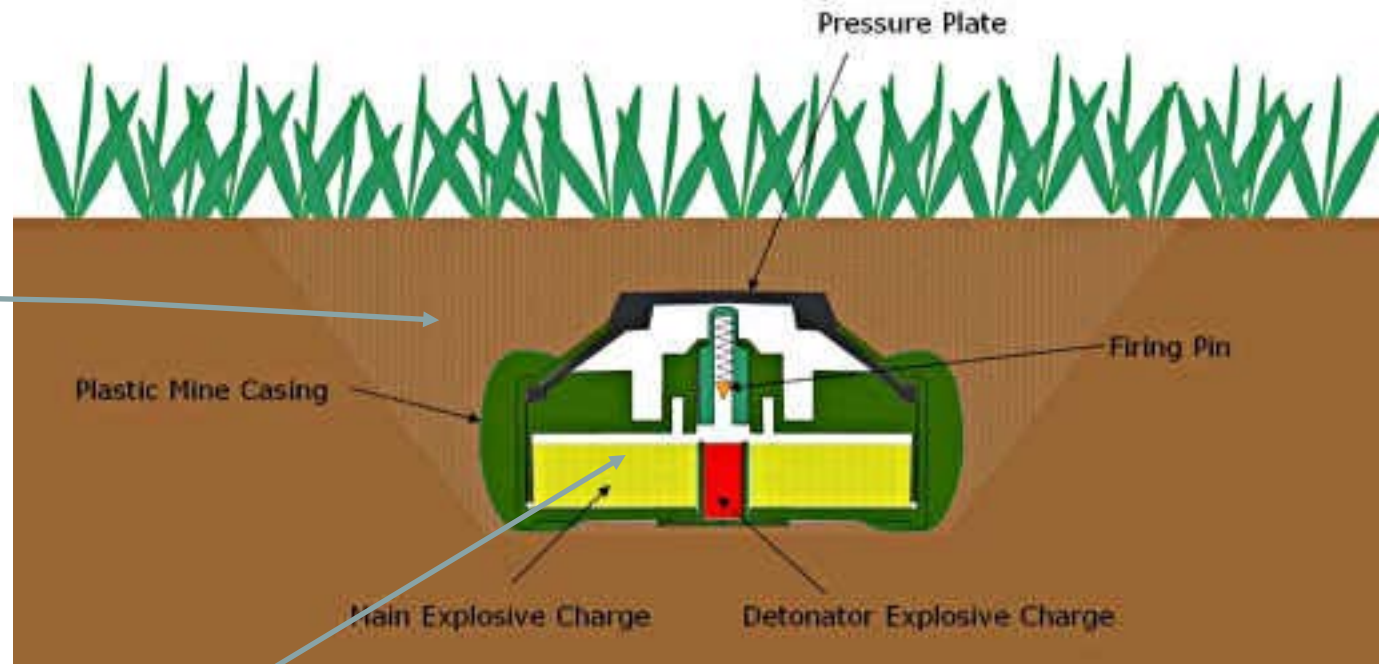
- CPM/UP Switch at 7 ms, curves A, B, C, D, E
- CPM all the way, curves F, G, H, I, J



# Particle Gas, Particle Blast-Real Gas Law

**Air particle  
(Ideal Gas Law)**

Anti-Personnel Mine (Blast Type) Components



**Discrete Element  
Method (DEM)**

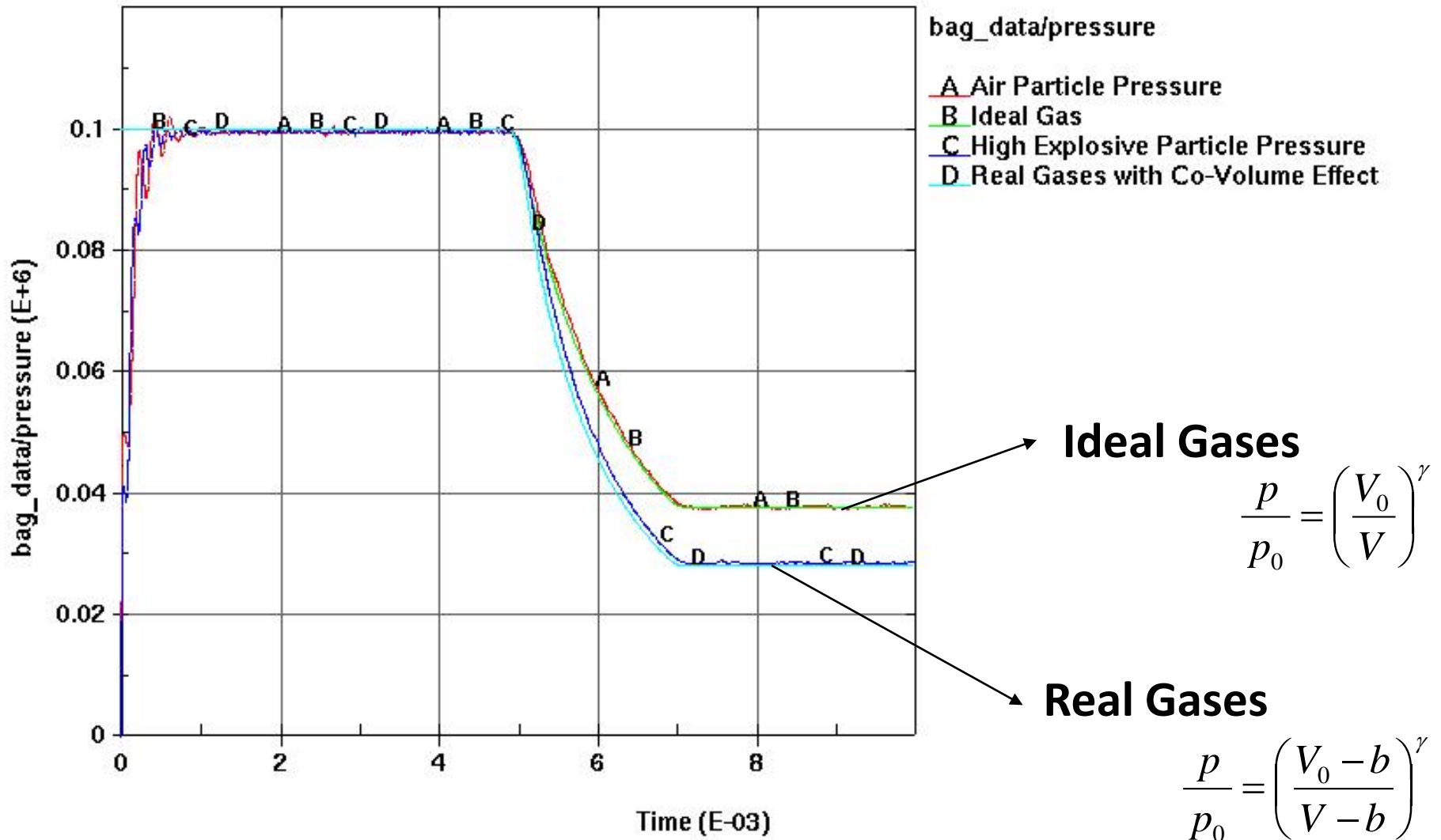
**HE Particle  
(Real Gas Law)**



# Particle Gas, Particle Blast—Real Gas Law

- Air Particle
  - Modeled by ideal gas law (CPM):  $pV=nRT$
- High Explosive (HE) Particles
  - Modeled by real gases:  $p(V-b)=nRT$
  - The co-volume effect is included
  - Work for high pressure and high temperature
  - Pressure drop sharply during adiabatic expansion

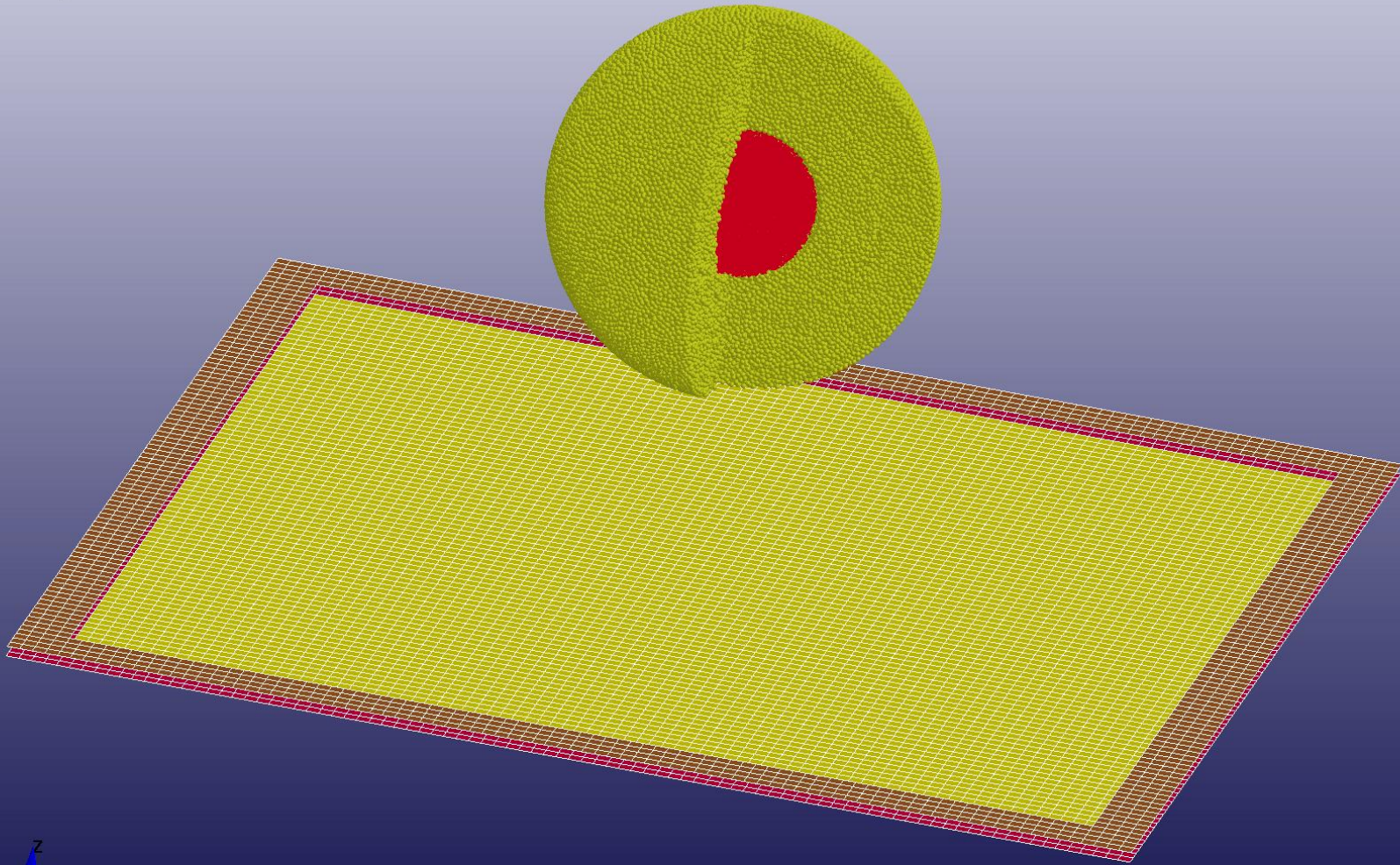
# Particle Gas, Ideal and Real Gas Law



# Particle Blast, Real Gas Law

## Numerical Example

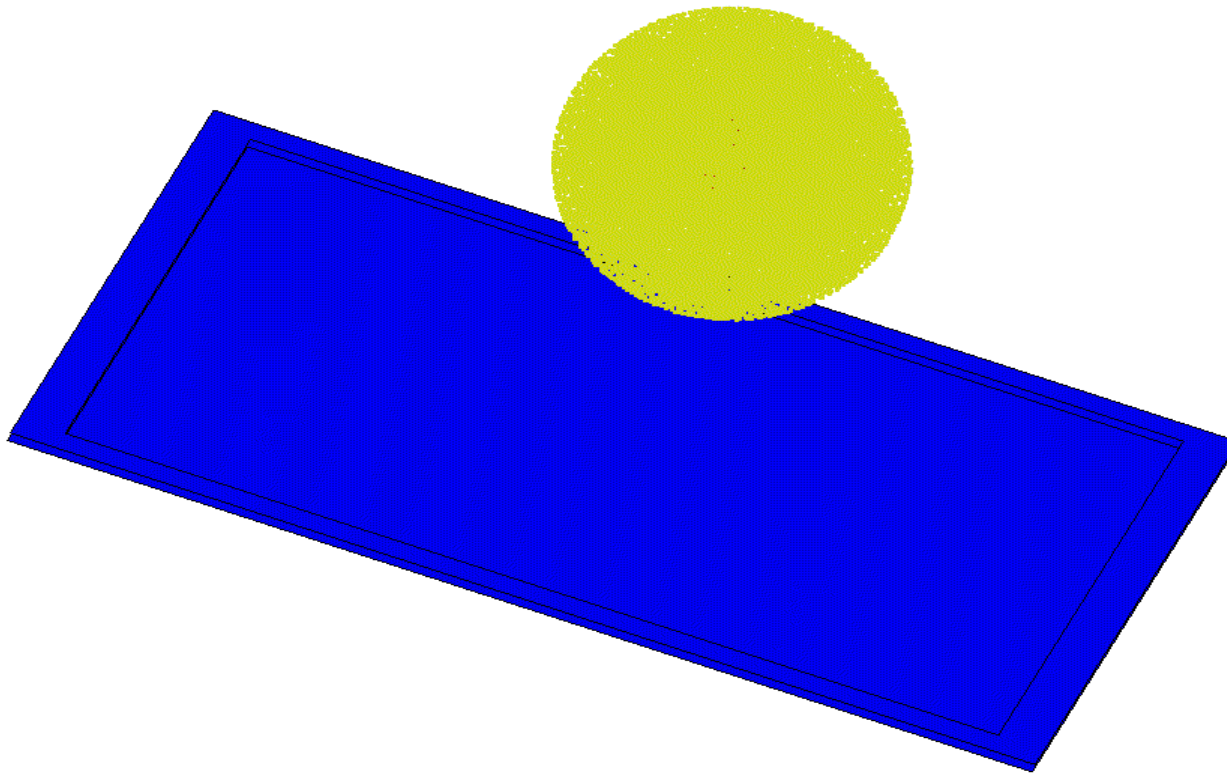
PARTICLE BENCHMARK  
Time = 0



# Particle Blast, Real Gas Law

## Numerical Example

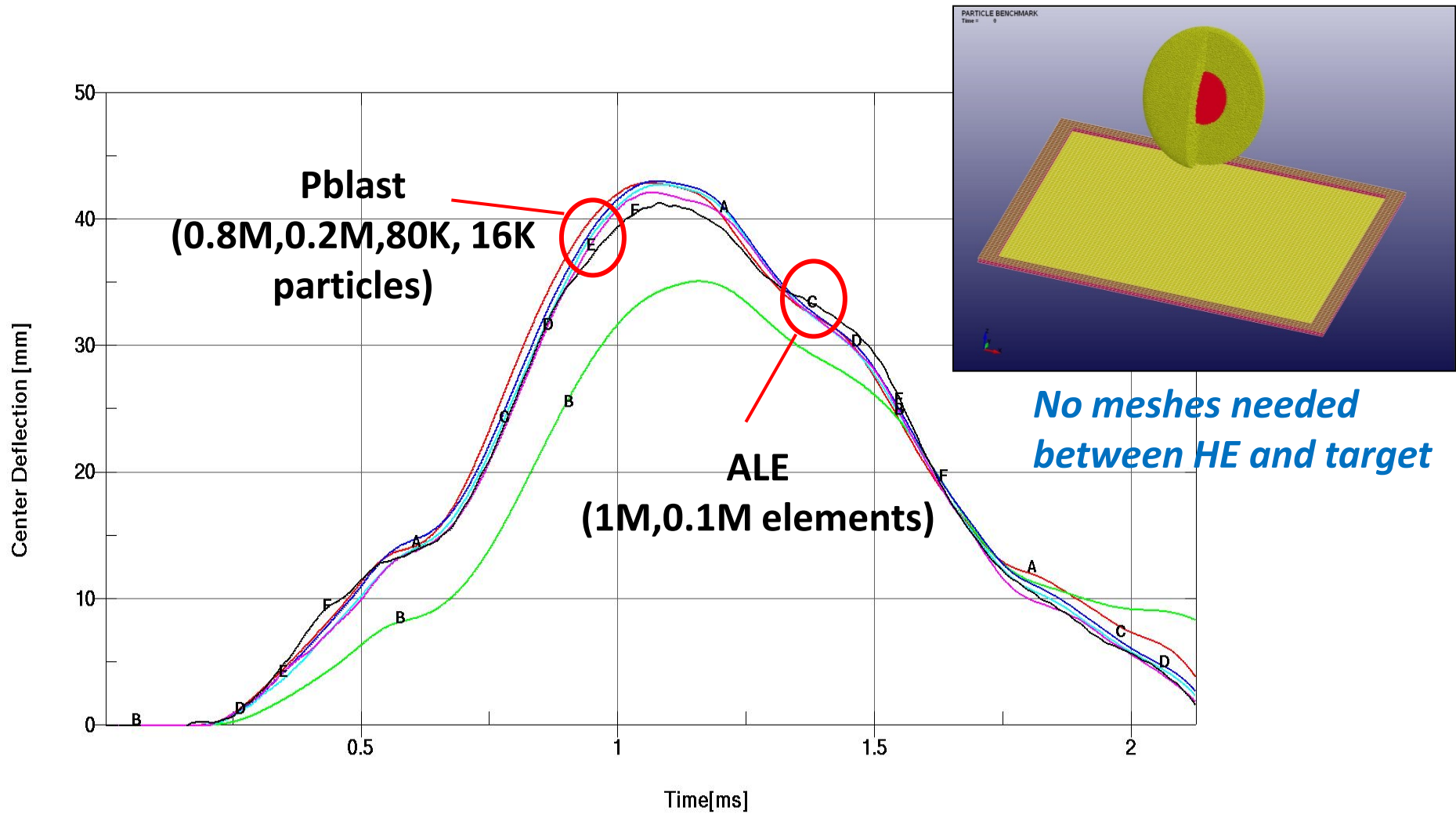
### Blast simulation with sand



# Particle Blast, Real Gas Law

## Numerical Example

### Simulation Results for 700mm Model



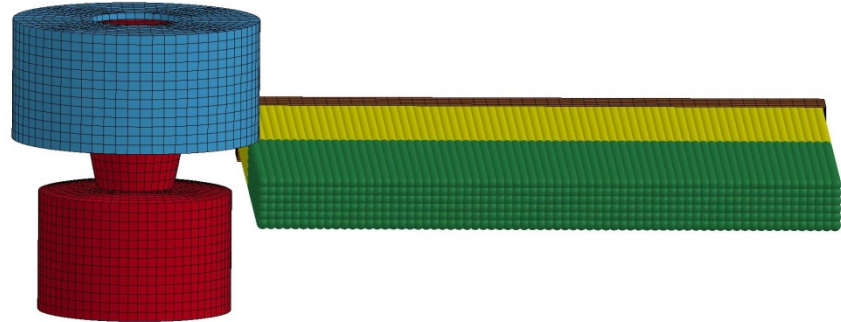
# Enhancement of SPH

- 1. Friction Stir Welding**
- 2. SPH to SPH contact**
- 3. SPH active region and new bucket sort**

# Friction Stir Welding

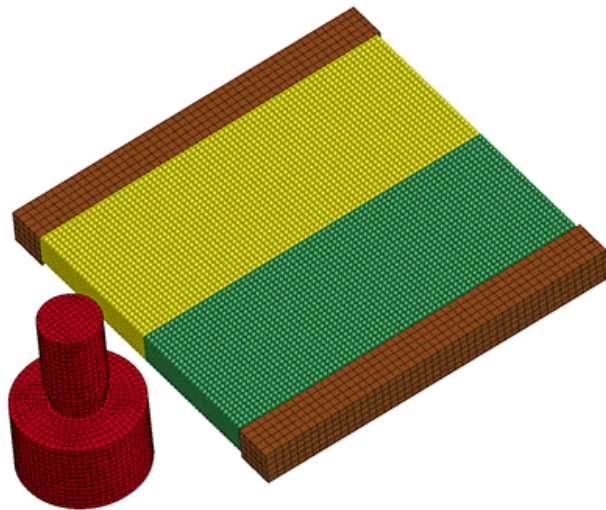
Double Sided FSW (Bobbin Tool) - 600 RPM, 1200mm/min  
Time = 0

*Extended SPH thermal solver  
for SPH form 7 and 8*



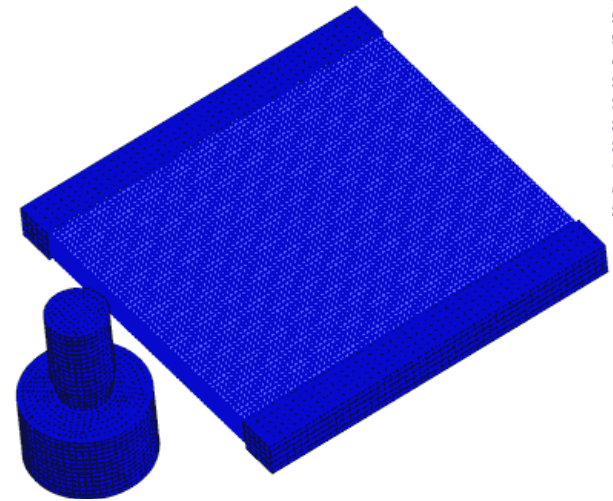
**Double sided FSW 600 RPM, 1200 mm/min (plastic work and friction energy to heat)**  
Courtesy of Kirk A. Fraser @ PredictiveEngineering

Double Sided FSW (Bobbin Tool) - 600 RPM, 1200mm/min  
Time = 0



**Material Mixing**

Double Sided FSW (Bobbin Tool) - 600 RPM, 1200mm/min  
Time = 0  
Contours of Temperature  
min=20, at node# 501819  
max=20, at node# 501819



**Temperature Contours**

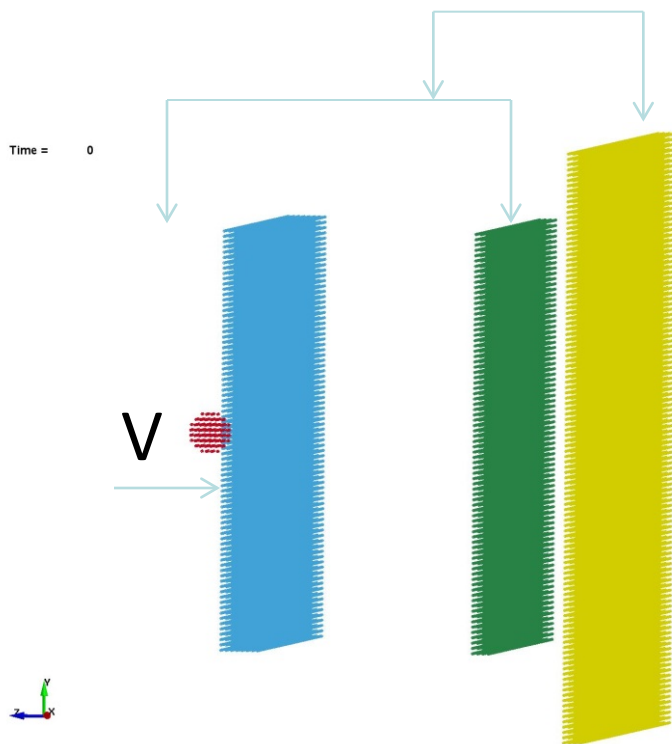
Fringe Levels  
6.310e+02  
5.699e+02  
5.088e+02  
4.477e+02  
3.866e+02  
3.255e+02  
2.644e+02  
2.033e+02  
1.422e+02  
8.110e+01  
2.000e+01

# SPH Interaction

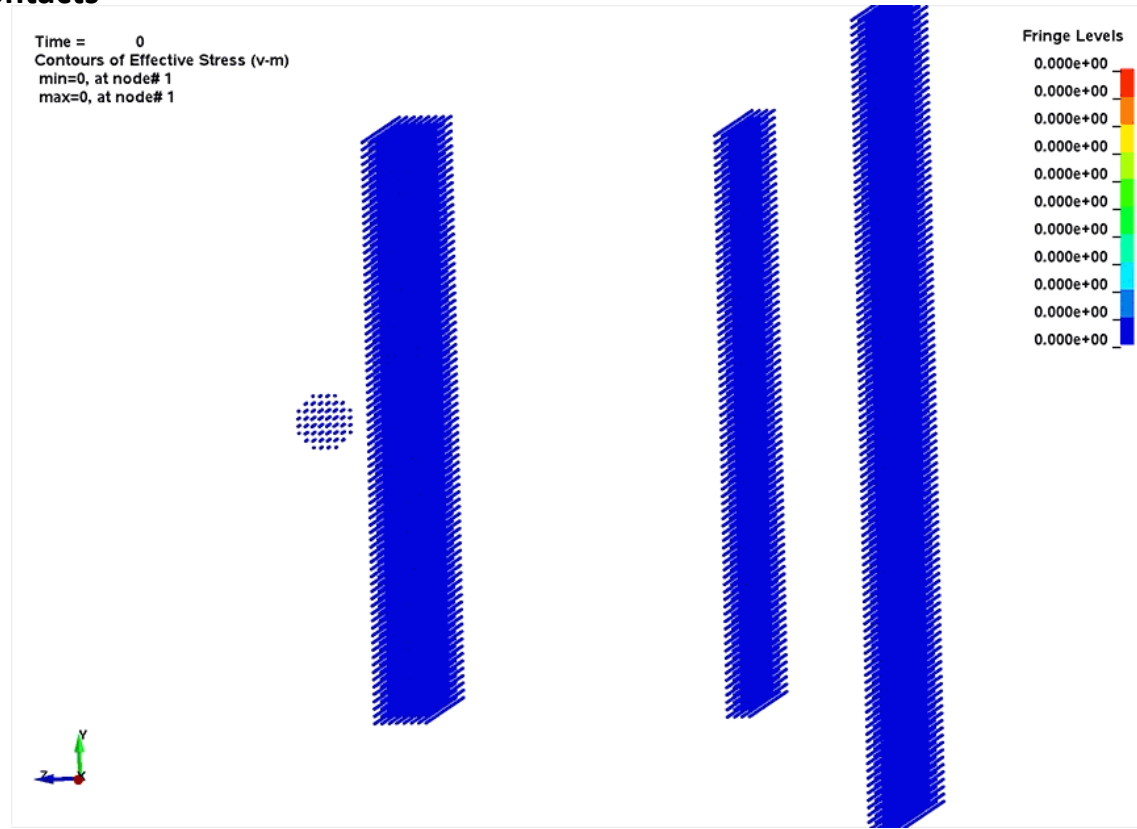
Multiple impacts with Keyword: SECTION\_SPH\_INTERACTION

Define the different type of interactions between SPH parts.

Metals with  
Standard interactions

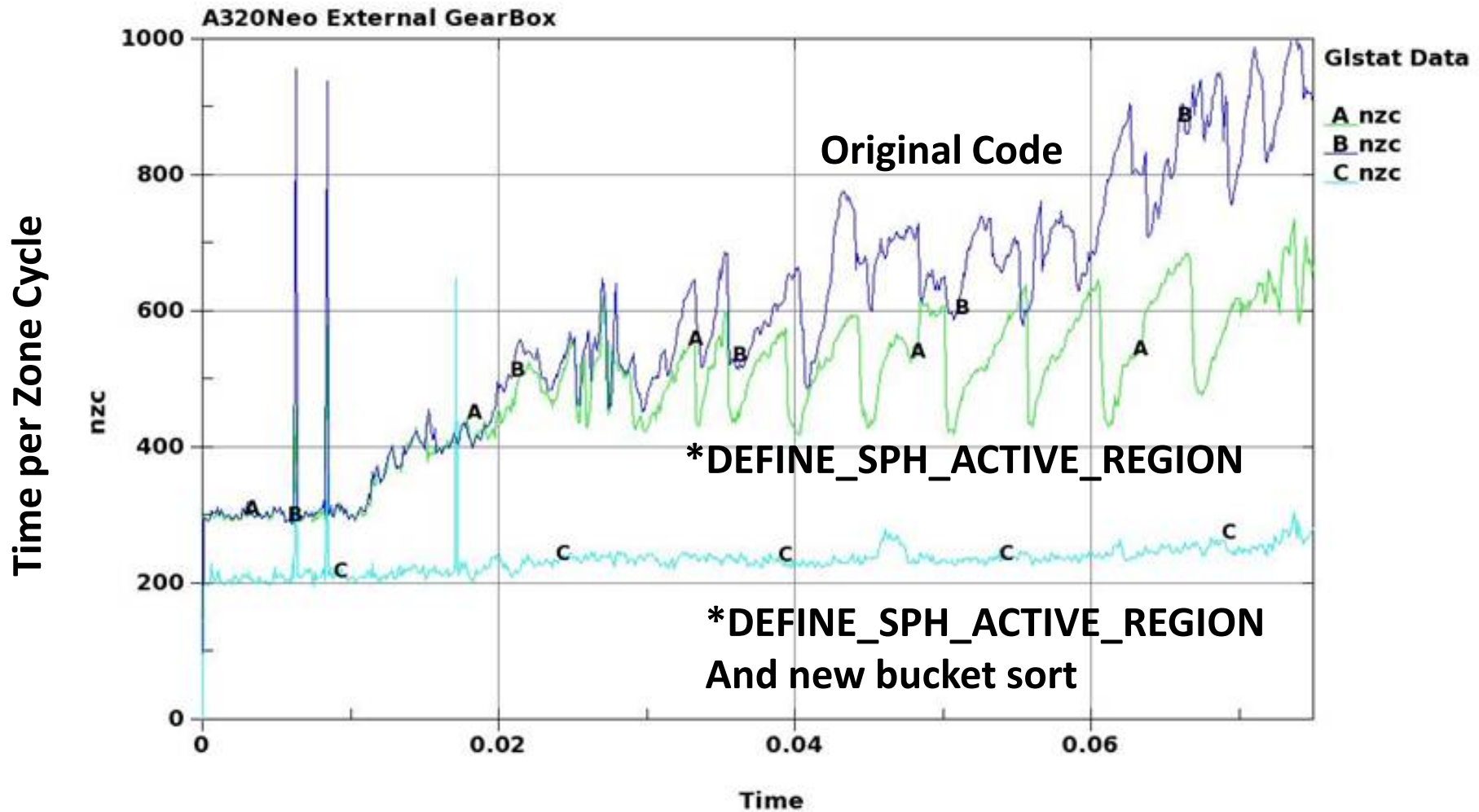


Ceramic with node  
to node contacts





# SPH Active Region & Better Bucket Sort



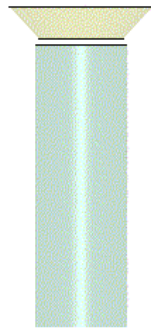
# Discrete Element Method (DEM)

- 1. Discrete Element Sphere**
- 2. Discrete Element Method with Bond**

# Discrete Element Method (DEM)

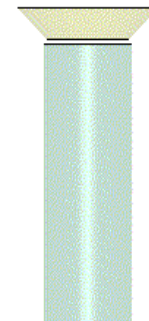
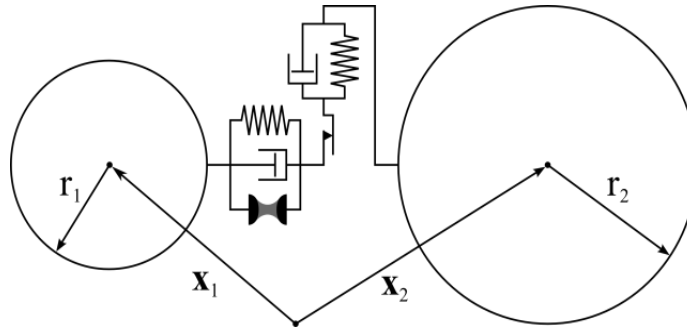
## Discrete Element Sphere

LS-DYNA keyword deck by LS-PrePost  
Time = 0



**Dry Particle**

LS-DYNA keyword deck by LS-PrePost  
Time = 0



**Wet Particle**



# Discrete Element Method (DEM)

## Discrete Element Sphere

Variation of the parameters Courtesy of Dr.-Ing. Nils Karajan, Dynamore GmbH

	1	2	3	4	5
RHO	0.80E-6	2.63E-6	2.63E-6	2.63E-6	1.0E-6
P-P Fric	0.57	0.57	0.57	0.10	0.00
P-P FricR	0.10	0.10	0.01	0.01	0.00
P-W FricS	0.27	0.30	0.30	0.10	0.01
P-W FricD	0.01	0.01	0.01	0.01	0.00
CAP	0	0	1	1	1
Gamma	0.00	0.00	7.20E-8	2.00E-6	7.2E-8

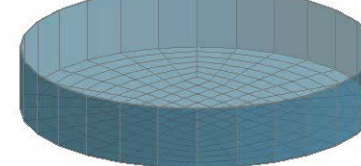
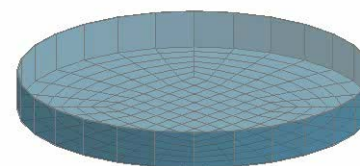
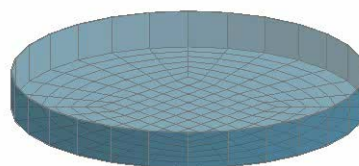
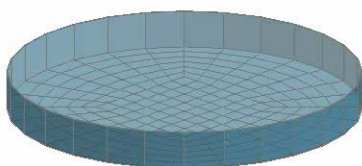
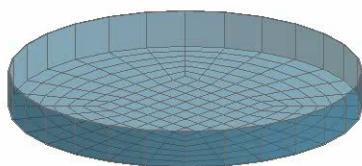
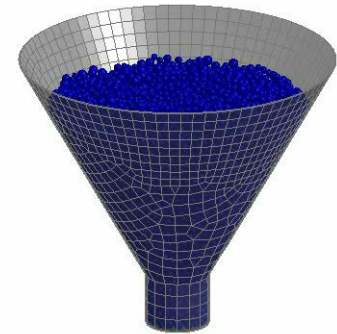
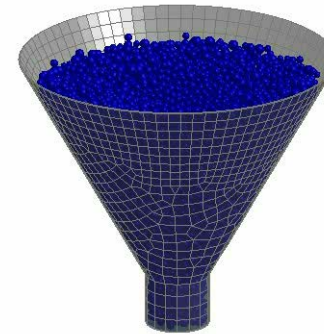
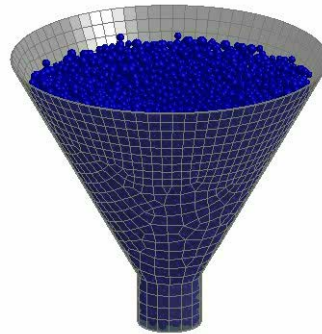
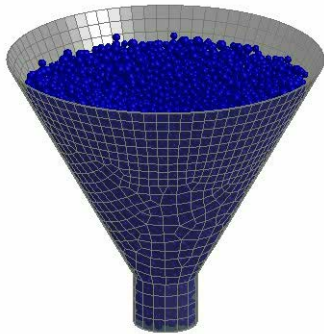
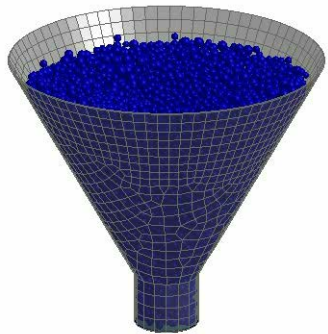
foamed clay

dry sand

wet sand

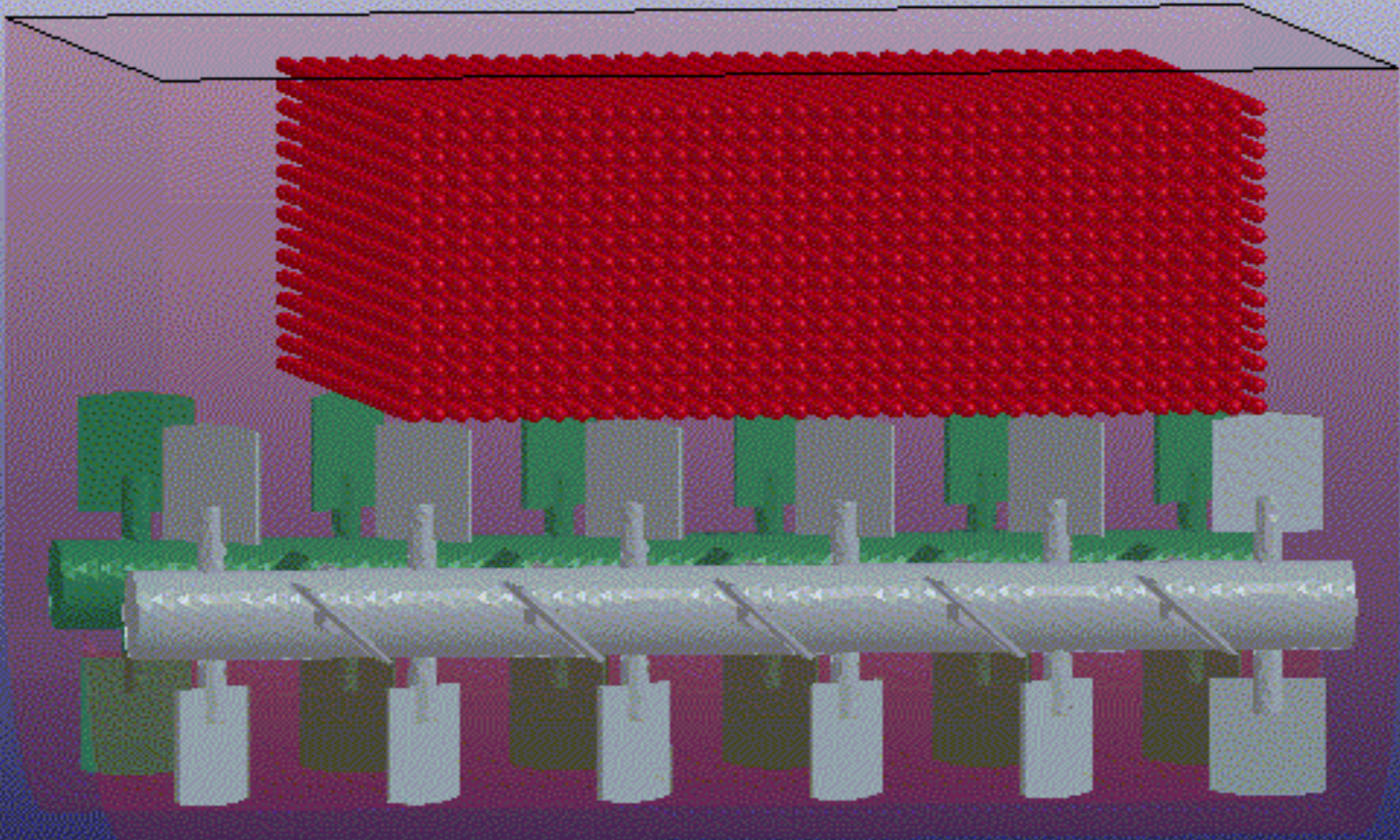
fresh concrete

“water”



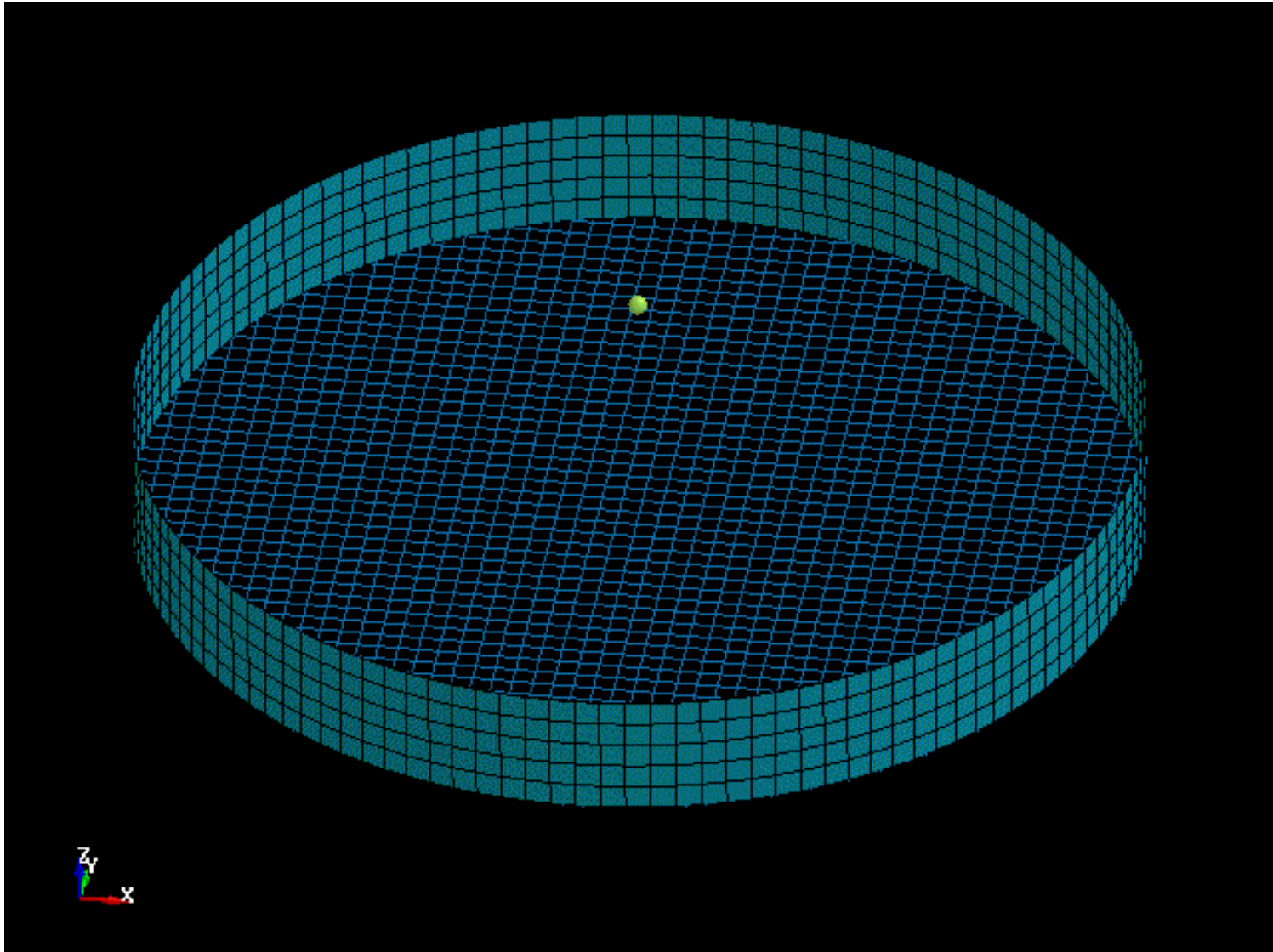
# DEM Mixer

Mixer 9.6L (kg-m-s)  
Time = 0



# Node to Beam Coupling

```
*DEFINE_DE_TO_BEAM_COUPLING
```



# Node to Surface Coupling

\*DEFINE\_DE\_TO\_SURFACE\_COUPLING

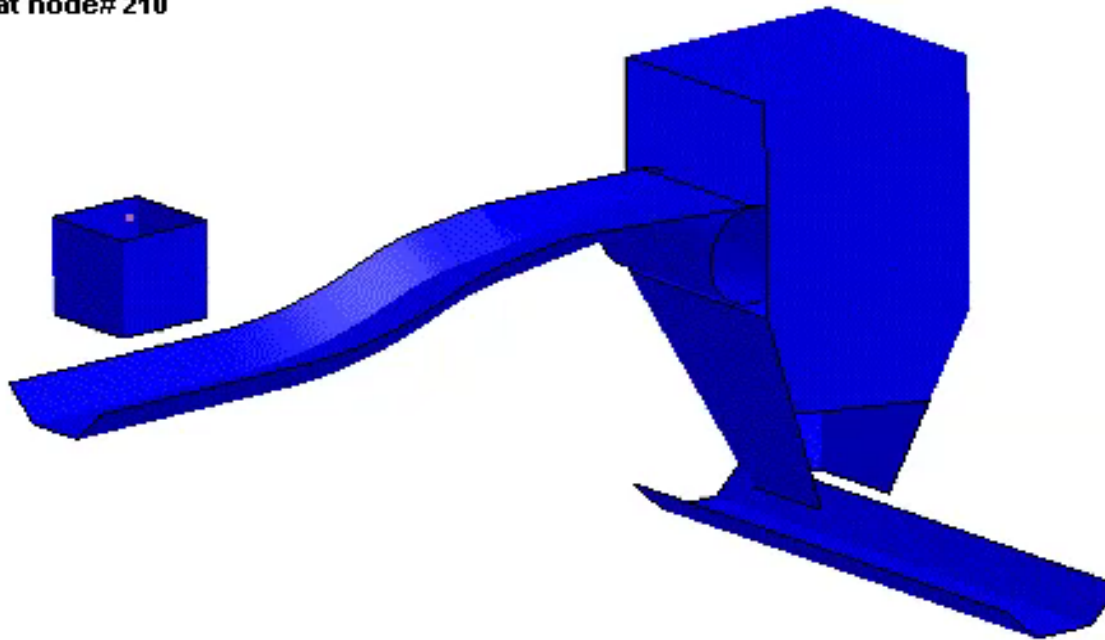
2 Conveyors

Time = 0

Contours of Resultant Velocity

min=0, at node# 210

max=0, at node# 210



Fringe Levels

6.000e+00

5.800e+00

5.600e+00

5.400e+00

5.200e+00

5.000e+00

4.800e+00

4.600e+00

4.400e+00

4.200e+00

4.000e+00

3.800e+00

3.600e+00

3.400e+00

3.200e+00

3.000e+00

2.800e+00

2.600e+00

2.400e+00

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1.600e+00

1.400e+00

1.200e+00

1.000e+00

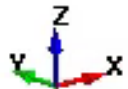
8.000e-01

6.000e-01

4.000e-01

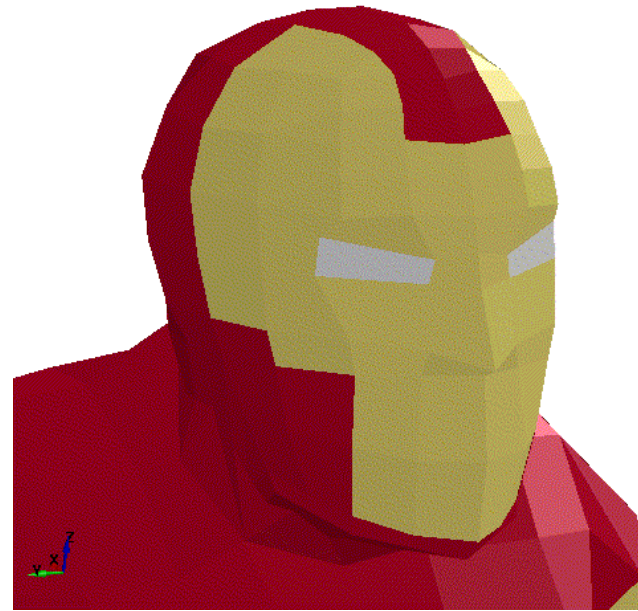
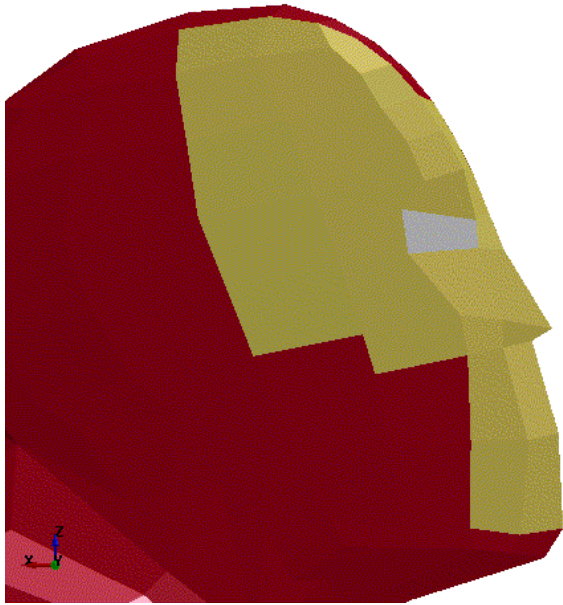
2.000e-01

0.000e+00



# Tied Node to Surface Coupling

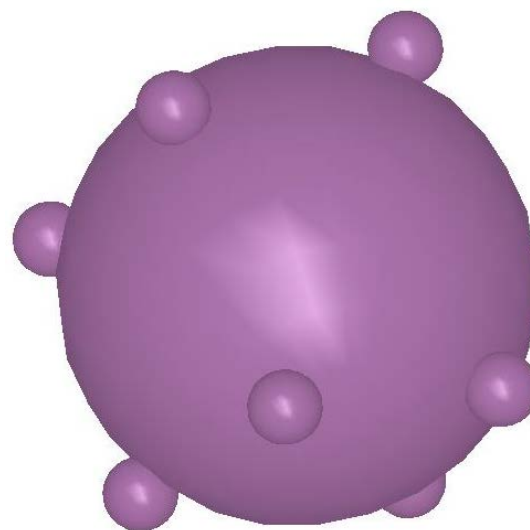
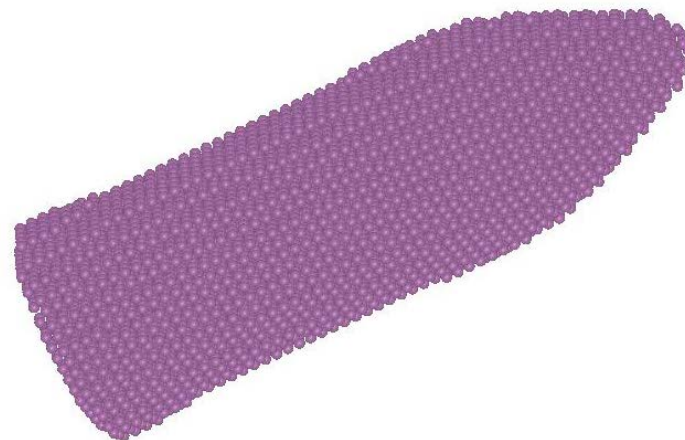
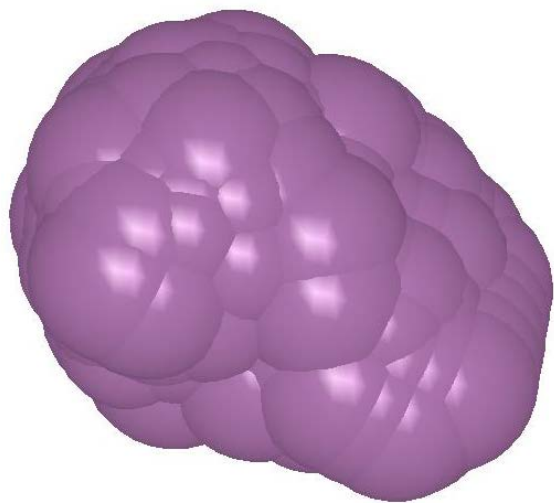
```
*DEFINE_DE_TO_SURFACE_TIED
```



**Throwing a pie in the face, Courtesy of Kazuya, Lancemore**



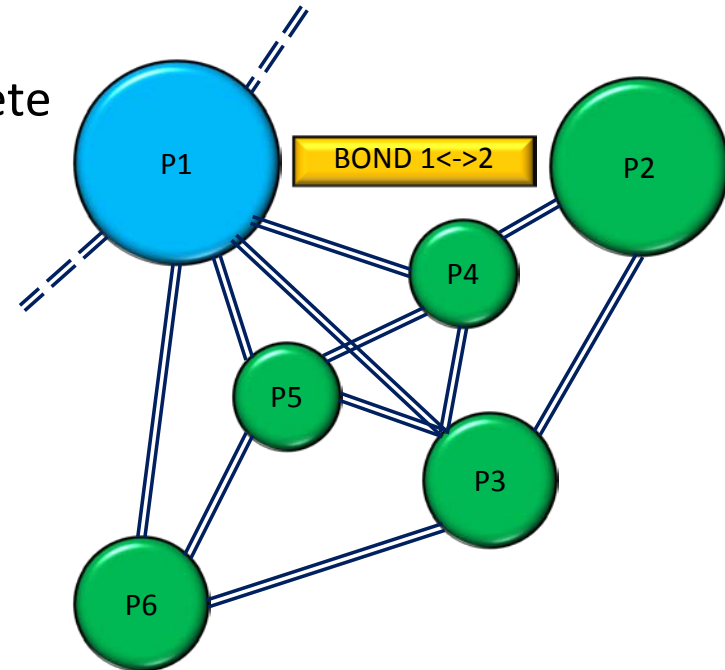
# Other Shapes



# Discrete Element Sphere with Bond Model

## Extending into Continuum Mechanics

- All particles are linked to their neighboring particles through Bonds.
- The properties of the bonds represent the complete mechanical behavior of Solid Mechanics.
- The bonds are independent from the DES model.
- They are calculated from Bulk Modulus and Shear Modulus of materials.
- Contact is disabled between bonded pair

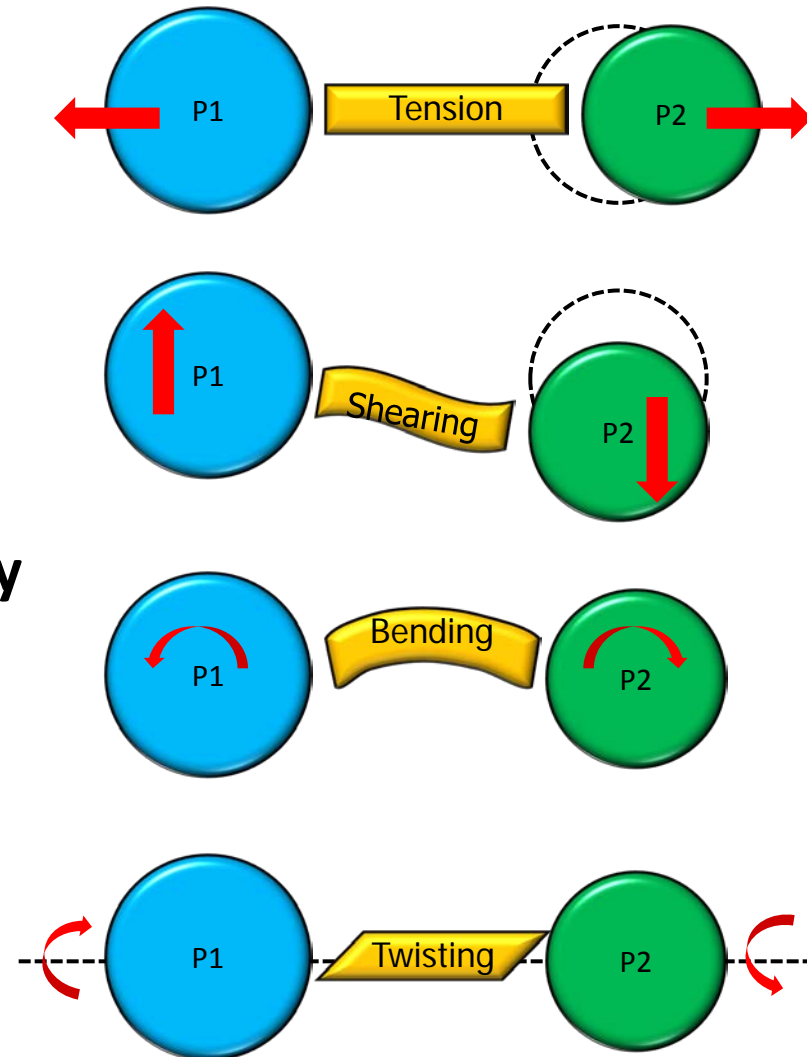


# DEM Bonds

- DE Bond Type I
  - Simple links, truss or beam, etc...
  - Extended Peridynamics
- DE Bond Type II
  - Heterogeneous links to model continuum mechanics
  - Extended features and will use regular \*MAT properties

# DE Bond TYPE I

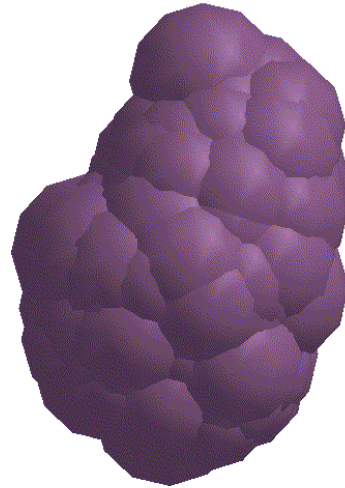
- **Every bond is subjected to:**
  - *Stretching*
  - *Shearing*
  - *Bending*
  - *Twisting*
- **The breakage of a bond results in Micro-Damage which is controlled by the critical fracture energy value  $J_{IC}$ .**



# DEM Bond TYPE I

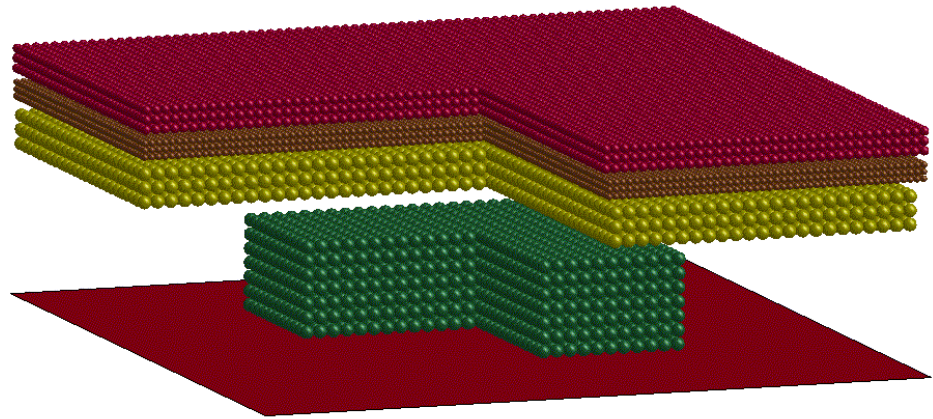
- Form different shapes of particles using DEM

LS-DYNA keyword deck by LS-PrePost  
Time = 0



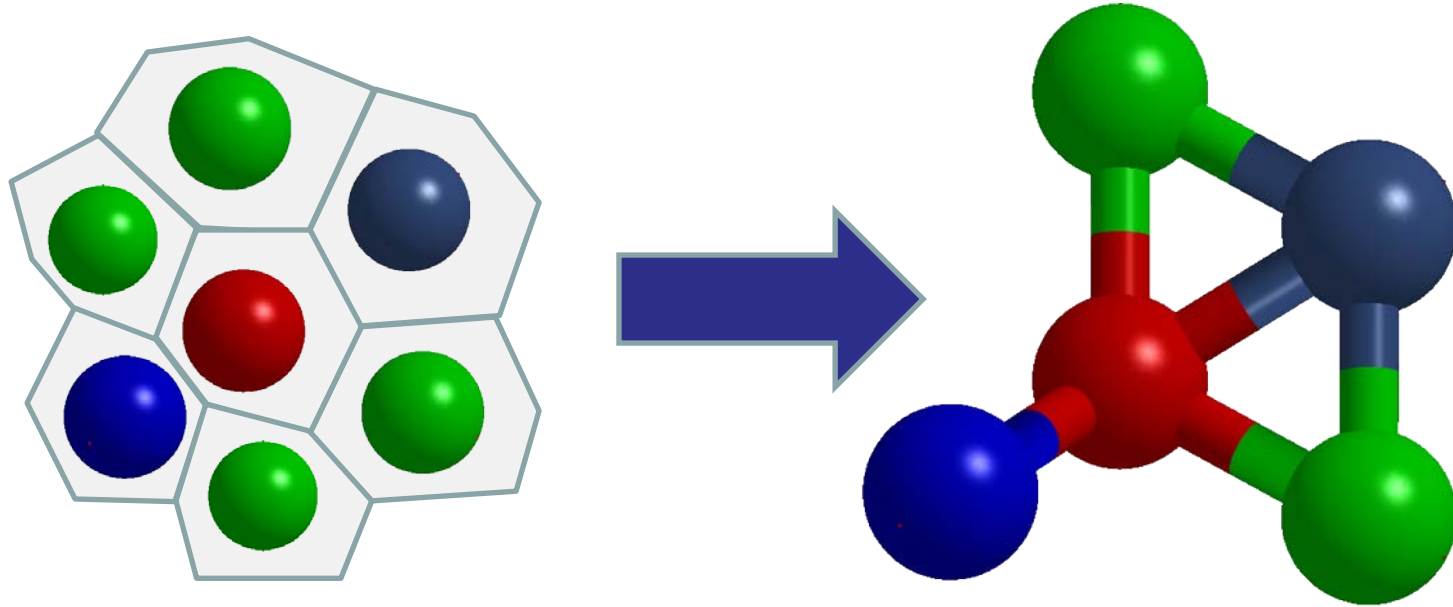
# DEM Bond TYPE I

DE Paste (microgm-micron-sec)  
Time = 0



# DEM TYPE II

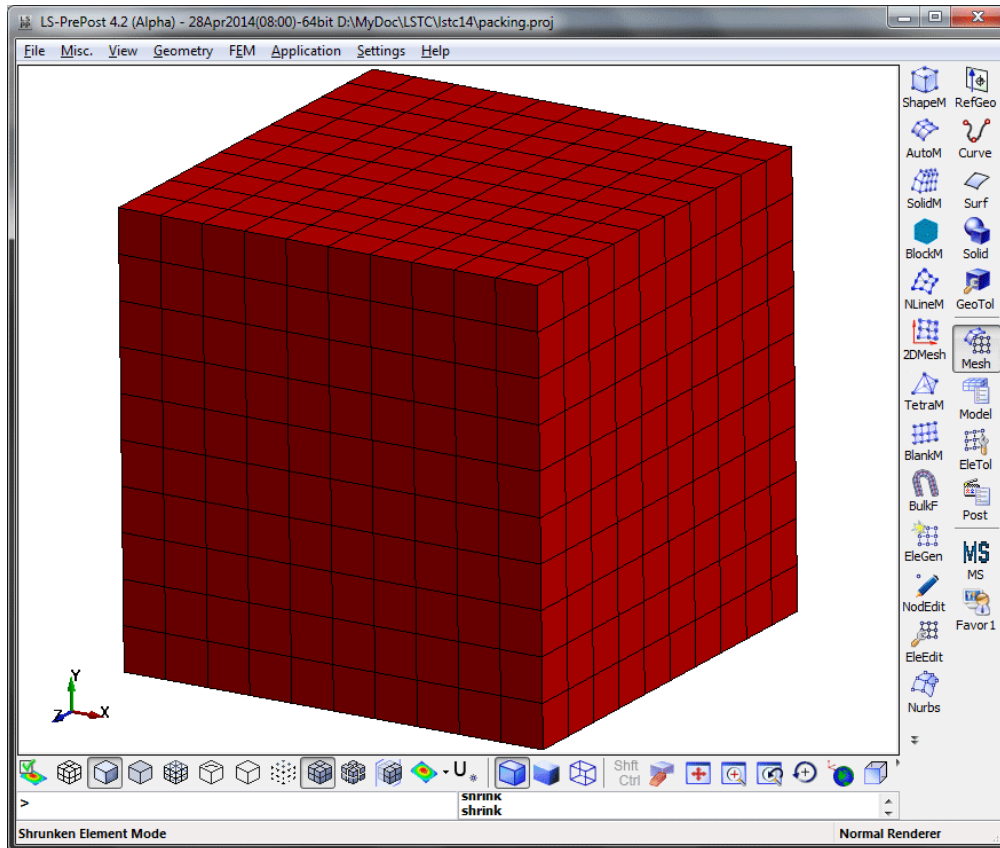
## Heterogeneous BOND(HBOND) Continuum Particle Method



- **\*DEFINE\_DE\_HBOND** connects two spheres with a heterogeneous bond.
- **\*MAT** properties are used to determine the stiffness of the bonds automatically.
- Strains, stresses, and history variables are computed for each particle independently.
- **\*INTERFACE\_DE\_HBOND** specifies various damage/failure models.
- Self contact will be activated for broken bonds.

# One Particle Method

From “Continuum” to “Discrete”

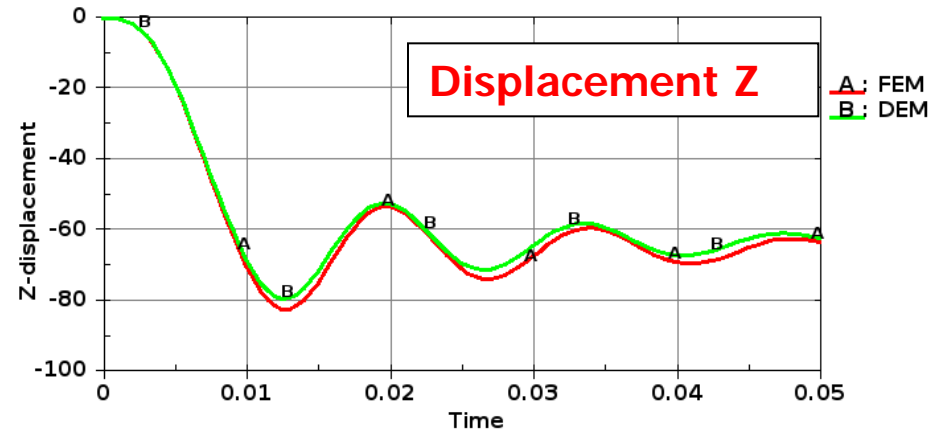
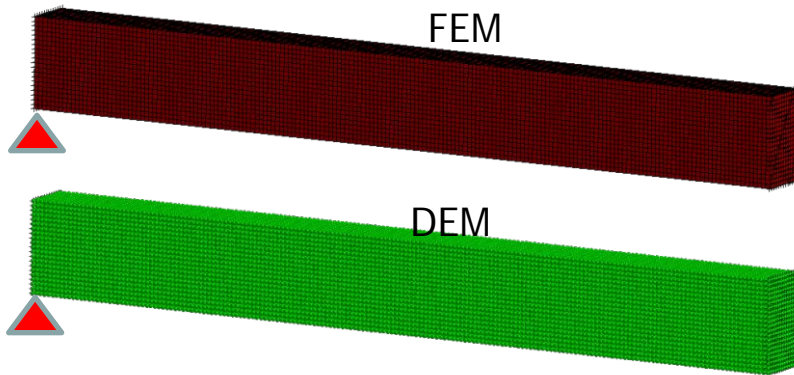


- One model setup
- One solver
- Same material models
- Multi-physics:
  - Continuum Mechanics*
  - Damage Mechanics*
  - Fracture Mechanics*
  - Discrete Mechanics*
- Built-in self contact
- Coupling with other FEM and particle methods

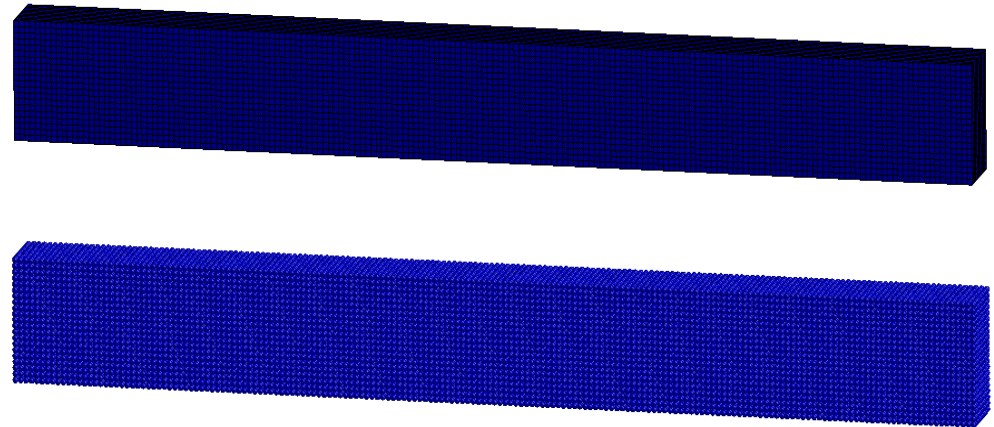


# HBOND Verification

## A simply-supported beam under a body force



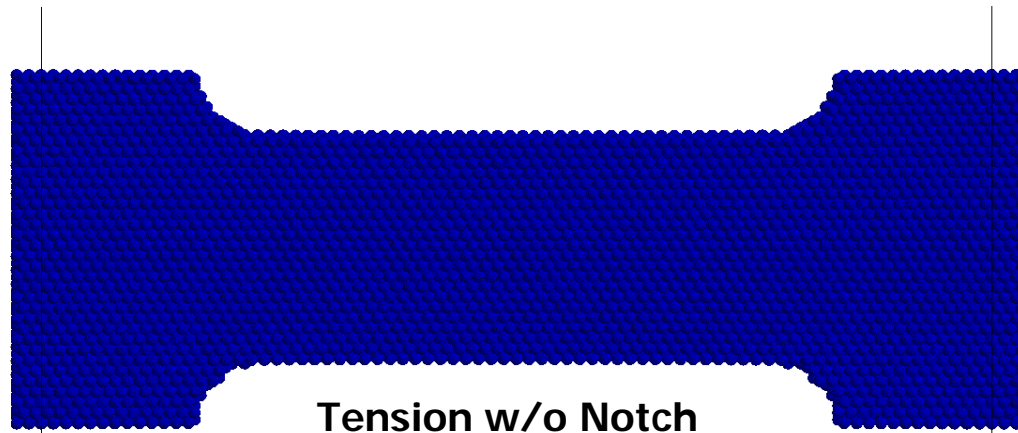
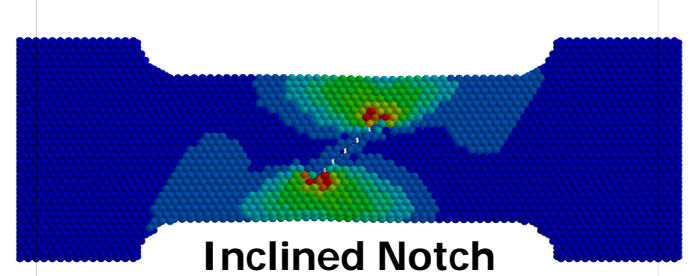
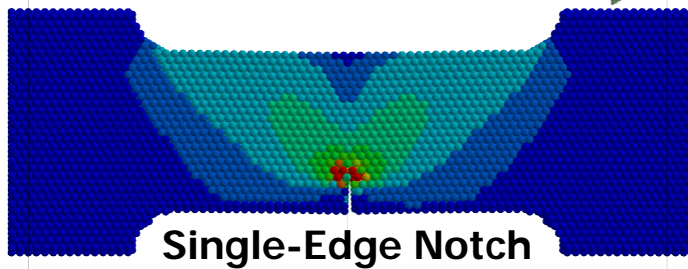
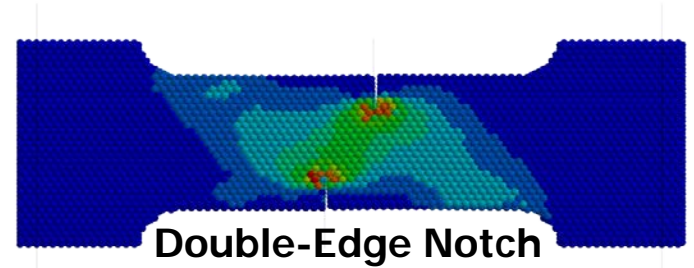
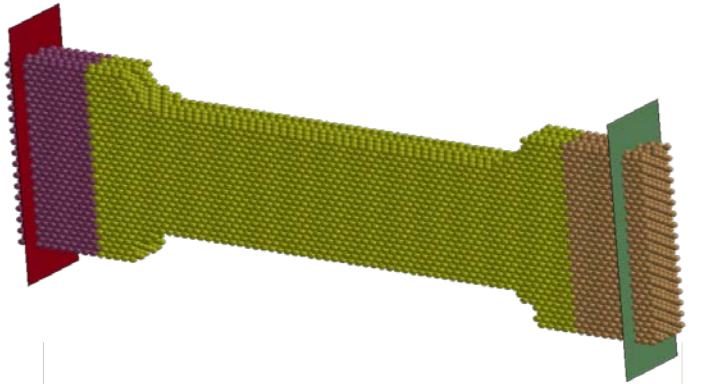
- FEM & DEM models are created for one half beam with the symmetric boundary conditions in the middle.
- The displacements & stresses obtained by the DEM are very close to those by the FEM.
- No boundary effects.



Tension Stress X

# HBOND Verification

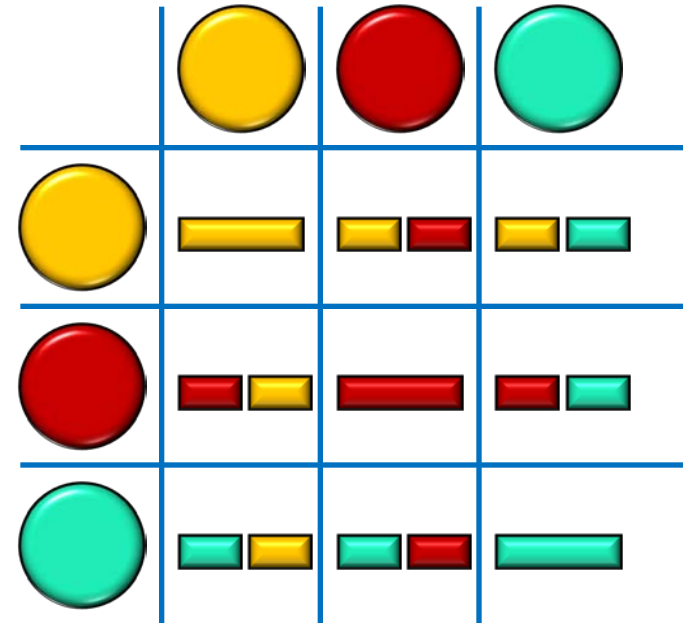
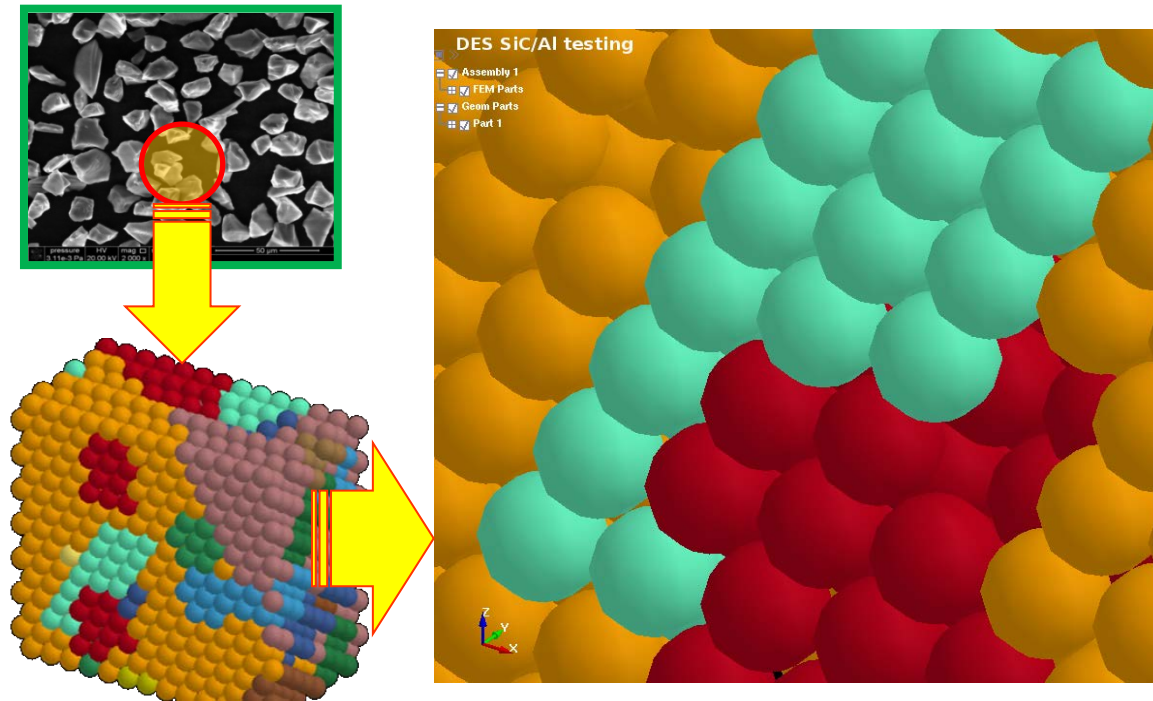
## Specimen under Tension



# HBOND Micro-Mechanics

\***DEFINE\_DE\_HBOND** creates a heterogeneous bond between different spheres.

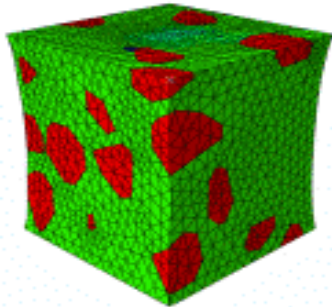
\***INTERFACE\_DE\_HBOND** defines various damage/failure models for the heterogeneous bonds based on the material properties of the connecting particles.



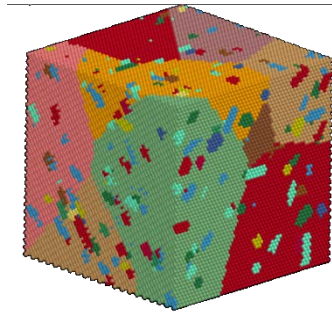
**Various heterogeneous bonds**

# SiC/Al Metal Matrix Composite

## DEM for Material Design



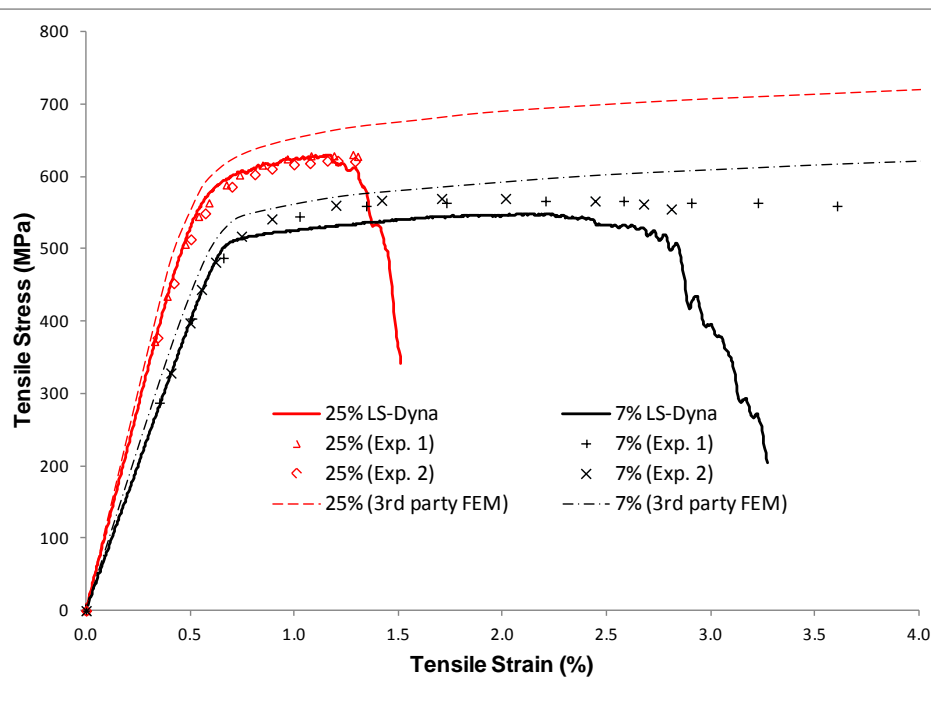
3<sup>rd</sup> party FEM



LS-DYNA DEM

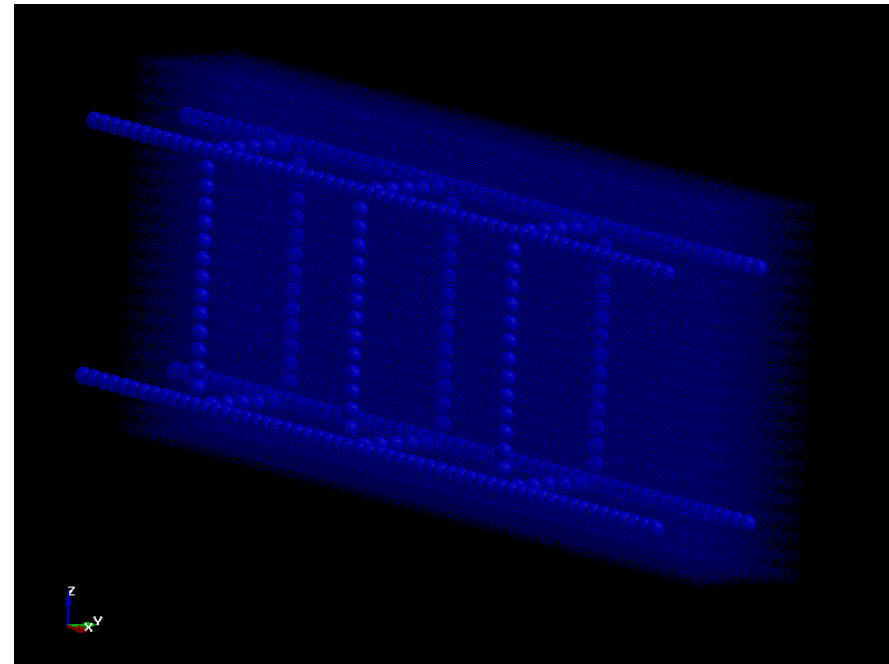
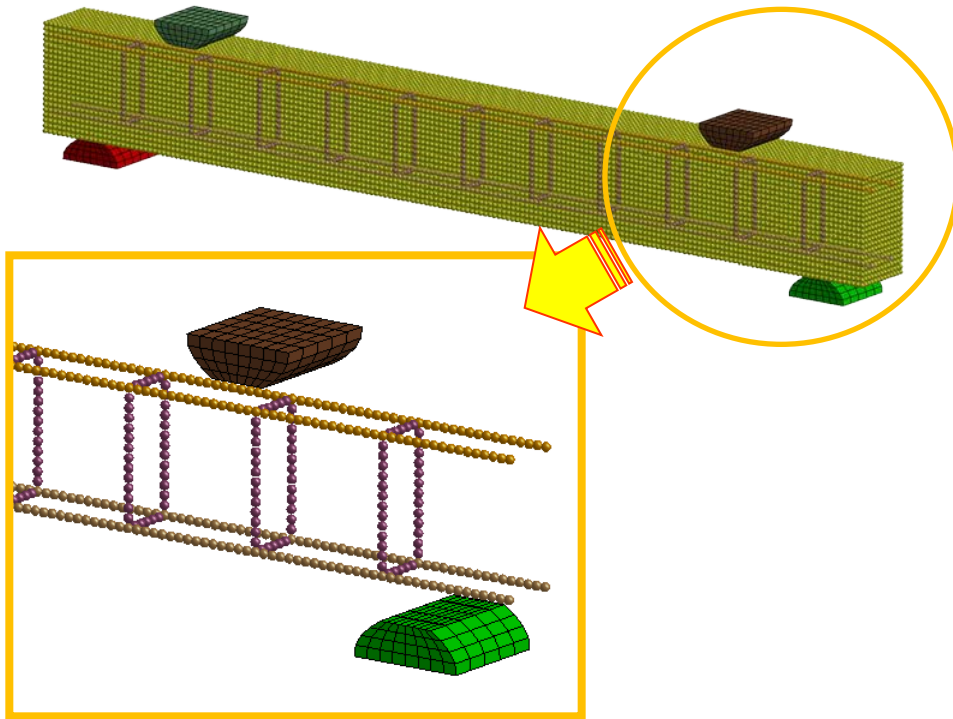
Material Properties	Al	SiC
Density: [kg/m <sup>3</sup> ]	2,700	3,100
Young's modulus [GPa]	71.7	427
Poisson's ratio:	0.33	0.17
Failure Energy Rate: [kJ/m]	40	15
Average Particle size: [um]	-	13

LS-DYNA Results	7% vol		25% vol	
	Exp.	Num.	Exp.	Num.
Young's modulus [GPa]	84.2	80.6	113.3	113.5
Tensile strength [MPa]	568.6	545.	623.	641.
Limit strain	1.8%	1.9%	1.2%	1.4%



# HBOND

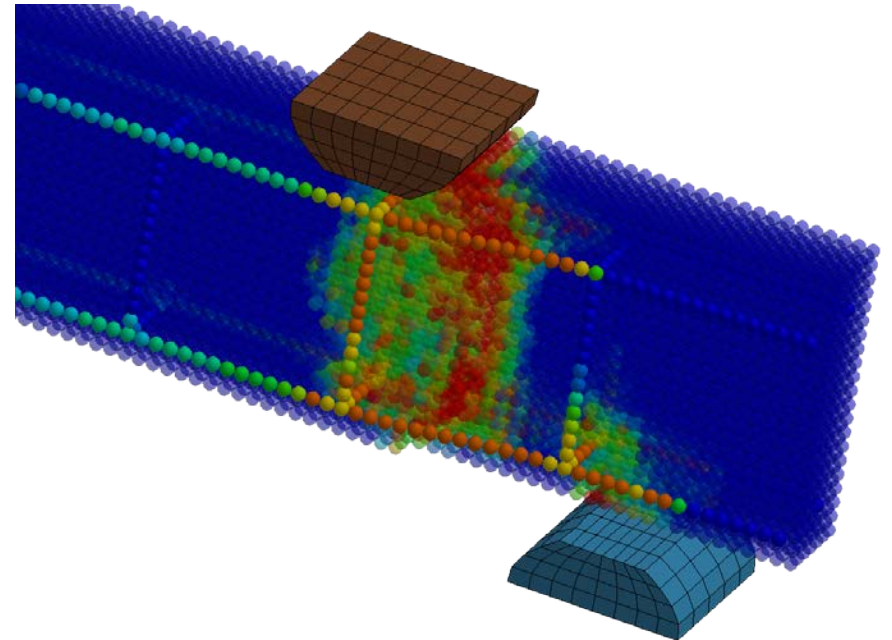
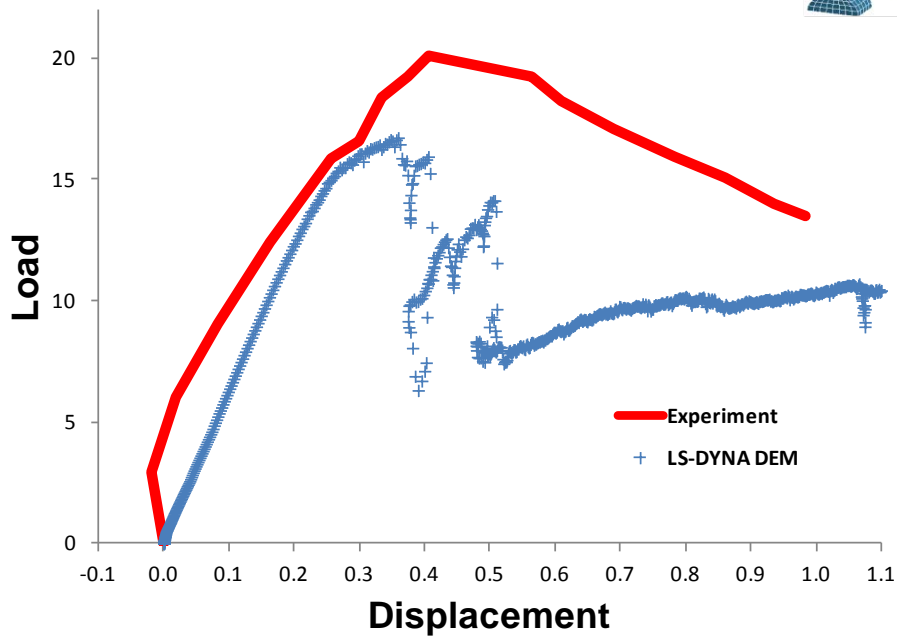
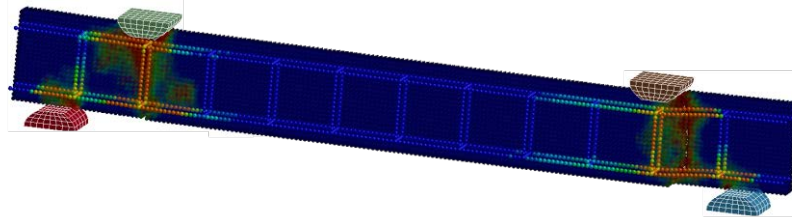
## A Reinforced Bar under Four-Point Bending



- \***DEFINE\_DE\_HBOND** bonds all parts.
- \***INTERFACE\_DE\_HBOND** defines different de-bonding criteria between parts

**De-bonding process**

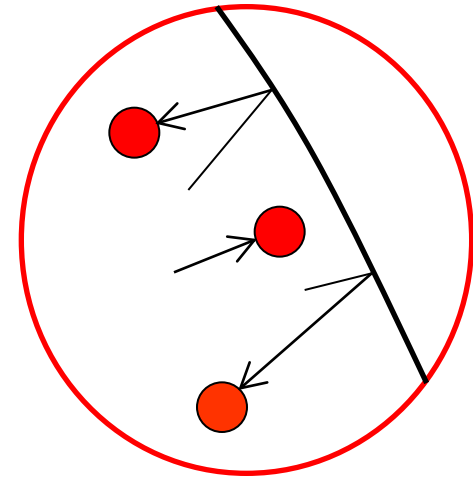
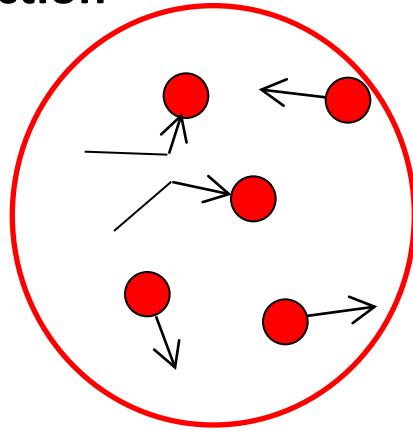
# Comparison between DEM & Experimental Results



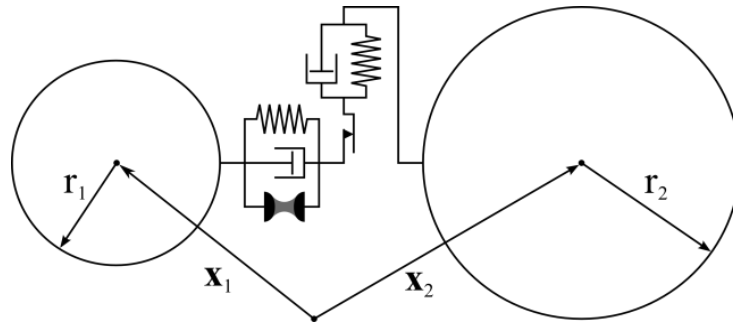
# Coupled Multi-Physics Solvers

## Particle-particle interaction

Particle Gas

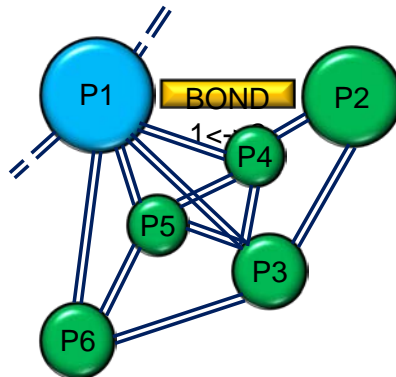


DEM



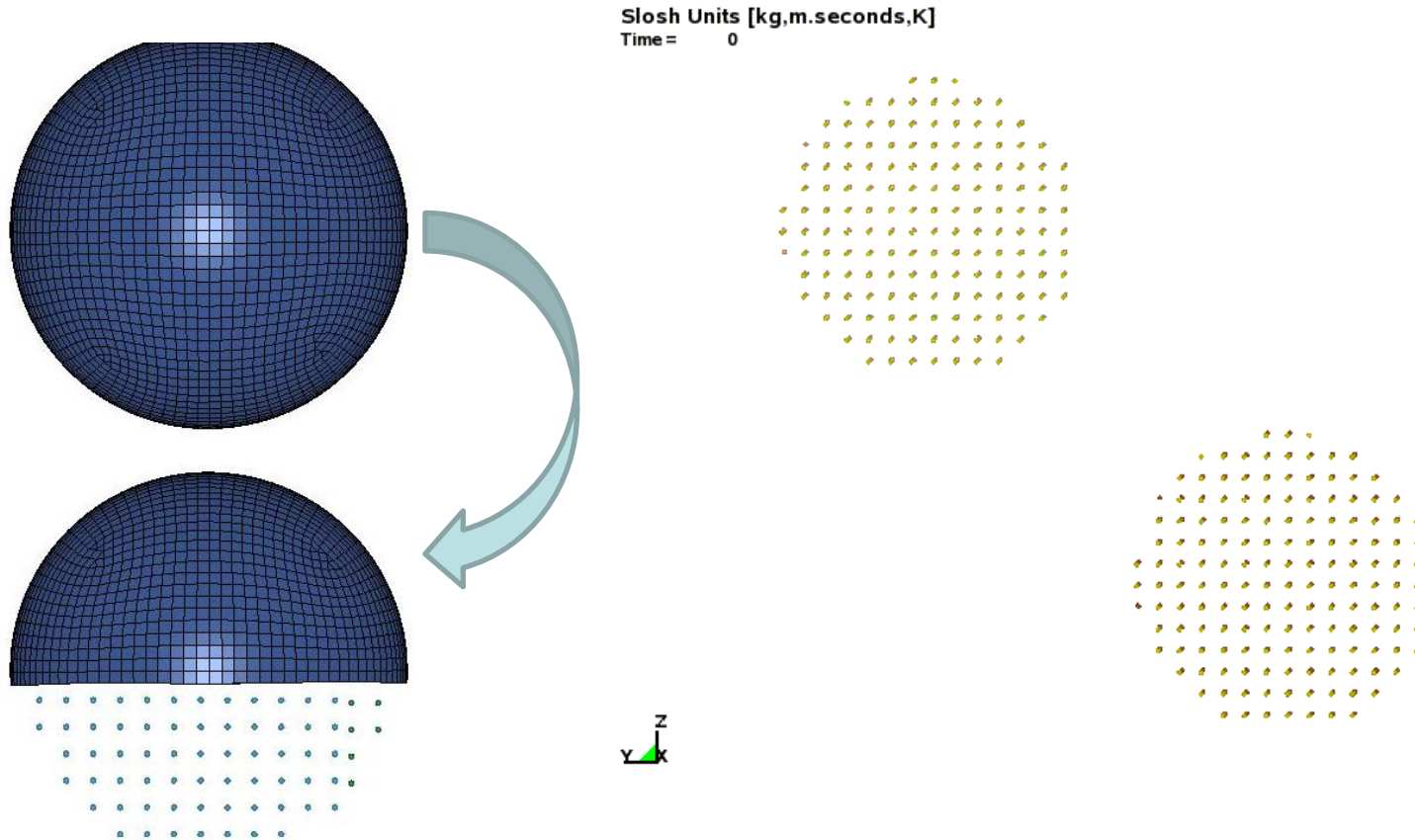
- 1) Node to Node
- 2) Node to Beam
- 3) Node to Segment
- 4) Node to Volume

DEM bond



# Node to Node Coupling

## SPH to SPH Contact

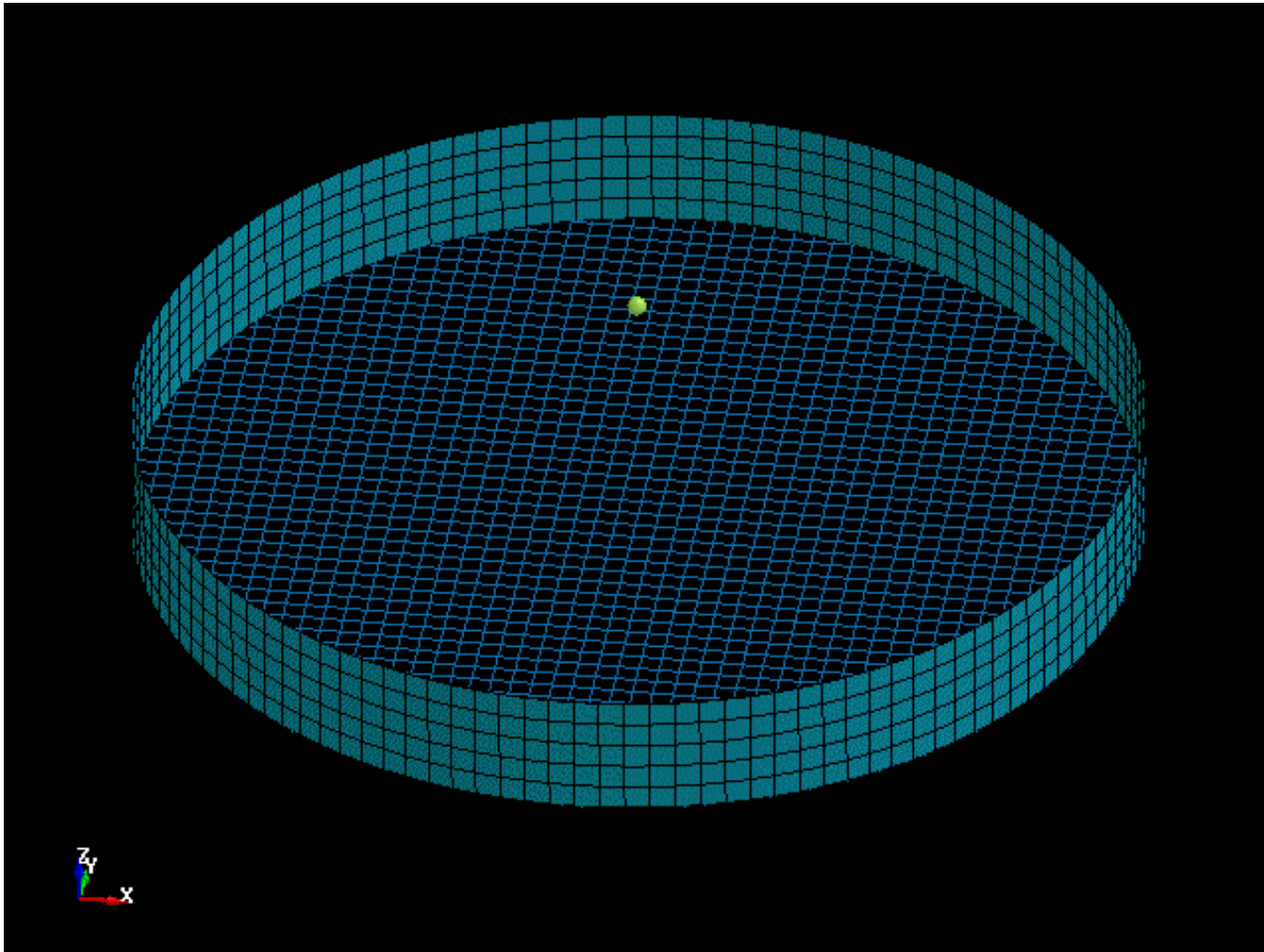


**Tank sloshing with fluid and vapor (node to node contact)**  
**Density ratio ~ 1000**



# Node to Beam Coupling

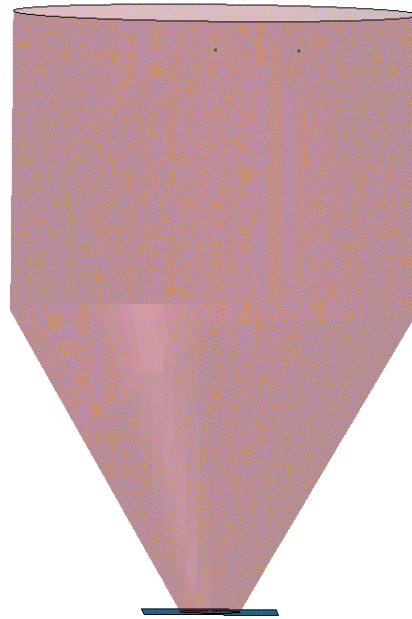
## DES to Beam Contact



# Node to Surface Coupling

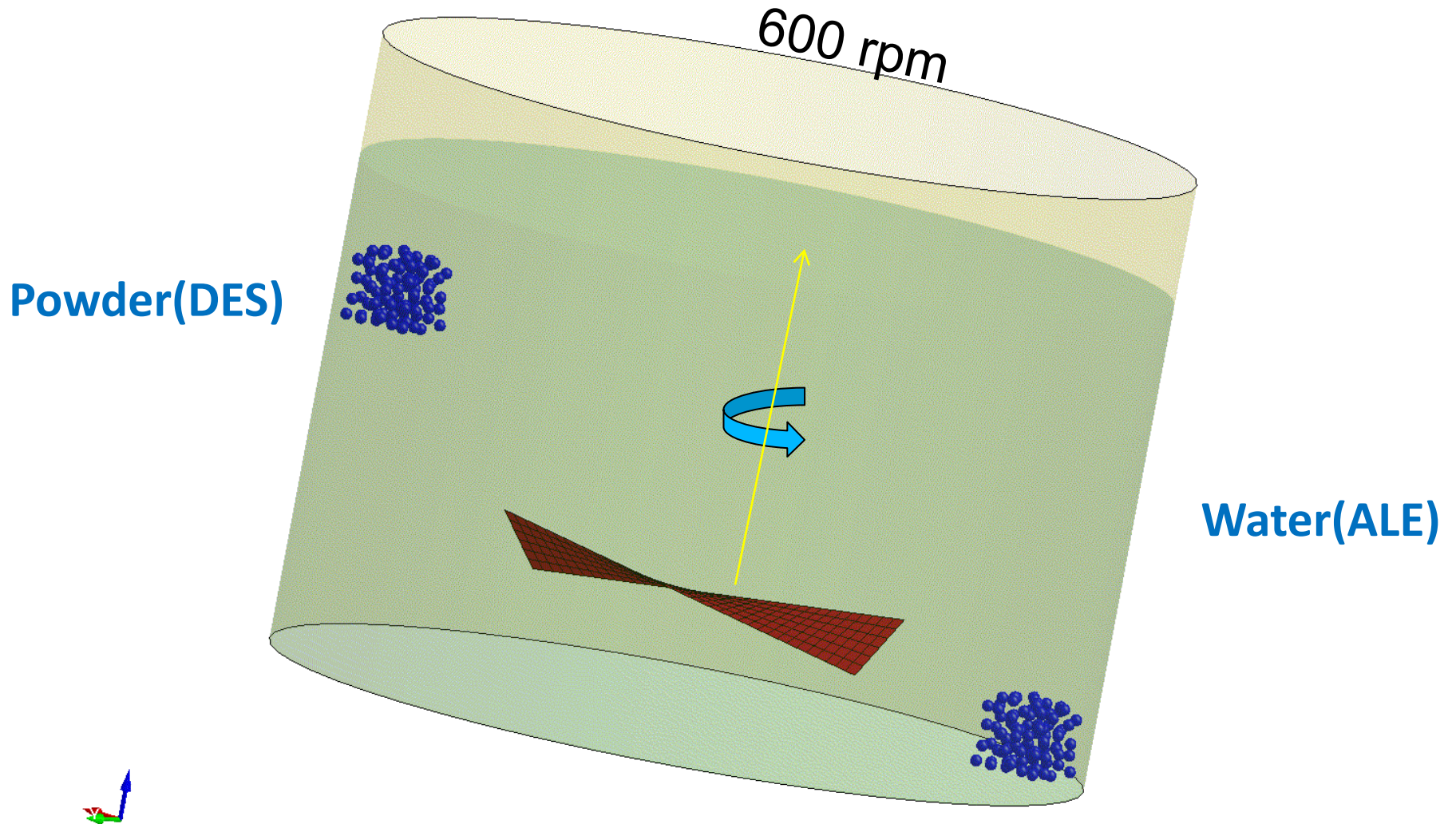
## DES to Segment Contact

Particles (kg-m-s)  
Time = 0



# Node to Volume Coupling

## DES to ALE Contact



# Modal Dynamics

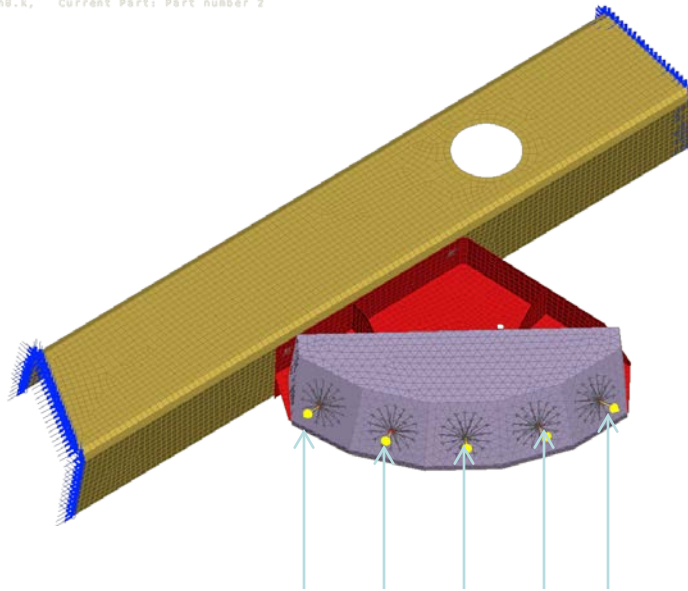
# Modal Dynamics

- Modal Dynamics in LS-DYNA has been enhanced in LS971 R71 with new features and keywords.
  - \*CONTROL\_IMPLICIT\_MODAL\_DYNAMICS
    - To activate modal dynamic analysis
  - \*CONTROL\_IMPLICIT\_MODAL\_DYNAMICS\_MODE
    - To select a subset of modes to use in the analysis
  - \*CONTROL\_IMPLICIT\_MODAL\_DYNAMICS\_DAMPING
    - To specify damping coefficients for modal damping.
- Good for model analysis
  - with multiple load cases
  - where the response is well captured with a small number of modes
  - linear behavior in the elements and materials

# Modal Dynamics Example

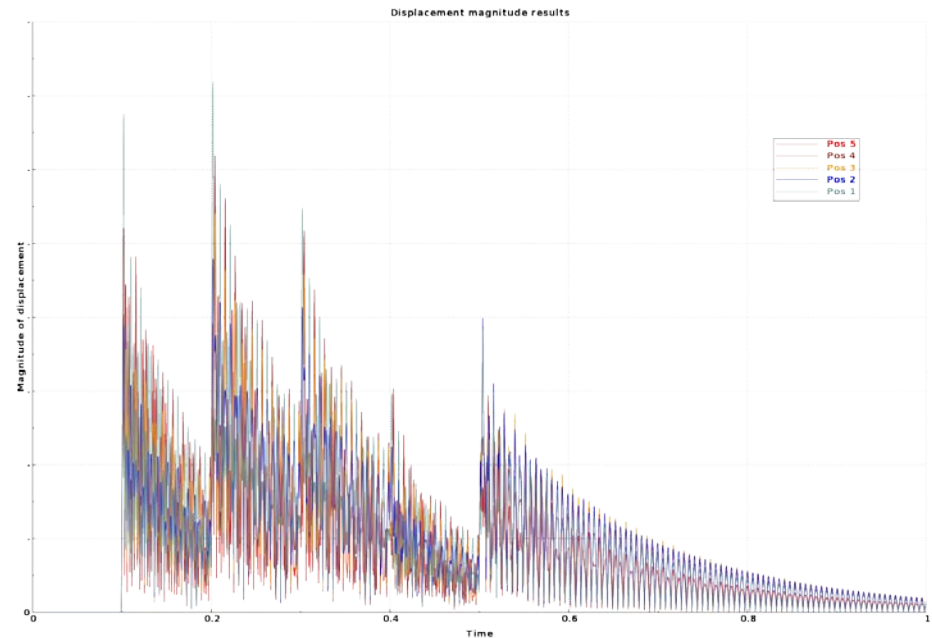
- Instrument Bracket from BAE Systems Hägglunds

main.k, Current Part: Part number 2



5 Impulse forces

Response for 5 impulse forces



# Modal Dynamics Example

- An instrument bracket is subjected to impulsive loading.
- Force spikes are applied sequentially to five points.
- Response is computed using Transient modal dynamics.
- 30 eigenmodes extracted up to 1370 Hz.
- Constant modal damping used, 1 % of critical.
- 4 CPU cores used in all cases.
- Transient Explicit: 3h 39 min (single precision)
- Transient non-linear implicit: 4 h 26 min
- Transient modal dynamics: 3 min 33 sec -- 60 Times faster!

LS-TaSC



# LS-TaSC

- Topology and Shape Computations using LS-DYNA.
- LS-TaSC goals:
- Solving for large, highly nonlinear problems. The methodology is valid for linear.
- Imbedded in the design process

# Version 3.0

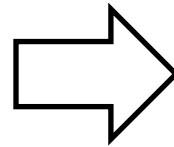
- Major New Features
- Free surface design of solids
  - Geometry definitions
  - Automatic mesh smoothing
- Integration with LS-PrePost
- Several minor features.

# Topology design of a bonnet

Multi-criteria problem: One design for three nonlinear load cases, each with constraints.

During the design process, the algorithm will remove 90%-99% of the structure. Mesh refinement is therefore required.

Example courtesy of JRL.

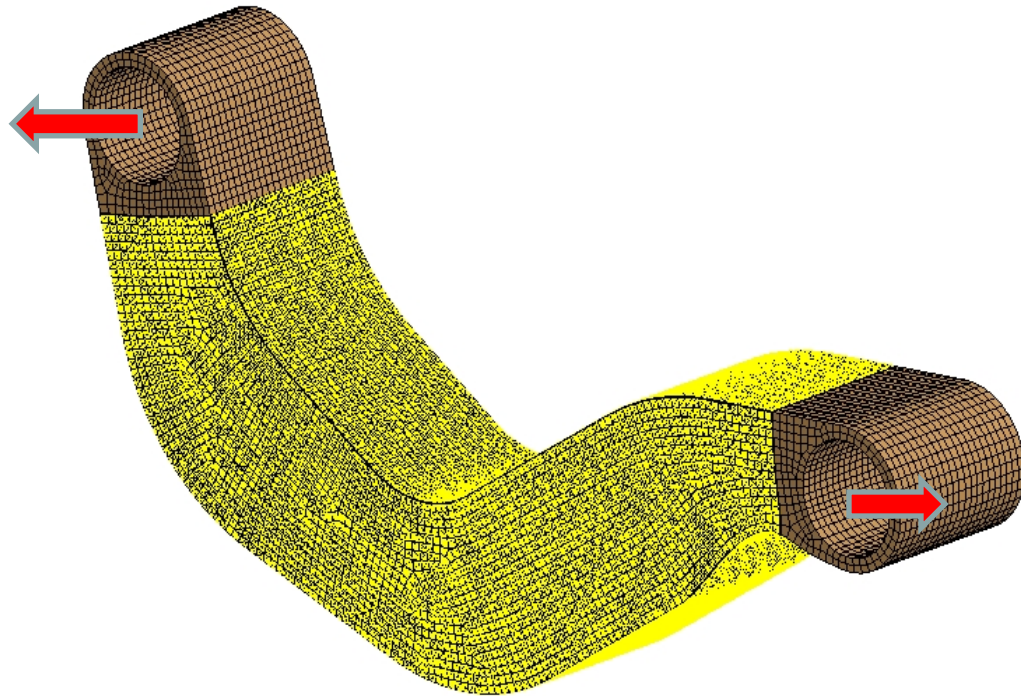


# Free surface design

- Redesign of the surface of solids to have a uniform stress (removing the stress concentrations).
- Geometry definitions are allowed.
- Very quick to set up the design problem.

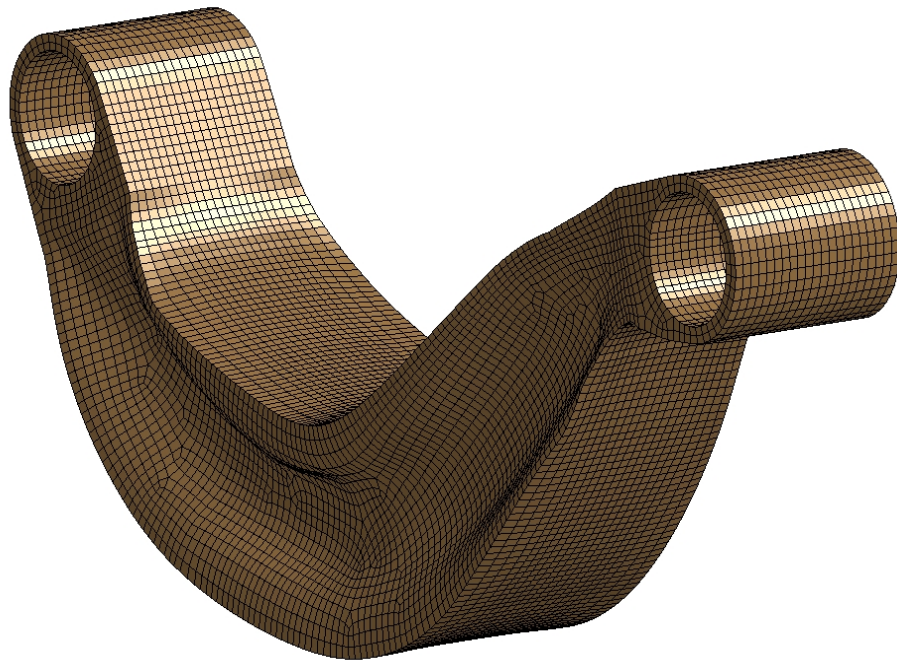
# Component surface design

- Component example with the yellow surface selected for redesign. Setting up a problem consists of defining a surface.

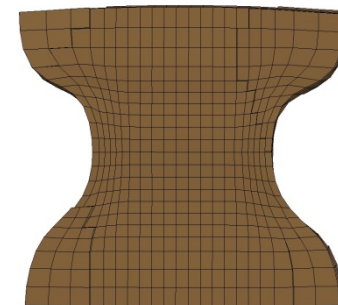


# Component shape design

- Design at iteration 16 shows the flanges that are created.

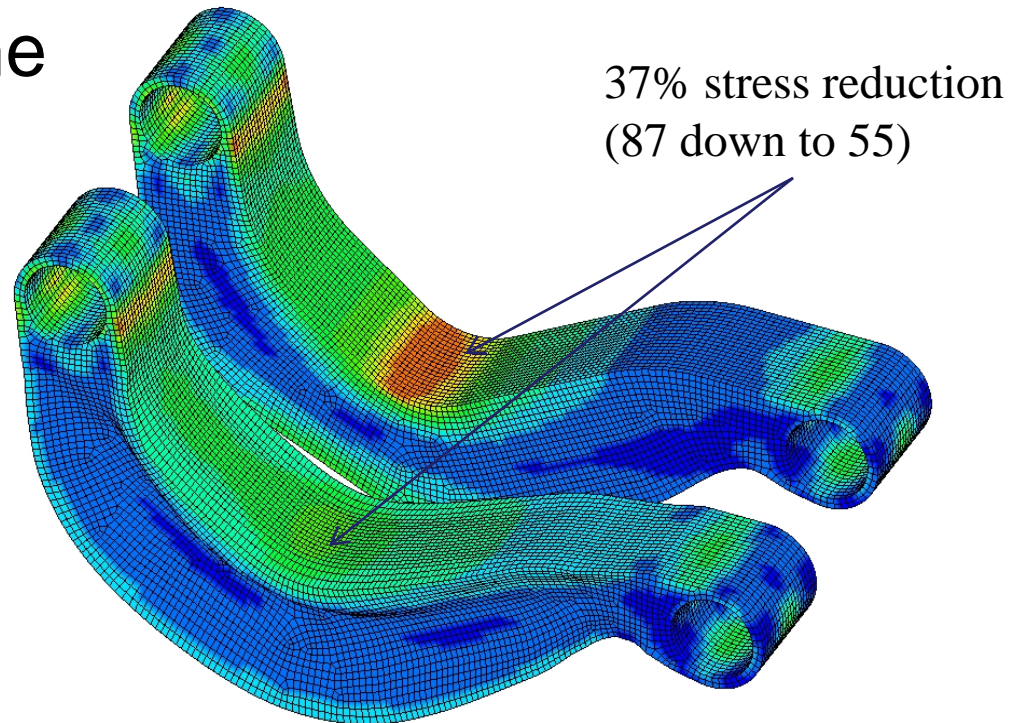


Cross-section

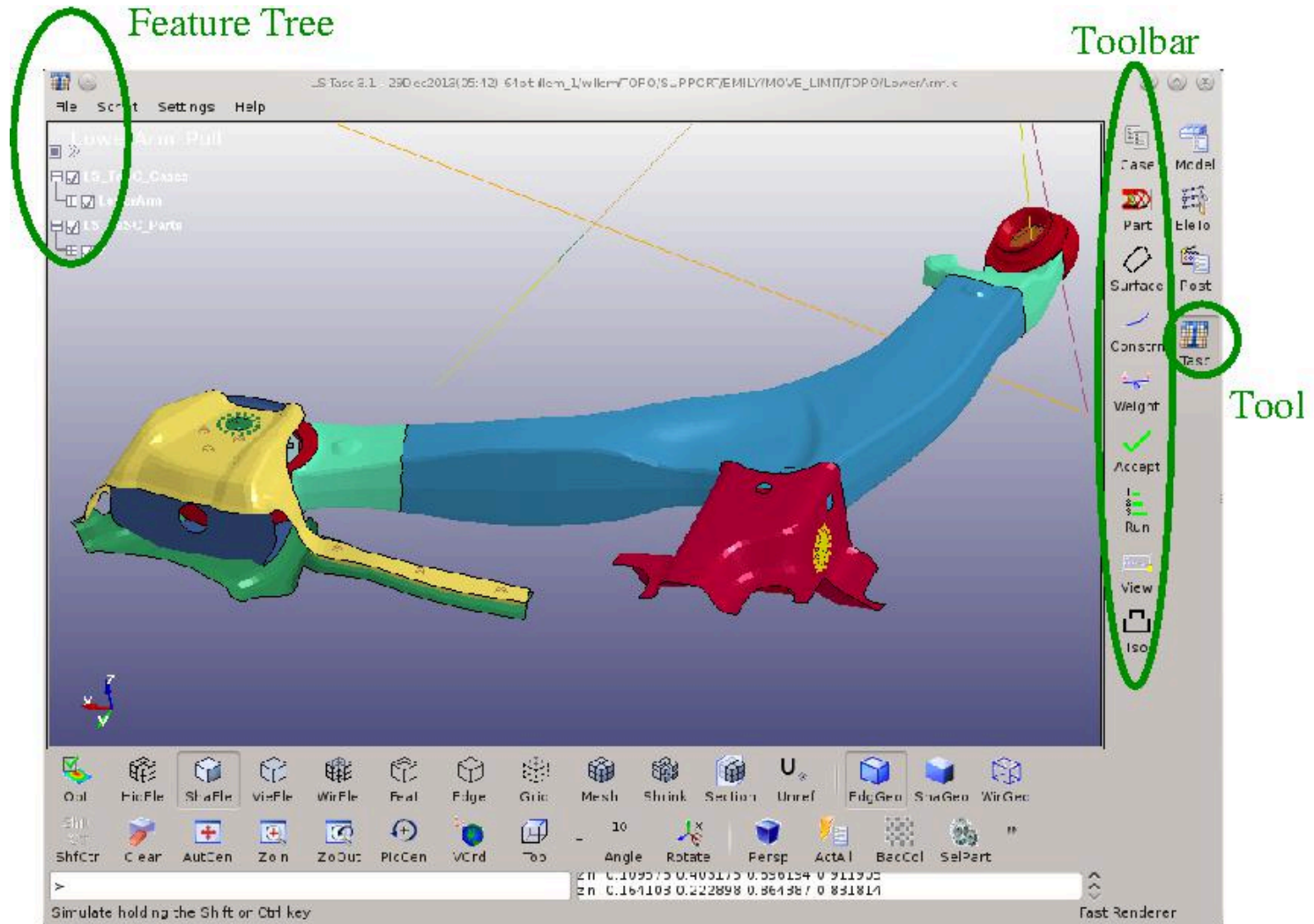


# Component shape design

- Comparing the Von Mises stress at iteration 16 with the original design shows a 37% improvement.



# LS-PrePost Integration





# LS-PrePost ICFD Capabilities

# ICFD Post Processor Introduction

- Since the official release of the ICFD solver in the R7.0 version, developments have been continuous and the number of users has been steadily growing.
- Currently, LS-PrePost offers some tools in order to post treat the results from the ICFD solver based on its solid mechanic counterpart.
- However, the requirements for CFD post treatment are often quite different and challenging. This meant that a radically new approach was needed for LS-PrePost to meet those specific requirements.
- LS-PrePost 4.2 will be the first version to incorporate post treatments specific to the ICFD Solver and to CFD solvers in general.

# ICFD Post Processor Introduction

## Object oriented structure:

The screenshot displays the ANSA ICFD software interface. The main window shows a 3D model of a car with a green body and blue wheels. The top-left corner displays "ANSA Car" and "Time = 1.9823". The top menu bar includes "File", "Misc.", "View", "Geometry", "FEM", "Application", "Settings", and "Help".

The **Object Tree** panel on the right lists several objects, all of which are checked (ticked):

- MS\_SHELL\_1
- MS\_SHELL\_2
- MS\_SHELL\_3
- MS\_SHELL\_4
- MS\_SHELL\_5
- MS\_SHELL\_6
- MS\_SOLID\_10

Below the object tree are buttons for "All", "None", and "Reverse".

The **Object Properties** panel shows the following data:

Number of Elements	1692524
Extend	
▶ Minimum	3206.210205; -3; 479.1
▶ Maximum	3218.210205; 0.00013

The **Display Options** panel includes the following settings:

Mode	Shade
Color by	PID
Fringe Contour	<input type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	0
AVG Scalar	0
Max Scalar	0

The **Animate** window at the bottom shows the following settings:

Eigen First: 1 Last: 58 Inc: 1 Time: 1.9823 State: 58

Animate  Loop

30S F

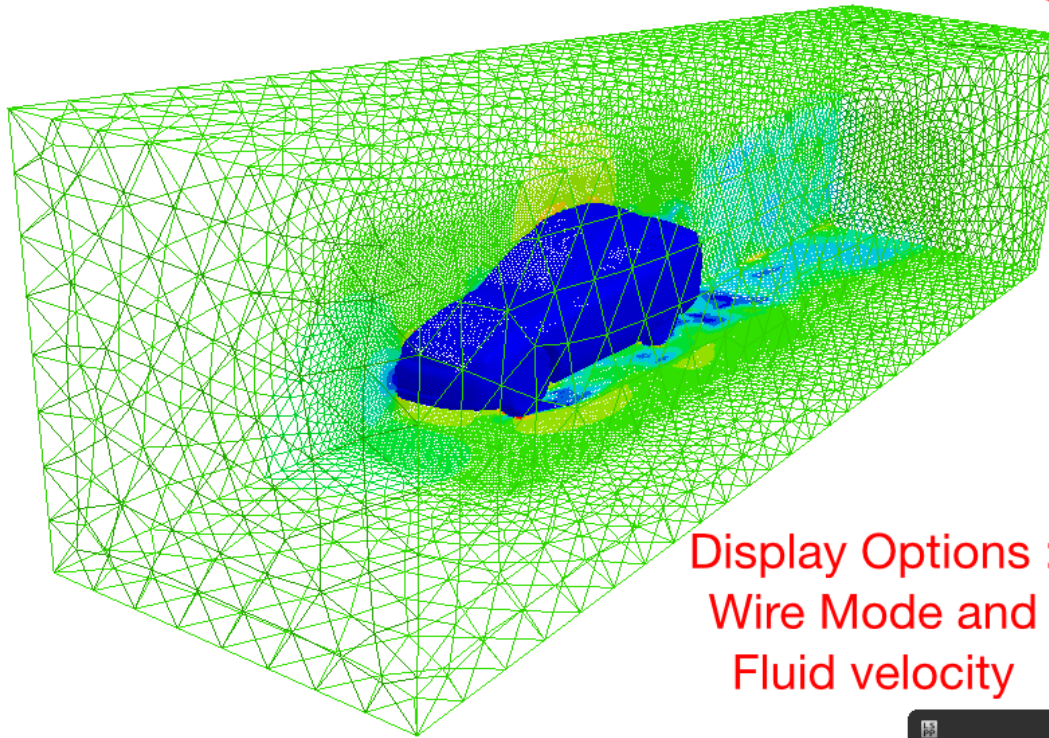
Annotations in red text with arrows point to the "Object Tree" and "Display Options" panels. The "ICFD" and "MS" icons in the right-hand toolbar are also circled in red.

# ICFD Post Processor Introduction

## Object oriented structure:

ANSA Car  
Time = 1.9823

Only Fluid Volume displayed



Display Options :  
Wire Mode and  
Fluid velocity

The screenshot shows the ICFD software interface. The 'Objects' list on the left contains several 'MS\_SHELL' objects and one 'MS\_SOLID\_10' object, which is selected. The 'Object Properties' panel shows the number of elements (1692524) and the extent of the object. The 'Display Options' panel shows the mode set to 'Wire', color by 'Fluid velocity', and 'Fringe Contour' and 'Fringe Legend' checked.

Objects	
<input type="checkbox"/>	MS_SHELL_1
<input type="checkbox"/>	MS_SHELL_2
<input type="checkbox"/>	MS_SHELL_3
<input type="checkbox"/>	MS_SHELL_4
<input type="checkbox"/>	MS_SHELL_5
<input type="checkbox"/>	MS_SHELL_6
<input checked="" type="checkbox"/>	MS_SOLID_10

All None Reverse

Object Properties	
Number of Elements	1692524
▼ Extend	
▶ Minimum	3206.210205; -3; 479.4
▶ Maximum	3218.210205; 0.00013

Display Options	
Mode	Wire
Color by	Fluid velocity
Fringe Contour	<input checked="" type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	0
AVG Scalar	57.333698
Max Scalar	114.667397



The screenshot shows the 'Animate' software interface. It includes a timeline with 'Eigen First' set to 1, 'Last' set to 58, 'Inc' set to 1, 'Time' set to 1.9823, and 'State' set to 58. There are buttons for 'Animate', 'Loop', and a progress bar at the bottom.

Eigen First: 1 Last: 58 Inc: 1 Time: 1.9823 State: 58

Animate [Play] [Stop] [Reset] [Loop] [30S] [F]

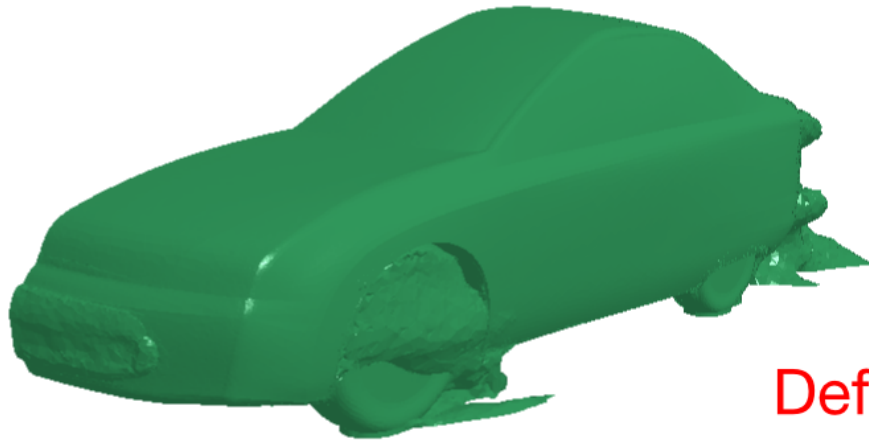
# ICFD Post Processor Introduction

A right click on the initial Object (here the fluid volume) pops up a Menu which allows the user to create new objects

ANSYS Car  
Time = 1.9823



New Isosurface  
Object created



Default :  
PID color

The screenshot shows the 'icfd' software interface. The 'Objects' panel lists several objects, with 'IsoSurface\_7' selected and checked. Below the list are buttons for 'All', 'None', and 'Reverse'. The 'Object Properties' section shows the following data:

Type	Fluid velocity
Iso Min	-38.582359
Iso Threshold	10
Iso Max	98.459824

Below the properties is the text 'Velocity isosurface'. The 'Display Options' section shows the following settings:

Mode	Shade
Color by	PID
Fringe Contour	<input checked="" type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	0
AVG Scalar	0
Max Scalar	0



The 'Animate' panel shows the following settings:

Eigen First: 1 Last: 58 Inc: 1 Time: 1.9823 State: 58

Animate  Loop

30S F

# ICFD Post Processor Introduction

## Modifying newly created object:

**ANSA Car**  
Time = 1.9823

The screenshot displays the ANSA ICFD post-processor interface. On the left, a 3D model of a car is shown with a green and yellow isosurface. A red arrow points to the car with the text: "Display options : Coloring isosurface with field variable (eg : pressure)".

The central window, titled "icfd", shows the "Objects" list with "IsoSurface\_7" selected. Below it, the "Object Properties" table is visible:

Type	Fluid velocity
Iso Min	-38.582359
Iso Threshold	10
Iso Max	98.459824

Below the properties is the "Display Options" section:

Mode	Shade
Color by	Fluid pressure
Fringe Contour	<input checked="" type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	-7199.337891
AVG Scalar	-2341.277344
Max Scalar	2516.782959

Red arrows point from the "Fringe Legend" checkbox in the "Display Options" to the "Optional Fringe legend display" text, and from the "Fringe Legend" checkbox to the "Optional Fringe legend display" text.

On the right, the "IsoSurface\_7" legend is shown with a color scale ranging from -7.199e+03 (blue) to 2.517e+03 (red). A red arrow points from the "Optional Fringe legend display" text to the legend.

At the bottom, the "Animate" window is visible, showing the current time as 1.9823 and the state as 58.

# ICFD Post Processor Introduction

New object created : Splane

The screenshot displays the ANSA software interface. The main view shows a car model in a wireframe mesh, with a new Splane object (SectionPlane\_10) created. The car is colored by PID, and the Splane is shown in wire mode. The Objects panel on the right shows the hierarchy of objects, with SectionPlane\_10 selected. The Object Properties panel shows the position, normal, and grid settings for the Splane. The Animate panel at the bottom shows the animation controls, including time and state settings.

ANSA Car  
Time = 1.9823

Objects

- MS\_SHELL\_6
- MS\_SOLID\_10
  - IsoSurface\_7
  - IsoSurface\_8
  - IsoSurface\_9
  - SectionPlane\_10

All None Reverse

Object Properties

Position	3212.21; -0.018675; 4
Normal	0; 1; 0
Plane	<input checked="" type="checkbox"/>
Grid	<input type="checkbox"/>
Grid nx	20
Grid ny	20

Display Options

Mode	Wire
Color by	PID
Fringe Contour	<input type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	0
AVG Scalar	0
Max Scalar	0

Wire Mode

Animate

Eigen First: 1 Last: 58 Inc: 1 Time: 1.9823 State: 58

Animate  Loop

30S F

# ICFD Post Processor Introduction

**ANSA Car**  
Time = 1.9823

Centd CG NmiX NmiY NmiZ 1P+NL N1-N2 3NPS

**Coloring Splane by Velocity**

**icfd**

Objects

- MS\_SHELL\_6
- MS\_SOLID\_10
  - IsoSurface\_7
  - IsoSurface\_8
  - IsoSurface\_9
  - SectionPlane\_10

All None Reverse

Object Properties

Position	3212.21; -0.018675; 4
Normal	0; 1; 0
Plane	<input checked="" type="checkbox"/>
Grid	<input type="checkbox"/>
Grid nx	20
Grid ny	20

Display Options

Mode	Shade
Color by	Fluid velocity
Fringe Contour	<input checked="" type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	0
AVG Scalar	57.333698
Max Scalar	114.667397

**Animate**

Eigen First: 1 Last: 58 Inc: 1 Time: 1.9823 State: 58

Animate  Loop

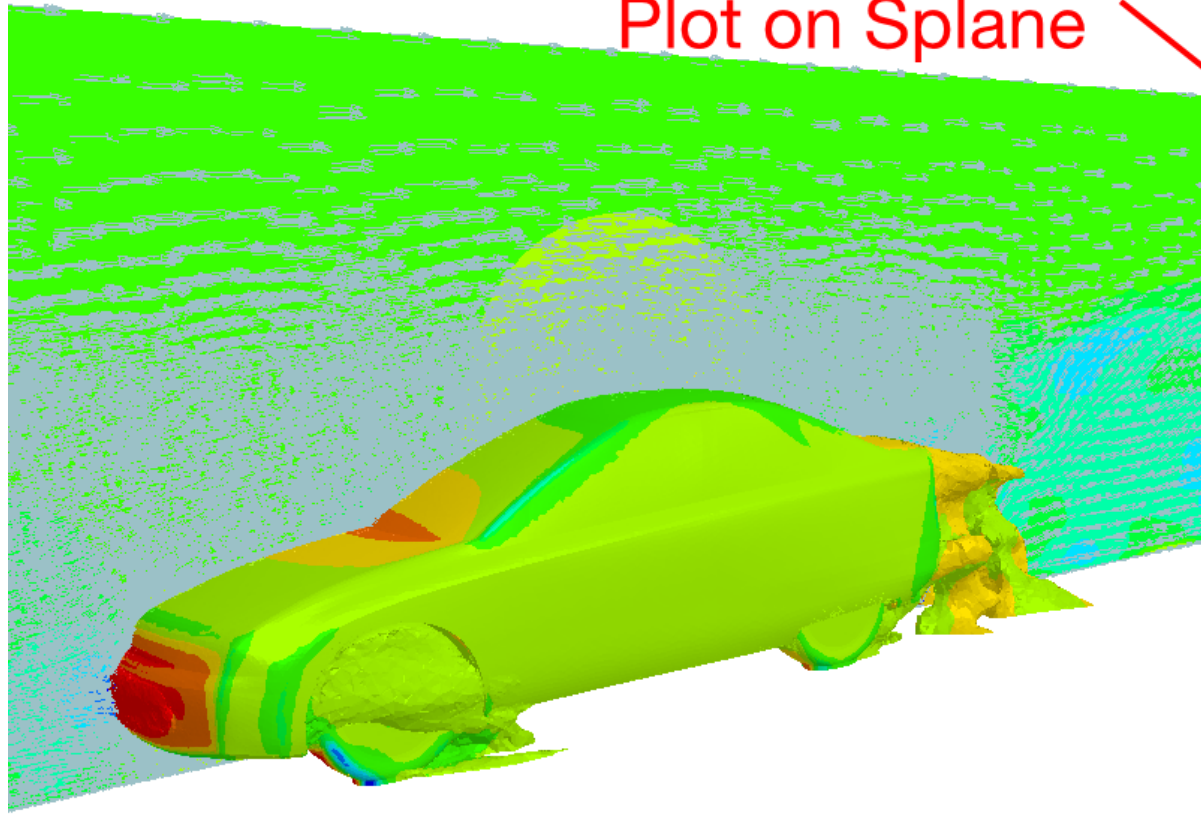
30S F



# ICFD Post Processor Introduction

ANSA Car  
Time = 1.9823

Tree hierachy : vector  
Plot on Splane



icfd

Objects

- MS\_SHELL\_6
- MS\_SOLID\_10
  - IsoSurface\_7
  - IsoSurface\_8
  - IsoSurface\_9
  - SectionPlane\_10
    - VectPlot\_11

All None Reverse

Object Properties

Type	Fluid velocity
Vect Component	X; Y; Z
Format Type	Line
Const Size	<input type="checkbox"/>
Length Scale	0.2
Head Scale	1
Tail Scale	1

Display Options

Mode	Shade
Color by	PID
Fringe Contour	<input type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	0.01024
AVG Scalar	43.429359
Max Scalar	86.84848



Animate

Eigen First: 1 Last: 58 Inc: 1 Time: 1.9823 State: 58

Animate     Loop

30S

# ICFD Post Processor Introduction

**ANSA Car**  
Time = 1.9823

Possible to display Vector  
in "Grid Mode" to better  
see Velocity gradients

**icfd**

Objects

- MS\_SHELL\_6
- MS\_SOLID\_10
- IsoSurface\_7
- IsoSurface\_8
- IsoSurface\_9
- SectionPlane\_10
- VectPlot\_11

All None Reverse

Object Properties

Position	3212.21; -0.018675; 4
Normal	0; 1; 0
Plane	<input checked="" type="checkbox"/>
Grid	<input checked="" type="checkbox"/>
Grid nx	40
Grid ny	50

Display Options

Mode	Shade
Color by	Fluid velocity
Fringe Contour	<input checked="" type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	0
AVG Scalar	57.333698
Max Scalar	114.667397

**Animate**

Eigen First: 1 Last: 58 Inc: 1 Time: 1.9823 State: 58

Animate  Loop

30S F

# ICFD Post Processor Introduction

**ANSA Car**  
Time = 1.9823

Second Splane displayed

Of course, it is possible to create several objects of one kind

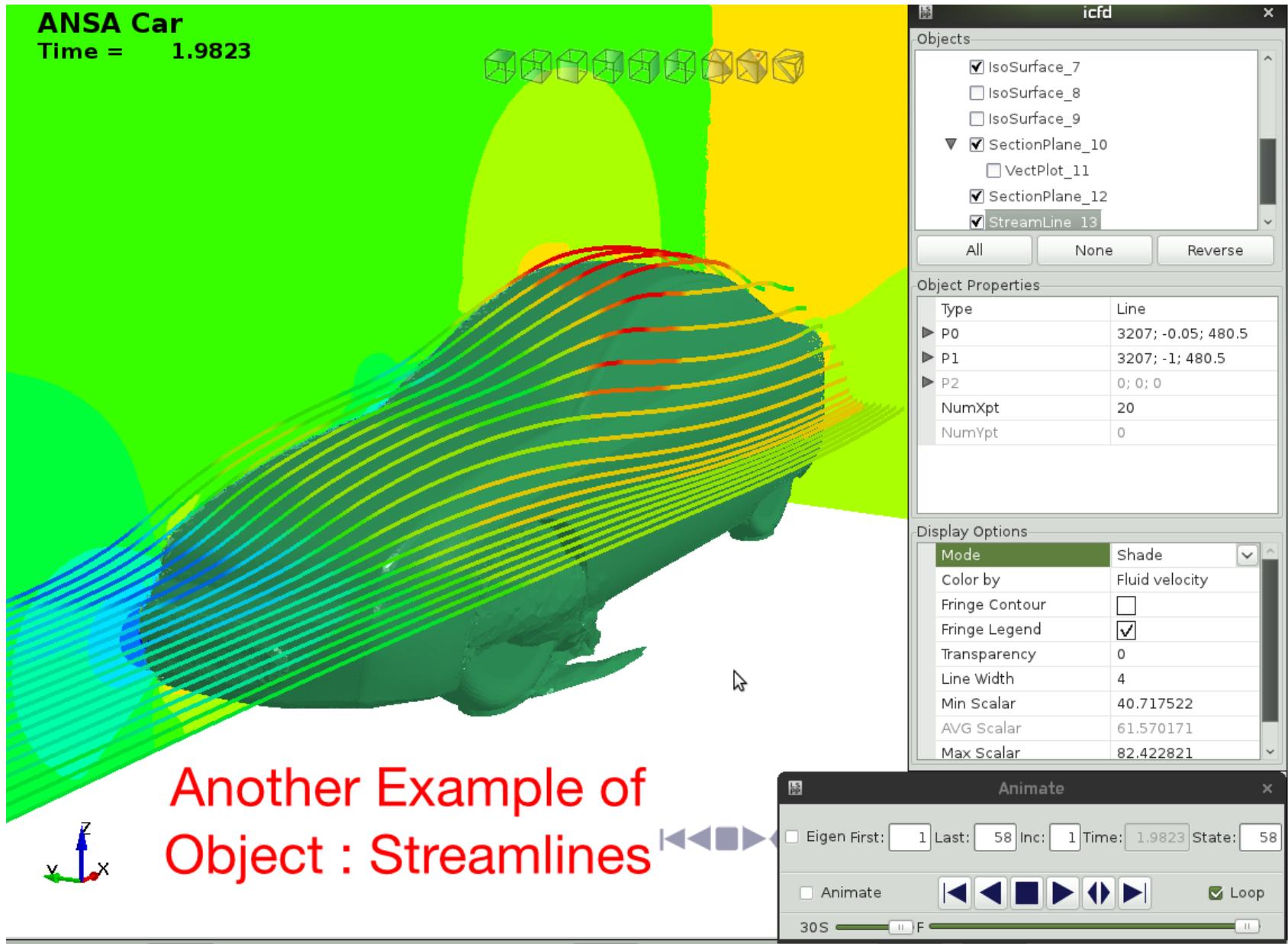
The screenshot shows the ANSA software interface. At the top left, it displays 'ANSA Car' and 'Time = 1.9823'. A toolbar with various icons is visible above the main 3D view. The 3D view shows a car model with a green wireframe grid and a yellow spline. A red arrow points to the spline with the text 'Second Splane displayed'. Below the car model, there are several buttons: 'Centd', 'CG', 'NmX', 'NmY', 'NmZ', '1P-NL', 'N1-N2', and '3NPS'. A coordinate system with X, Y, and Z axes is shown in the bottom left. On the right side, there are two panels. The top panel is titled 'icfd' and contains an 'Objects' list with checkboxes for 'MS\_SOLID\_10', 'IsoSurface\_7', 'IsoSurface\_8', 'IsoSurface\_9', 'SectionPlane\_10', 'VectPlot\_11', and 'SectionPlane\_12'. Below the list are buttons for 'All', 'None', and 'Reverse'. The 'Object Properties' section shows 'Position' (3213.21; -1.49993; 48), 'Normal' (1; 0; 0), 'Plane' (checked), 'Grid' (unchecked), 'Grid nx' (20), and 'Grid ny' (20). The bottom panel is titled 'Animate' and shows 'Eigen First: 1 Last: 58 Inc: 1 Time: 1.9823 State: 58'. It has an 'Animate' checkbox, navigation buttons, a 'Loop' checkbox, and a progress bar.

Object	Checked
MS_SOLID_10	<input type="checkbox"/>
IsoSurface_7	<input checked="" type="checkbox"/>
IsoSurface_8	<input type="checkbox"/>
IsoSurface_9	<input type="checkbox"/>
SectionPlane_10	<input checked="" type="checkbox"/>
VectPlot_11	<input type="checkbox"/>
SectionPlane_12	<input checked="" type="checkbox"/>

Property	Value
Position	3213.21; -1.49993; 48
Normal	1; 0; 0
Plane	<input checked="" type="checkbox"/>
Grid	<input type="checkbox"/>
Grid nx	20
Grid ny	20

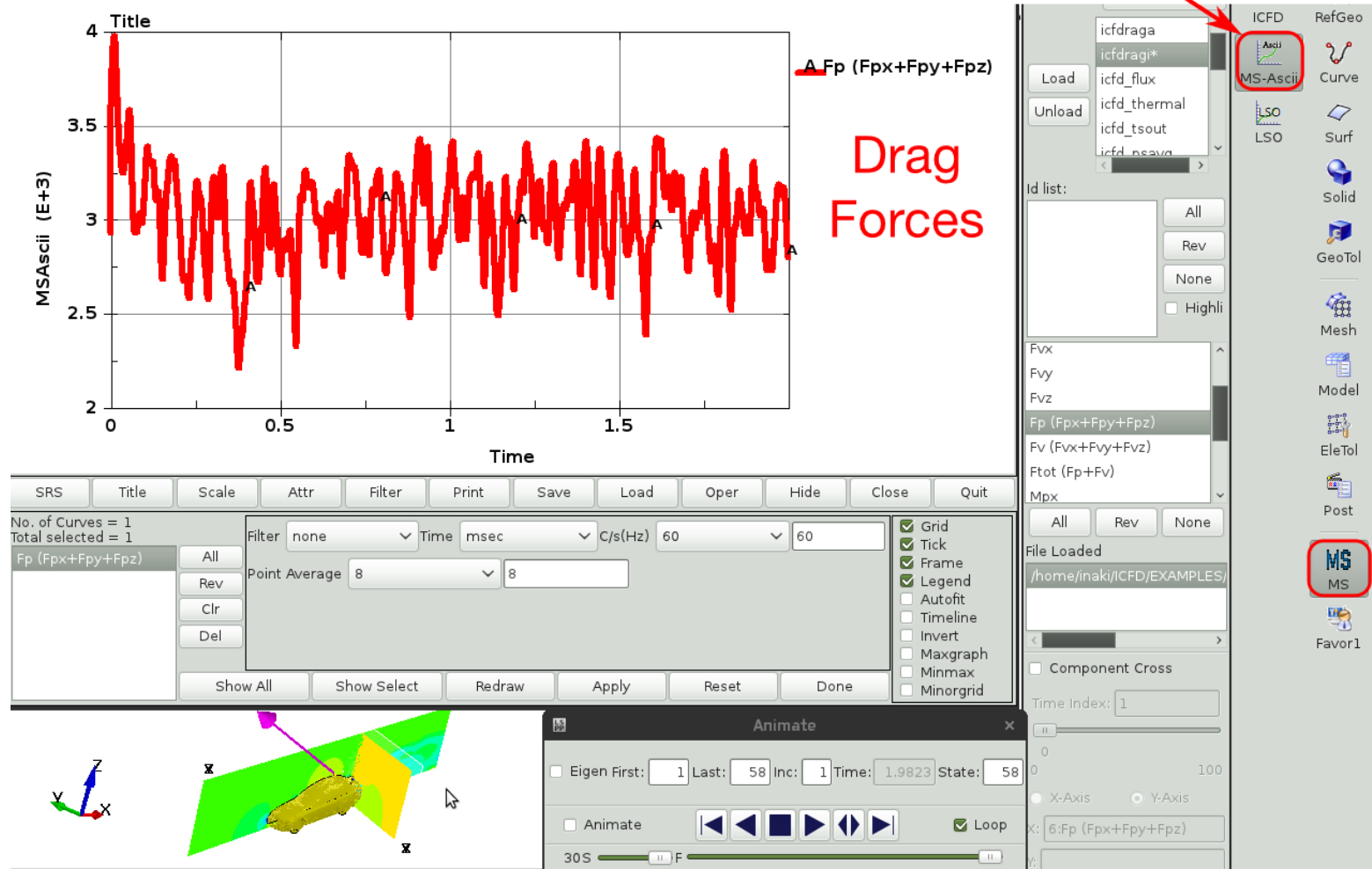
Property	Value
Mode	Shade
Color by	Fluid pressure
Fringe Contour	<input checked="" type="checkbox"/>
Fringe Legend	<input checked="" type="checkbox"/>
Transparency	0
Line Width	1
Min Scalar	-7210.088379
AVG Scalar	-2286.58252
Max Scalar	2636.923096

# ICFD Post Processor Introduction



# ICFD Post Processor Introduction

It is also possible to post treat all ASCII files (See \*ICFD\_DATABASE family) dumped by the ICFD solver (forces, flux, point data etc)

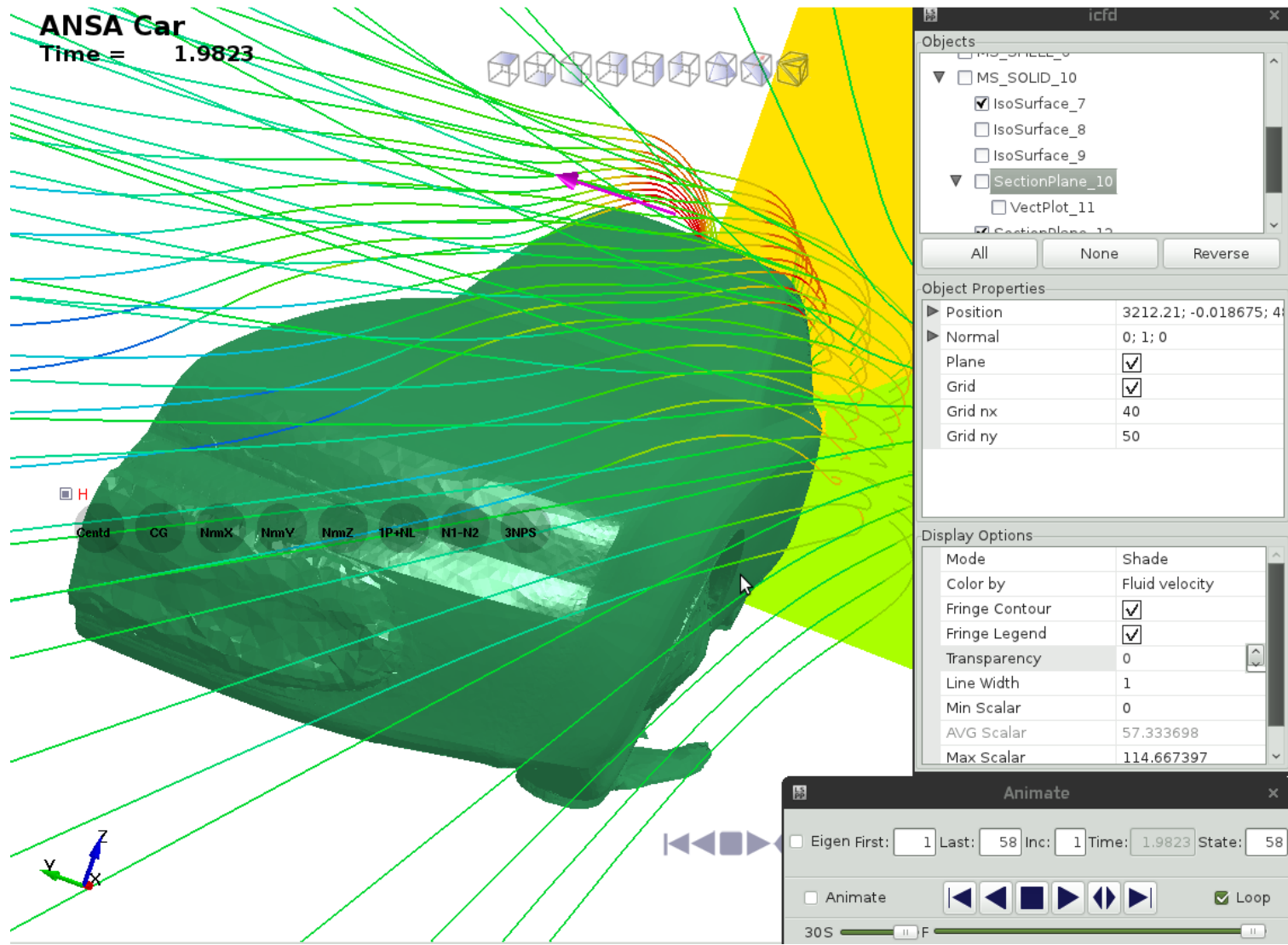


# ICFD Post Processor Introduction

- Currently, four type of objects can be created : Splane, isosurfaces, streamlines and vectors. Those are the most commonly used visualization tools in order to study flow patterns.
- More object types may be implemented in the future. A project is underway in collaboration with the University of San Diego in order to offer further advanced tools for qualitative post treating of the results.
- The next step of development will include some new features for a more flexible and dynamic post treatment of results and data (easy extraction of values from the mesh, curve plotting options etc..)
- LS-PrePost 4.2 is currently in alpha stage but can already be used for users eager to beta test its current functionalities.

# ICFD Post Processor Introduction

Thank you for your attention !



# Summary

- **LSTC is working to be the leader in cost effective large scale numerical simulations**
  - LSTC is providing dummy, barrier, and head form models to reduce customer costs.
  - LS-PrePost, LS-Opt, and LS-TaSC are continuously improving and gaining more usage within the LS-DYNA user community
  - LSTC is actively working on seamless multistage simulations in automotive crashworthiness, manufacturing, and aerospace
- **The scalable implicit solver is quickly gaining market acceptance for linear/nonlinear implicit calculations and simulations**
  - Robustness, speed, accuracy, and scalability have rapidly improved
  - New developments:
    - Combined implicit and explicit running together
    - Linear analysis combined with h-adaptivity



# Future

- LSTC is not content with what has been achieved
- New features and algorithms will be continuously implemented to handle new challenges and applications
  - Electromagnetics,
  - Acoustics,
  - Compressible and incompressible fluids
  - Isogeometric shell elements
  - Discrete element methodology for modeling granular materials
  - Simulation based airbag folding and THUMS dummy positioning underway
- Multiscale capabilities are under development
  - Implementation underway (New approach which is more user friendly)
- Hybrid MPI/OPENMP developments are showing significant advantages at high number of processors for both explicit and implicit solutions

Thank You !